

Days and Inauguration of the MOSAIC platform

Conception and realization of a plasma space thruster demonstrator

September 25, 2024 – Romain BELLET

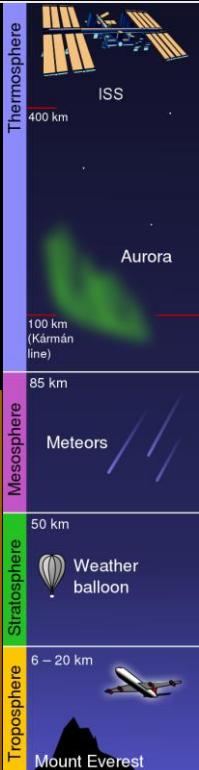
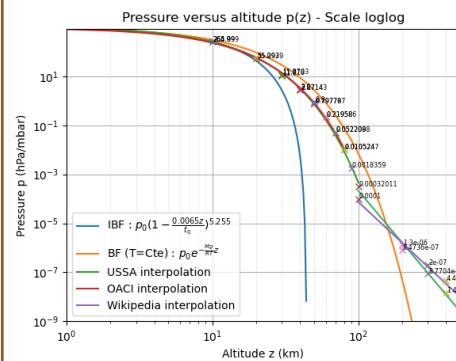
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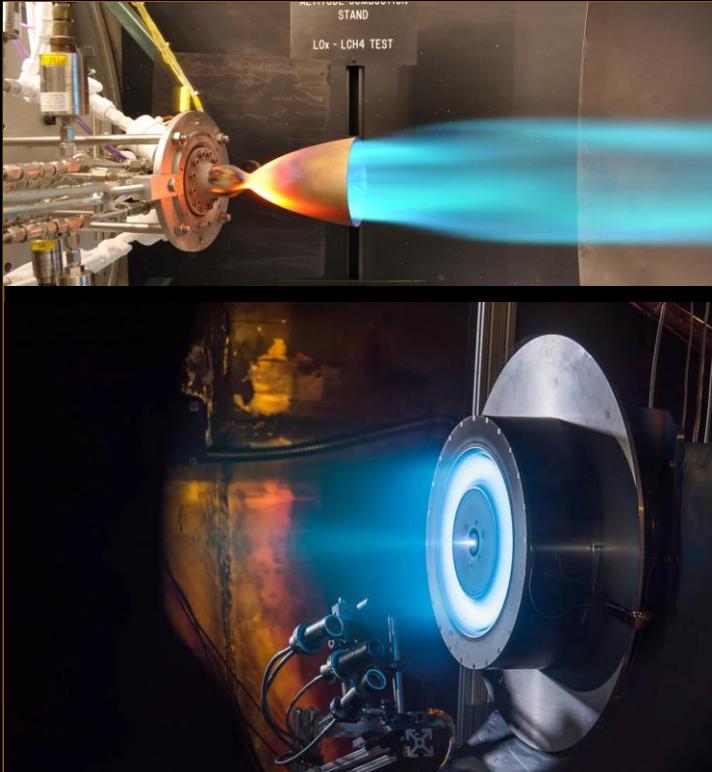
Space Electric Propulsion

Space

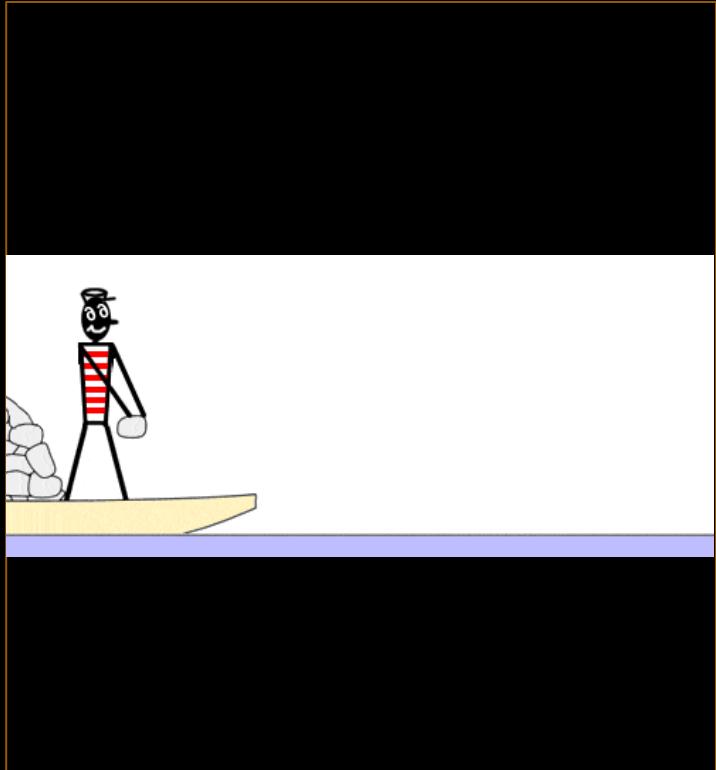
- $z_{EP} \geq 350 \text{ km}$
- $(\theta \simeq 850^\circ C)$
- $p \leq 10^{-7} \text{ mbar}$



Electric

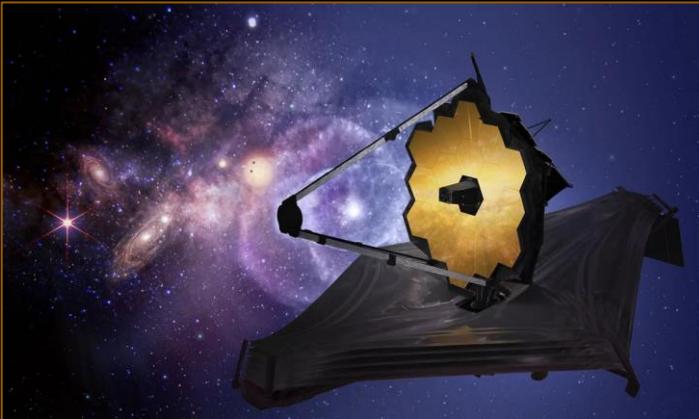


Propulsion





Outlook



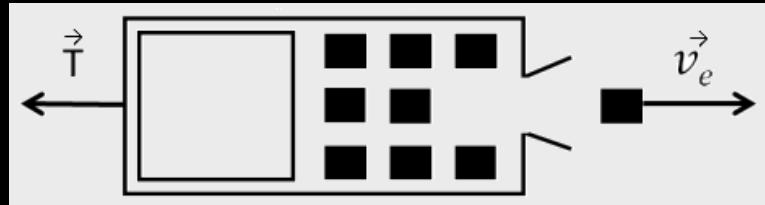


Performance criteria

Thrust

$$\vec{T} = \frac{dm}{dt} \vec{v}_e$$

- Reaction force (Newton's 3rd law)



Specific Impulse

$$I_{sp} = \frac{Impulse}{Weight} = \frac{v_e}{g_0}$$

- “How many **seconds** a given propellant, when paired with a given engine, can accelerate its own initial masse at g.”





Project reminder

Thesis : Conception and realization of a plasma space thruster demonstrator

Multicharged Ion Source Thruster (**MIST**)

$[T = 1 \text{ N}; I_{sp} = 100\,000 \text{ s}]$
 $(17\,000 \text{ s})$

Based on ECR ion sources

- From particle accelerator physics
- High current intensities of multicharged ions



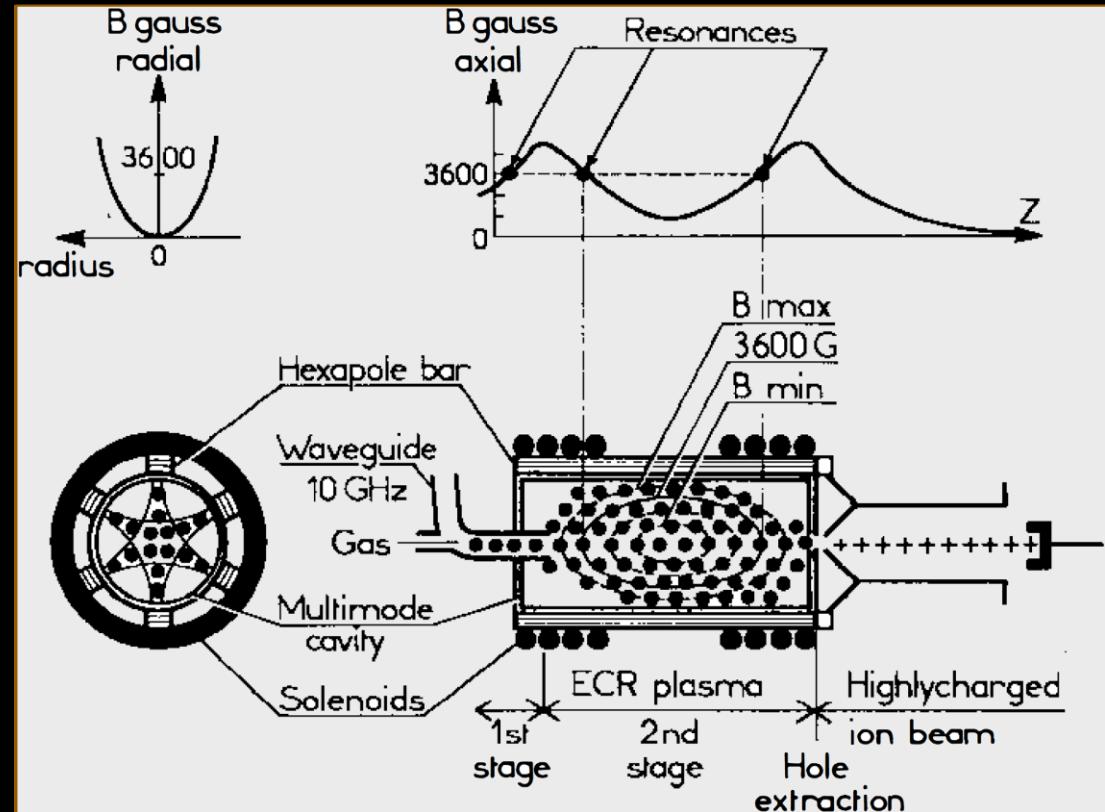
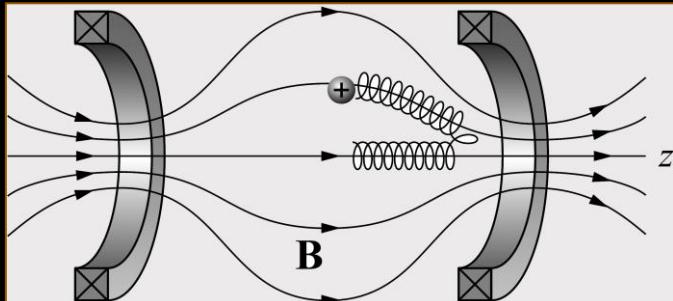


ECR Ion Source (ECRIS)

Electron Cyclotron Resonance

$$\omega_c = \frac{eB}{m}$$

$$\rho_L = \frac{mv_{\perp}}{eB}$$





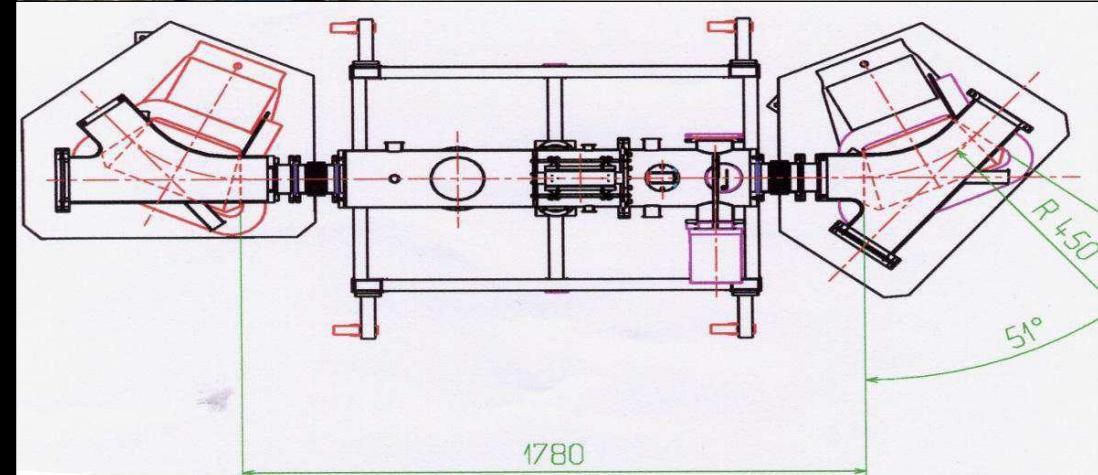
Tancrede



- Multicharged ion probe 30 kV
 - PhDc training
 - First studies

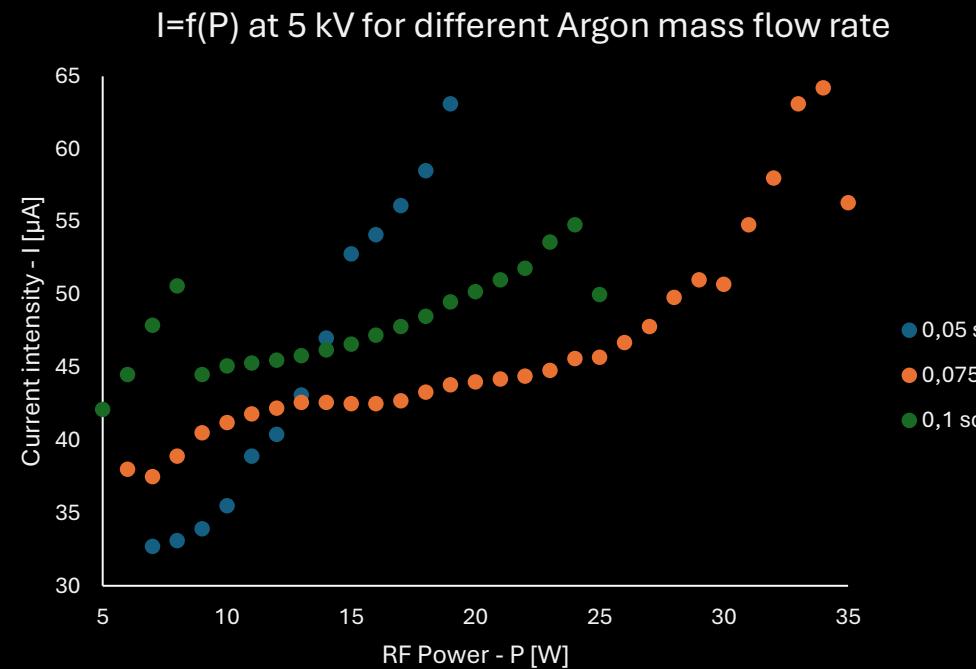
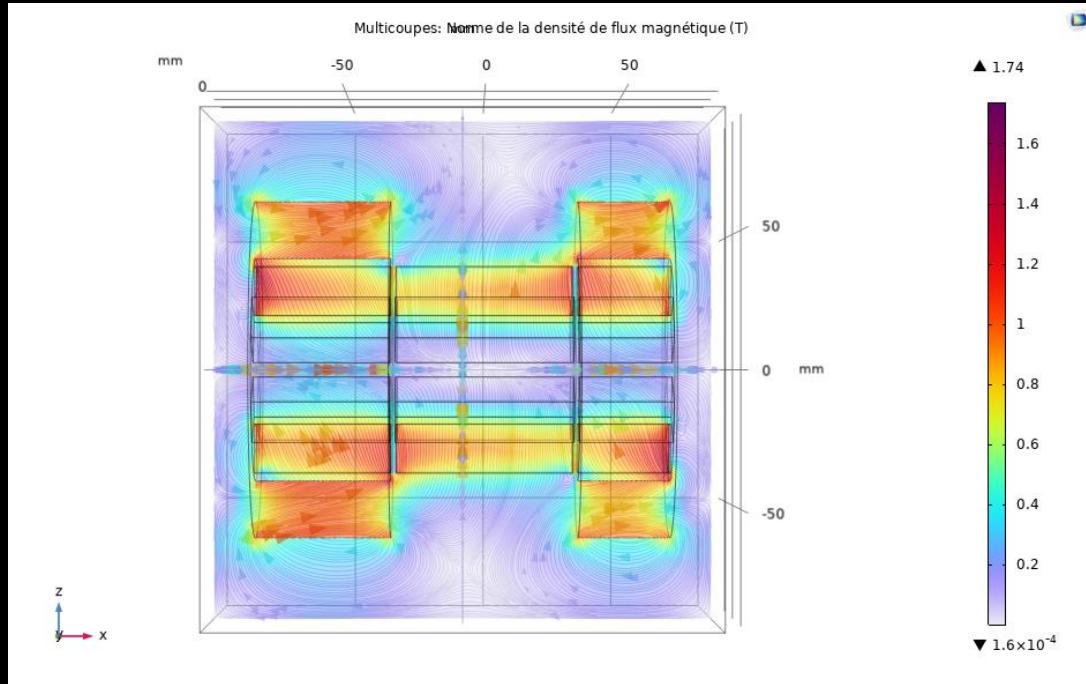
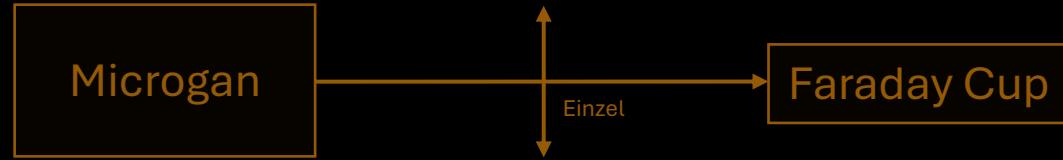
ECRIS → Einzel → Dipole
↳ Faraday Cup (0°)
↳ Diag. (51°)

- Microgan ECRIS 10 GHz





Microgan characterization

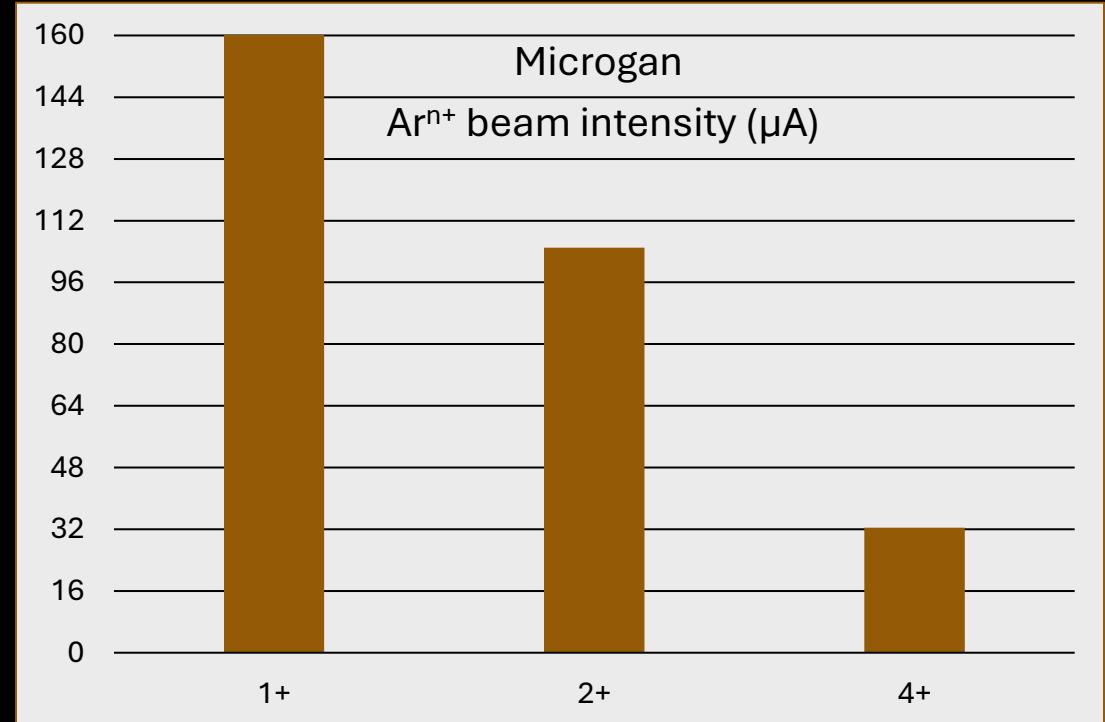




Microgan compute

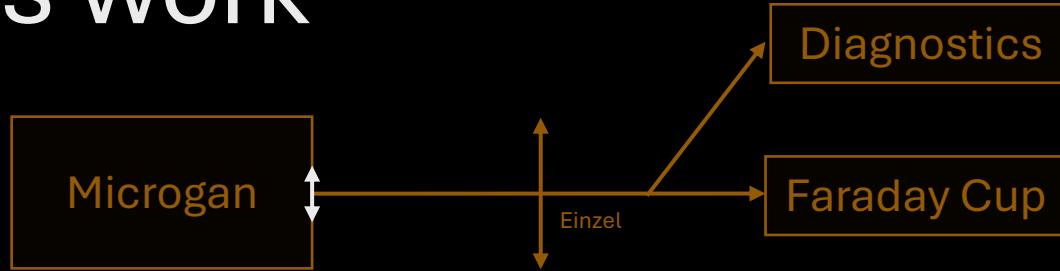


Microgan first starts	Full Power Microgan
0,075 sccm	→ 1,00 sccm
34,0 W	200 W
5,00 kV	30,0 kV
64,2 µA (Ar^{1+})	See fig. →
4,1 µN	35,2 µN
15,8 ks	38,7 ks
$\eta \sim 0,10$	$\eta \sim 0,10$

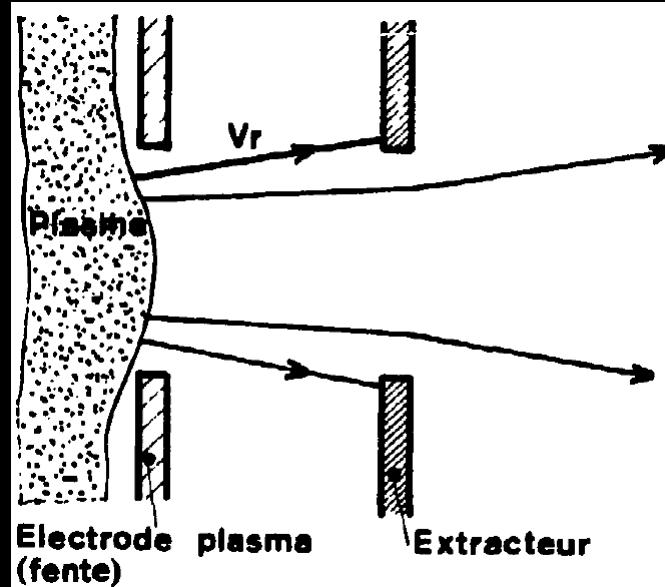




Thesis work



- Increase extracted current
 - ↳ $I \propto \phi_{\text{extraction}}^2$
 - ↳ Plasma meniscus behavior
- Measure Thrust from ECRIS



Conclusion

- Space Electric Propulsion
- Multicharged Ion Space Thruster based on ECR Ion Source
- Tancrede to compute Thrust by ECRIS

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