

# Data decomposition for unsteady combustion analysis

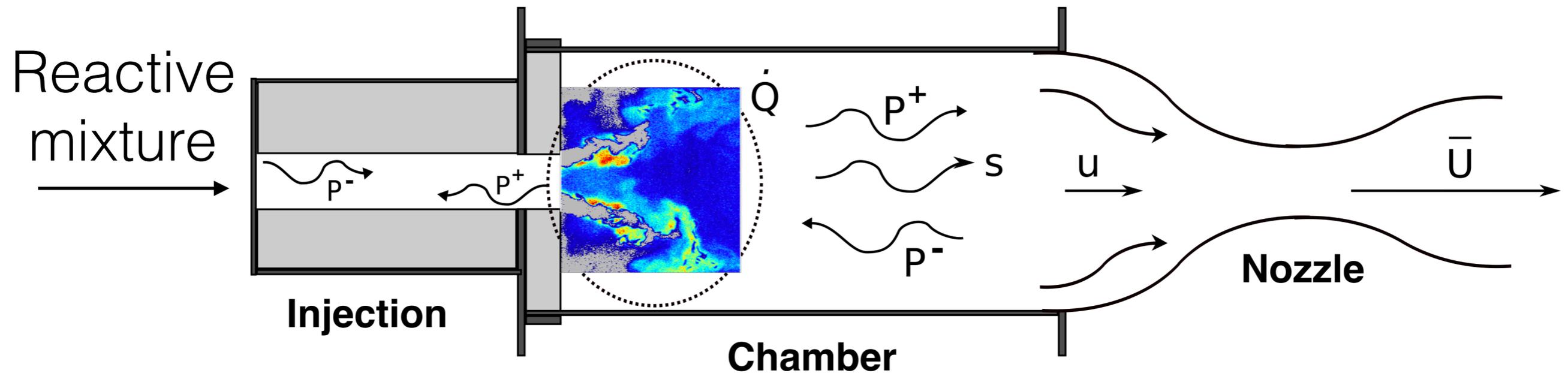
Franck Richecoeur

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Ecole Centrale Paris

29 Novembre 2017

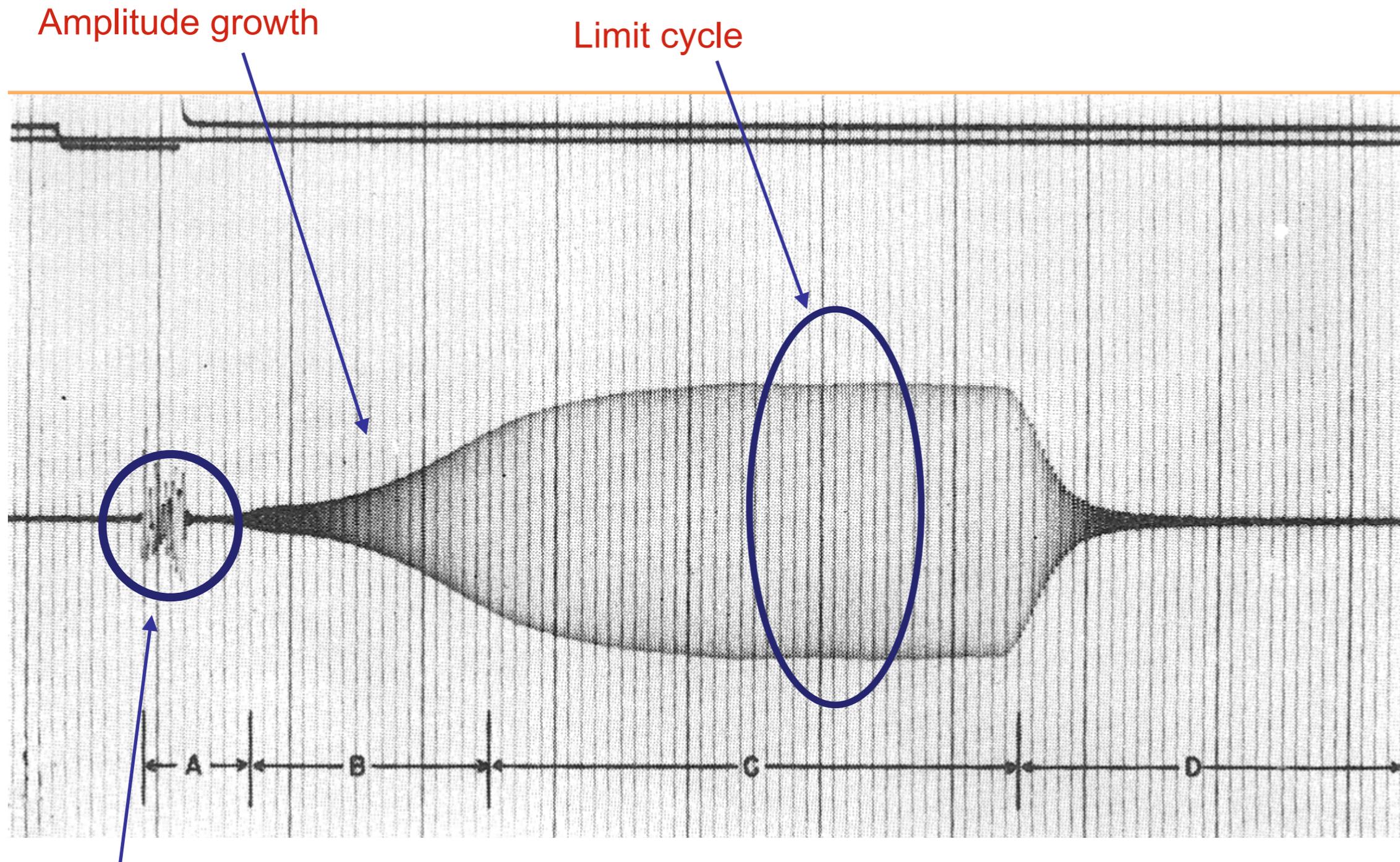


# Combustion is based on multi-physics process that interact in a chamber



Turbulent combustion is an unsteady process that generate interactions between **velocity, pressure and temperature**, in strong interactions with the boundaries of the system.

# Combustion is based on multi-physics process that interact in a chamber

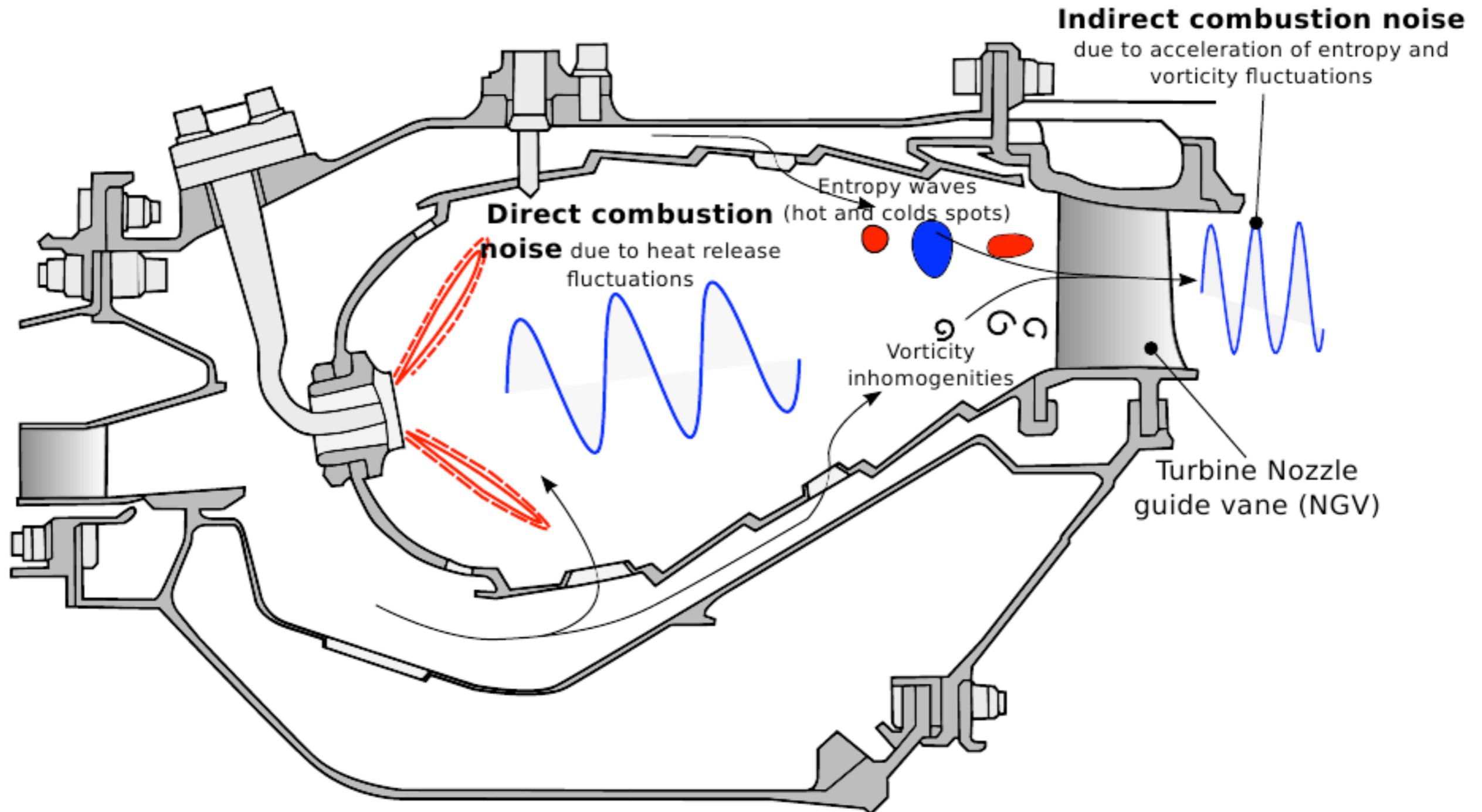


Pressure signal in a burner  
(Horton & Price, 1962)

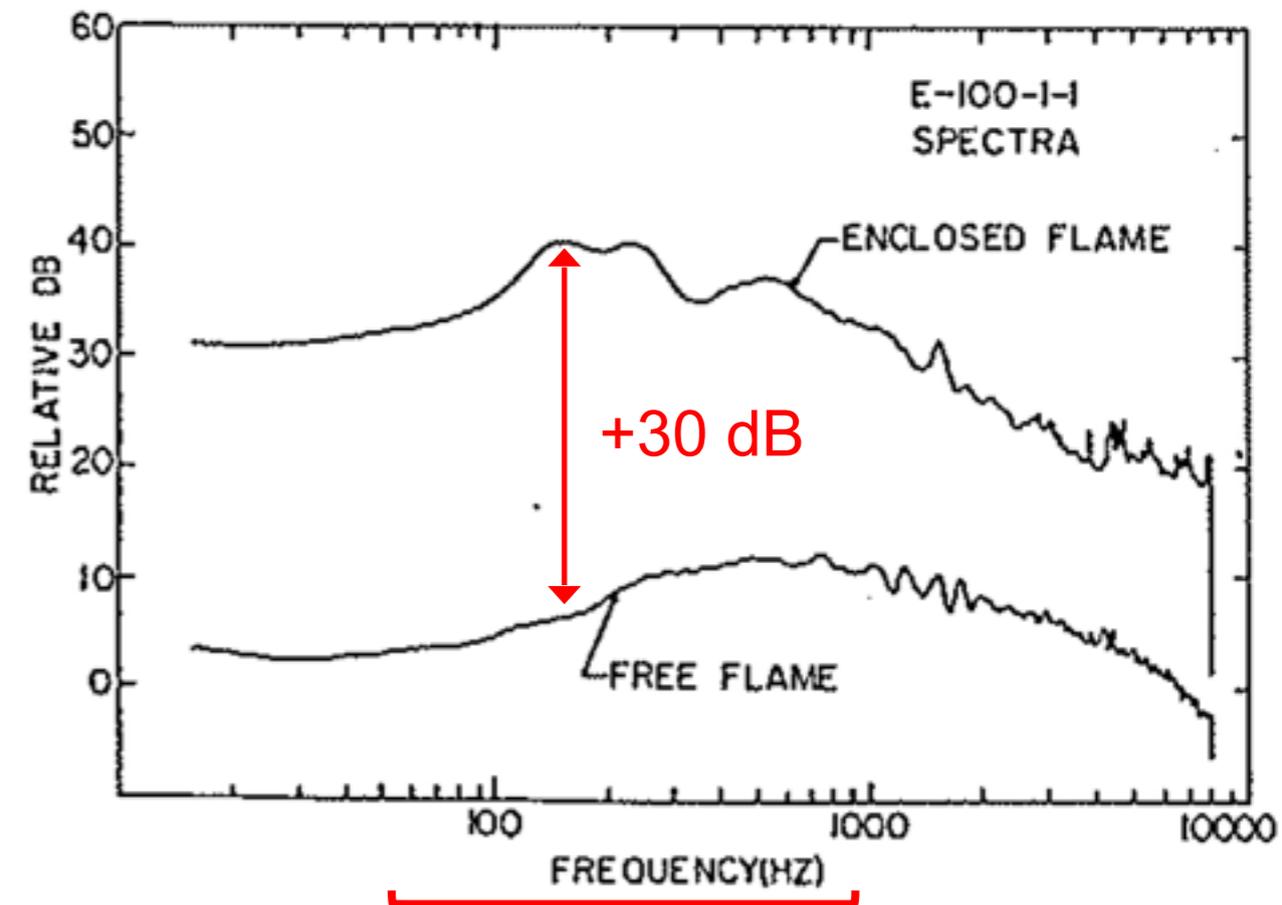
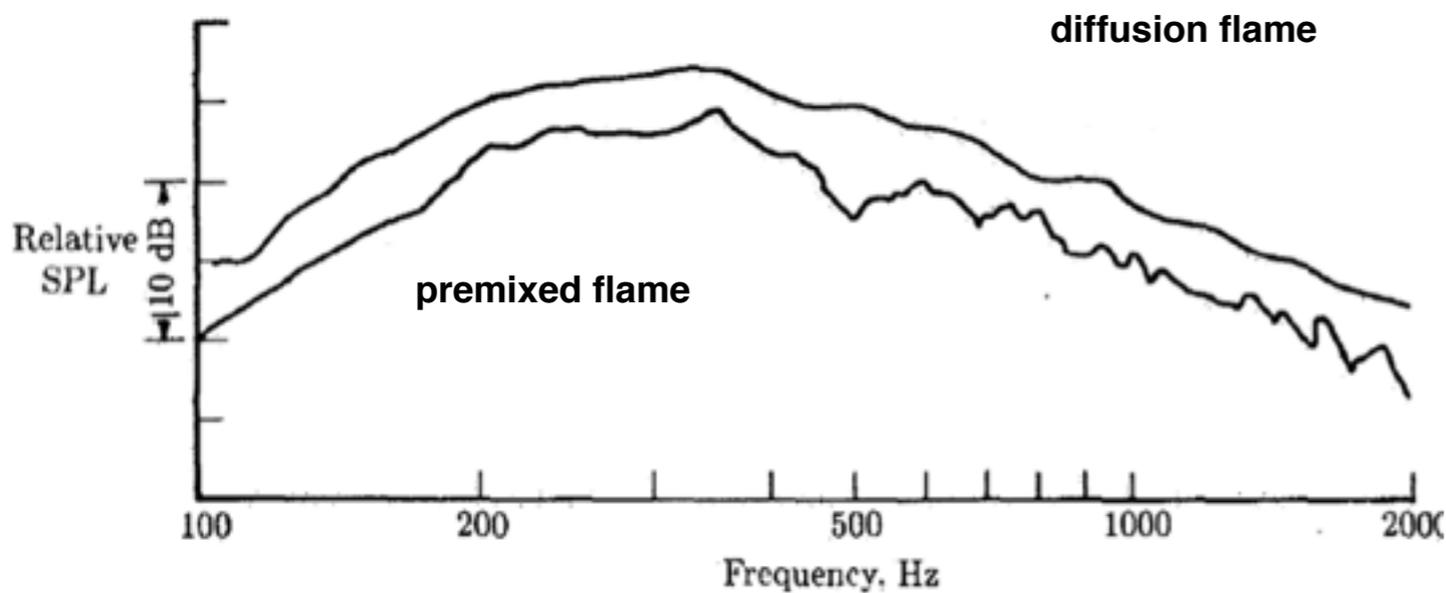
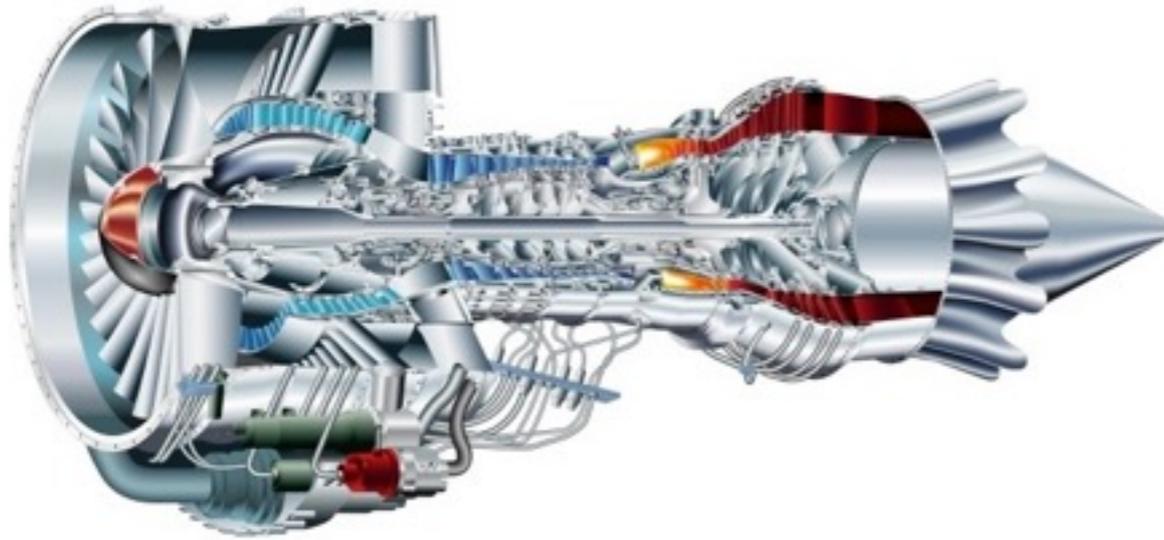
Small perturbation

Most of the time, combustion chambers are unstable and we try to minimize the amplitude of the limit cycle.

# Mastering of the pressure perturbations in the surrounding of the chamber is one the challenges in combustion



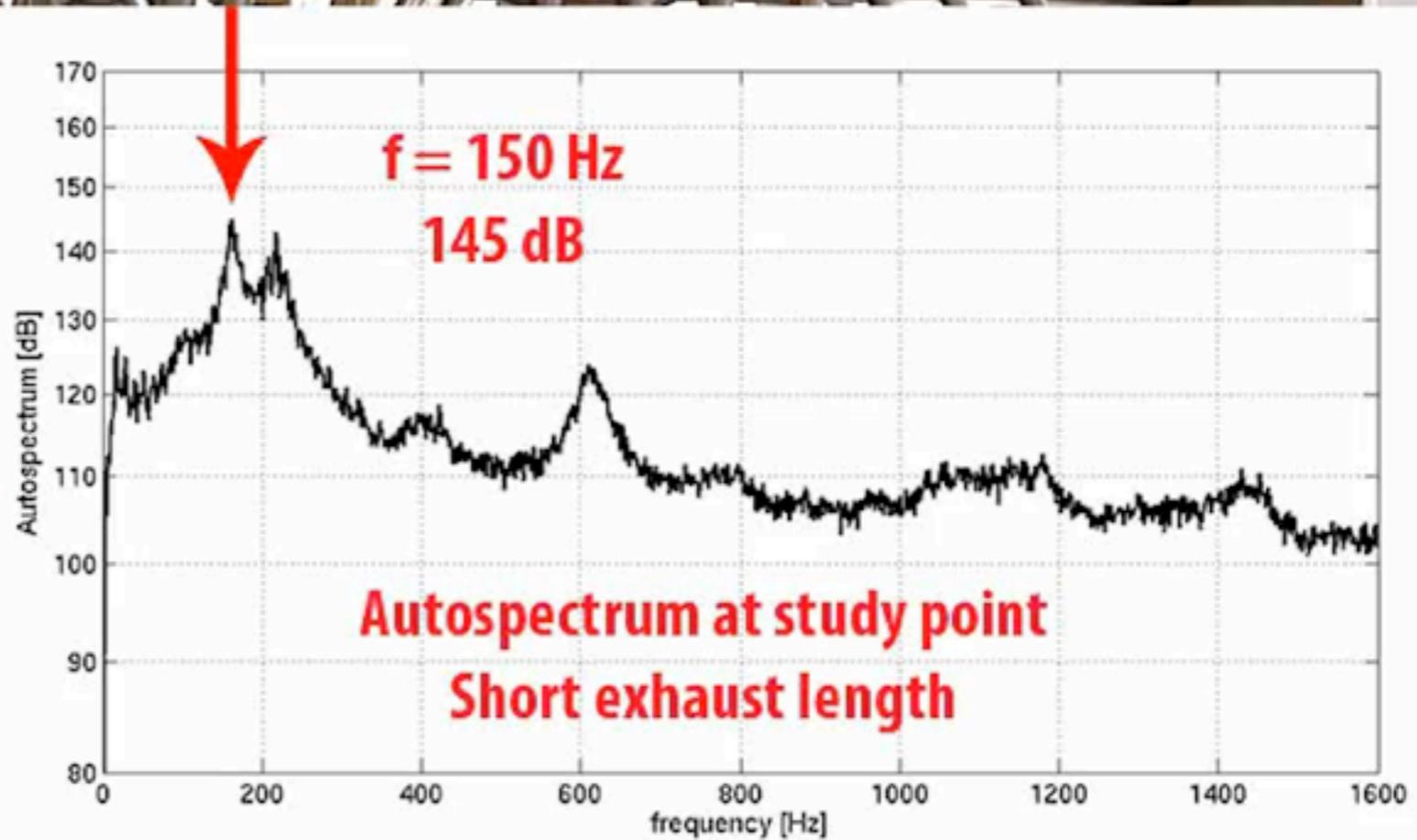
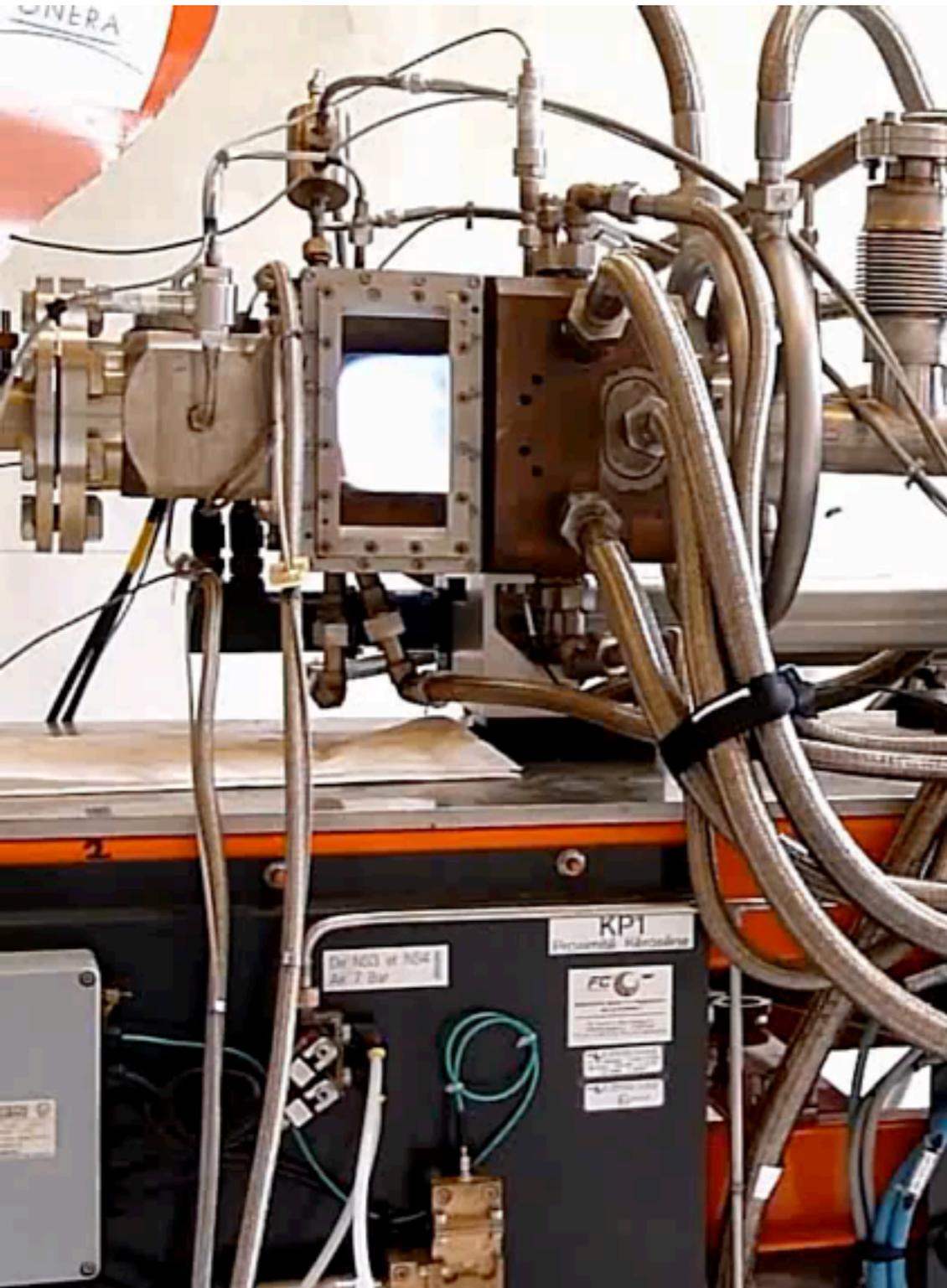
# Noise depends on the operating conditions.



Typical sound pressure level spectra in the far field of unconfined, turbulent, premixed and diffusion flames. AIAA paper No.73-1023, Oct.1973

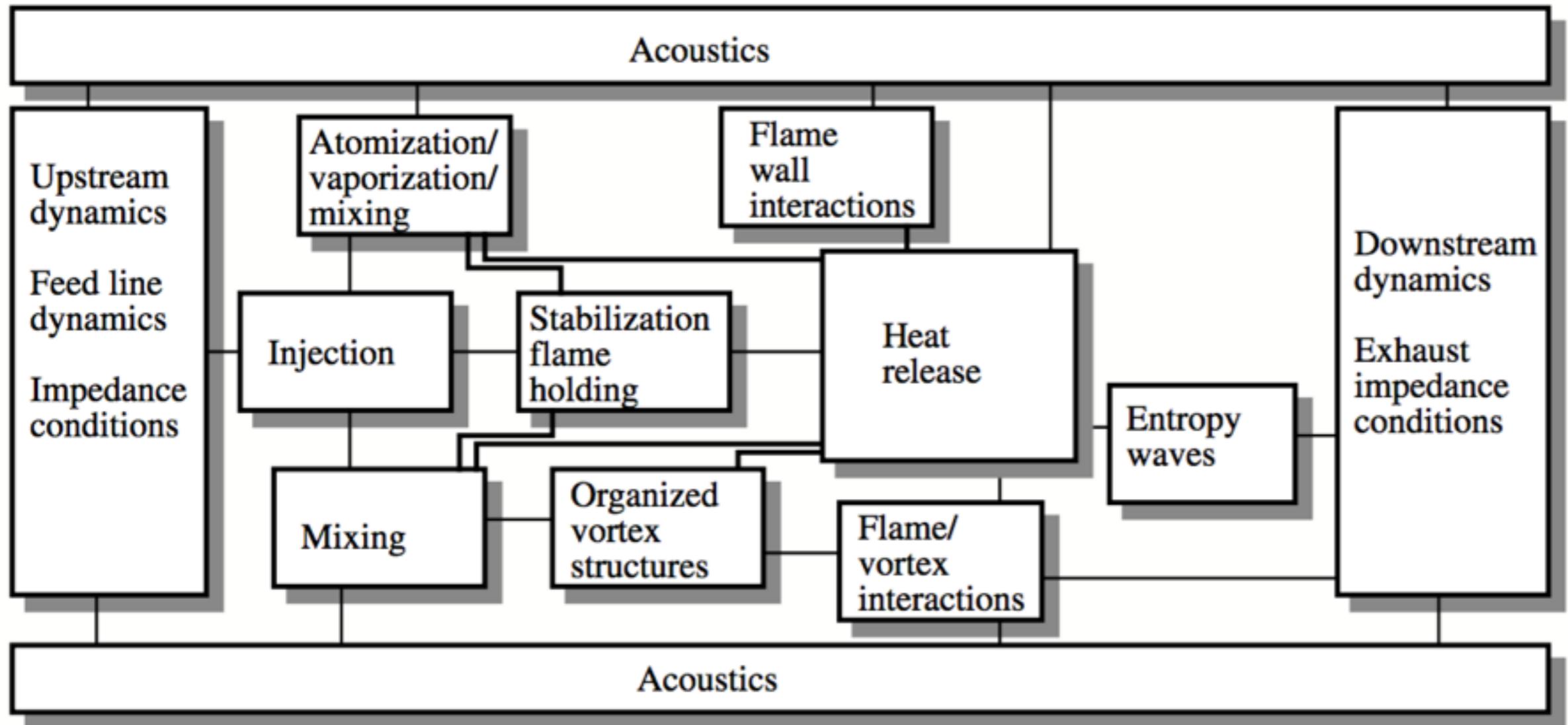
Strahle et al., (1975) AIAA Paper 75-127

# LACOM Bench at ONERA test the influence of outlet conditions on the pressure field.

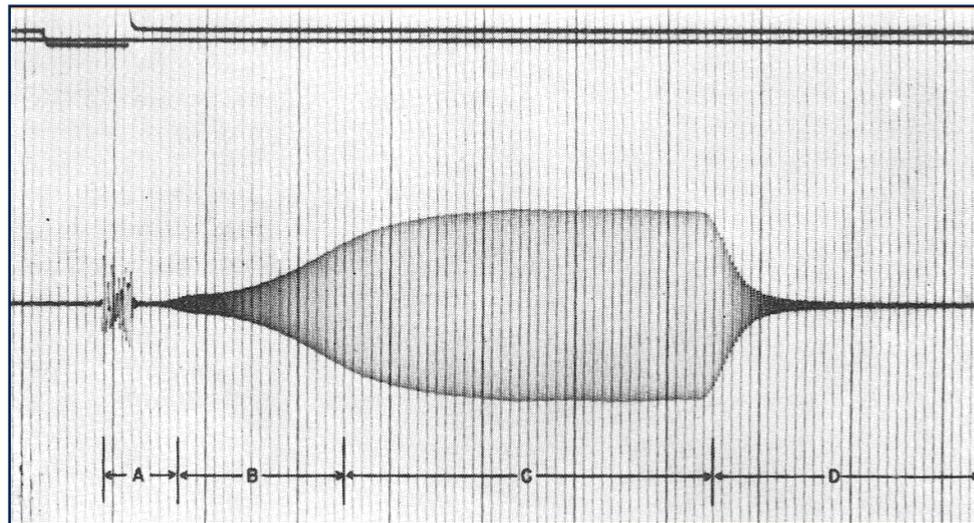


# Couplings

Pressure field perturbations are due to numerous couplings that need intense data processing to understand



# The importance of data decomposition to characterize couplings.



The challenge is to keep pressure amplitudes in a safe domain.

Understand the transient and limit phases.

These amplitudes depend on

- the acoustic sources (jet, combustion, vortex, shear layer...),
- the damping mechanisms (acoustic impedance, damping, turbulence),
- and propagation medium (upstream in fresh gases, downstream in burnt gases...)

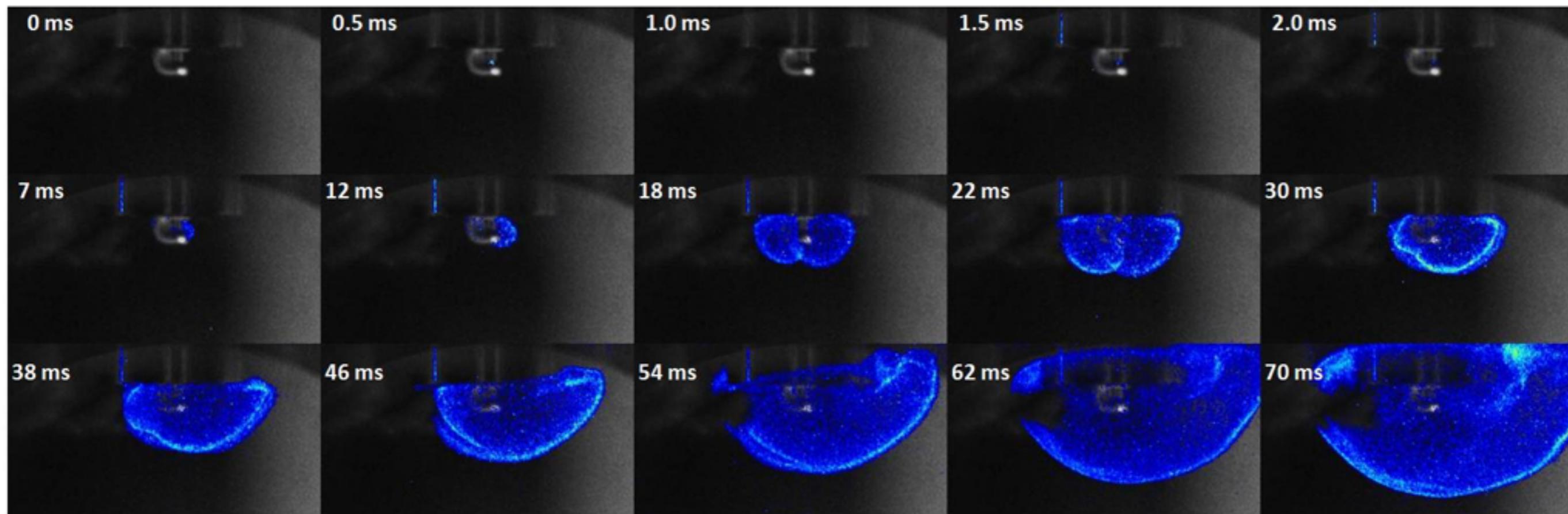
Data decomposition helps in separating time and space scales.

# IC Engine Example

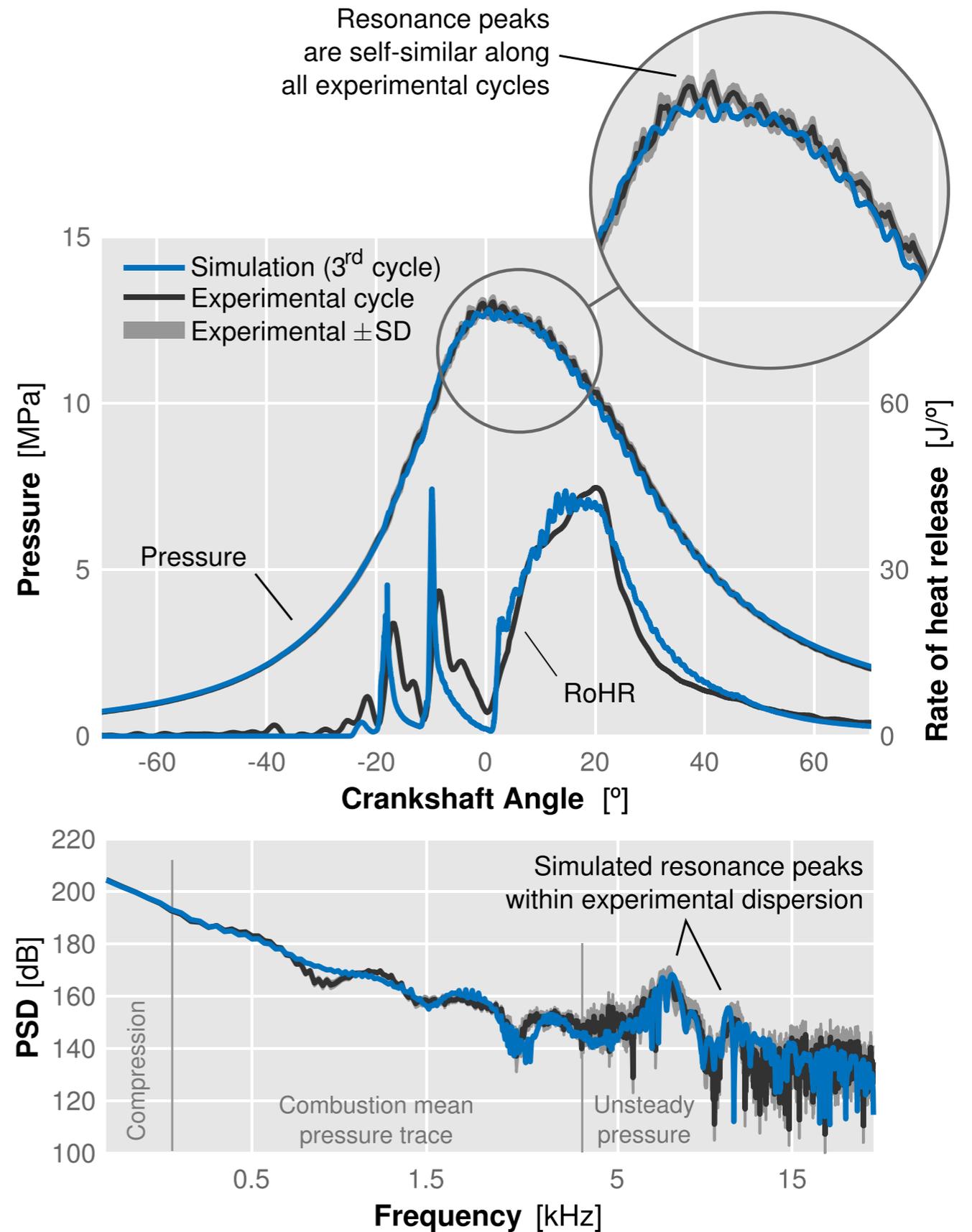
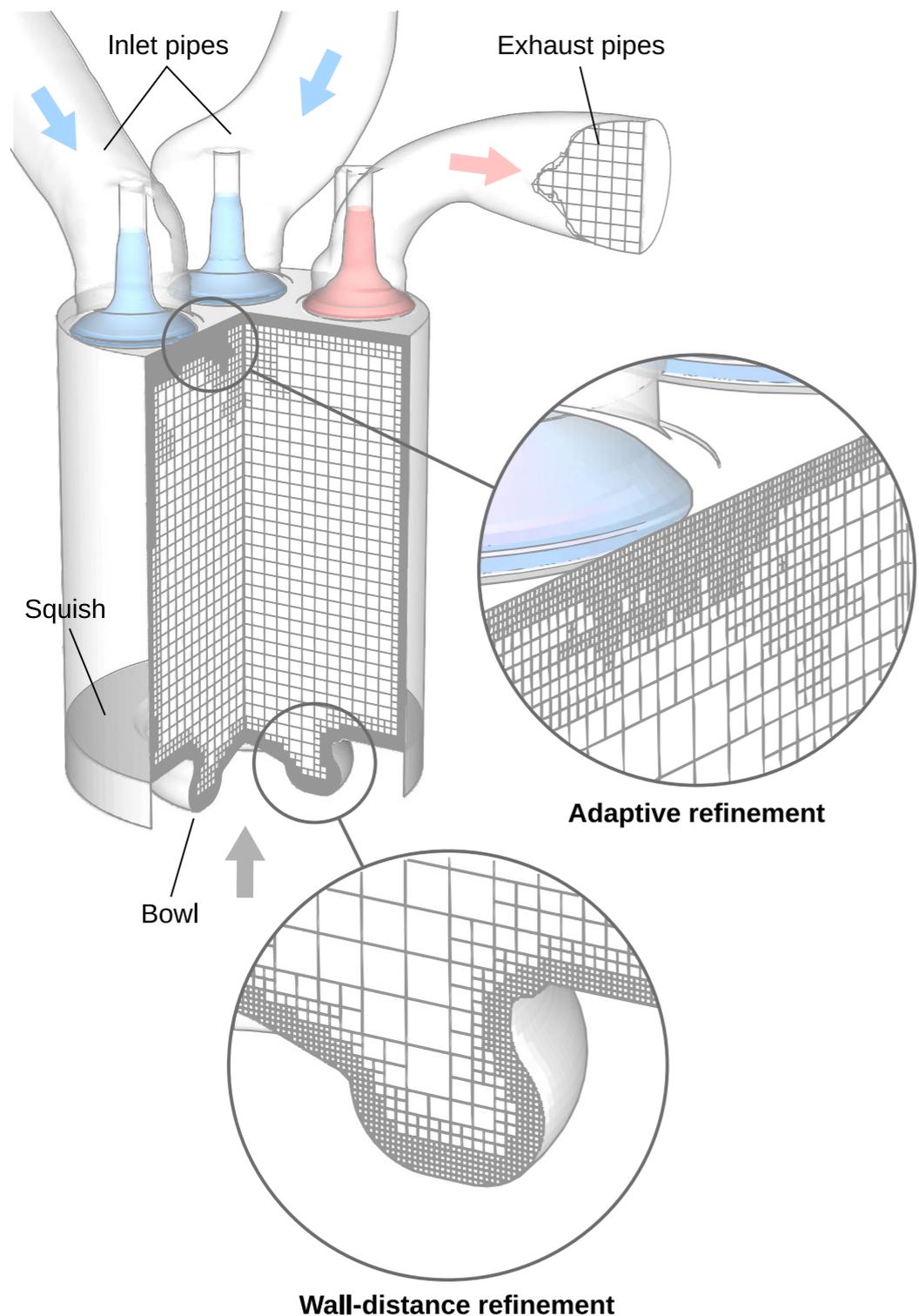
Modal decomposition of the unsteady flow field in compression-ignited combustion chambers

*A. Torregrosa, A. Broatch, J. García-Tíscar, J. Gomez-Soriano*

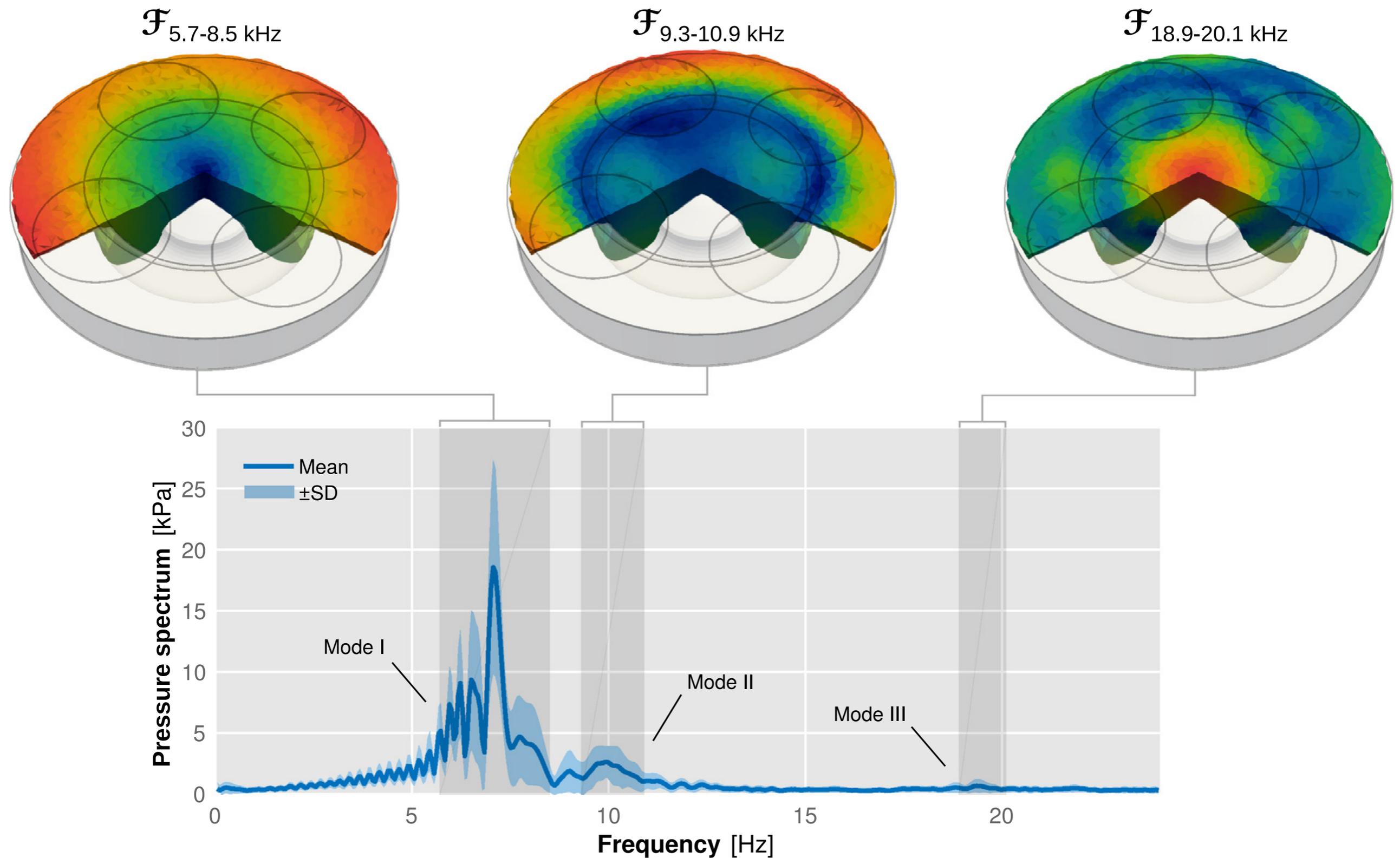
Combustion and Flame 188 (2018) 469–482



# Pressure inside the engine has a broadband frequency content.



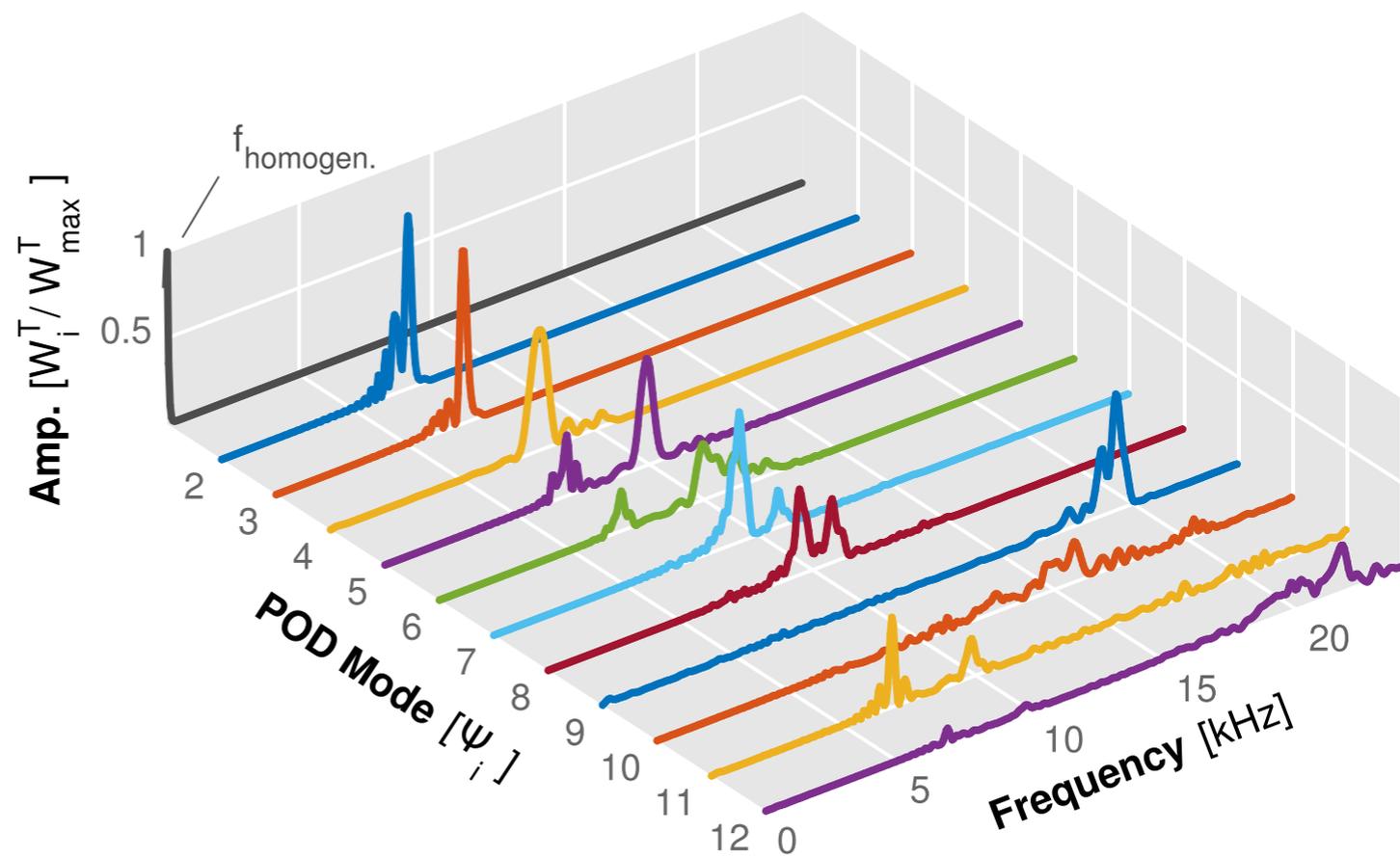
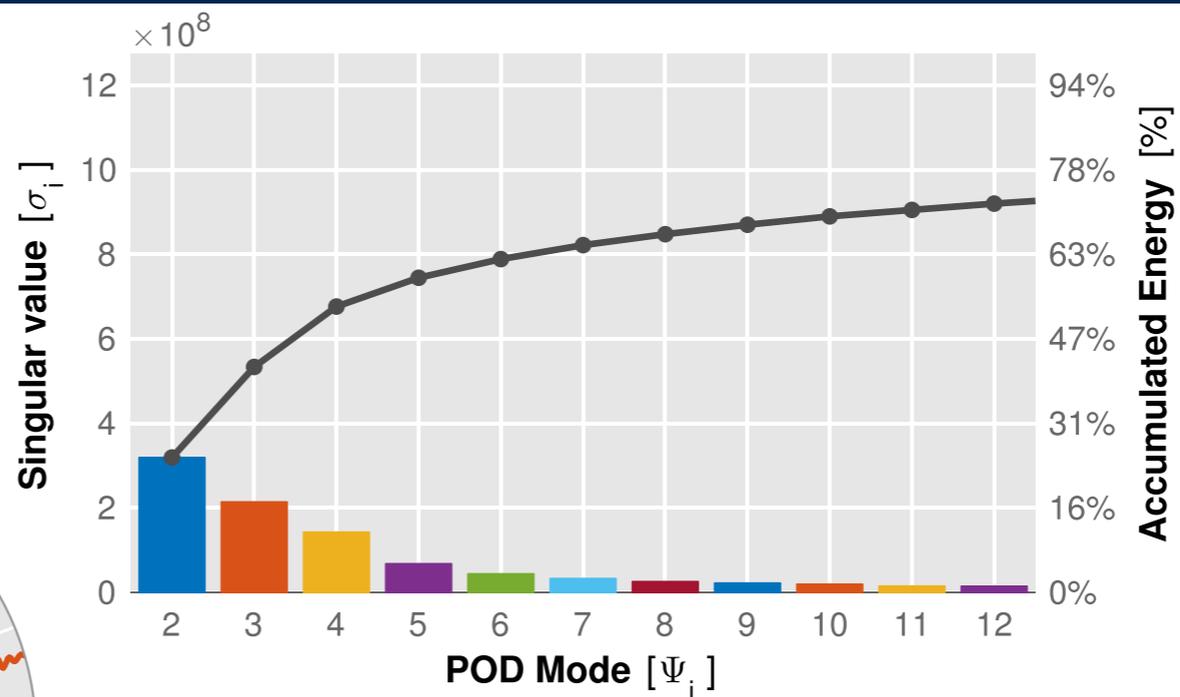
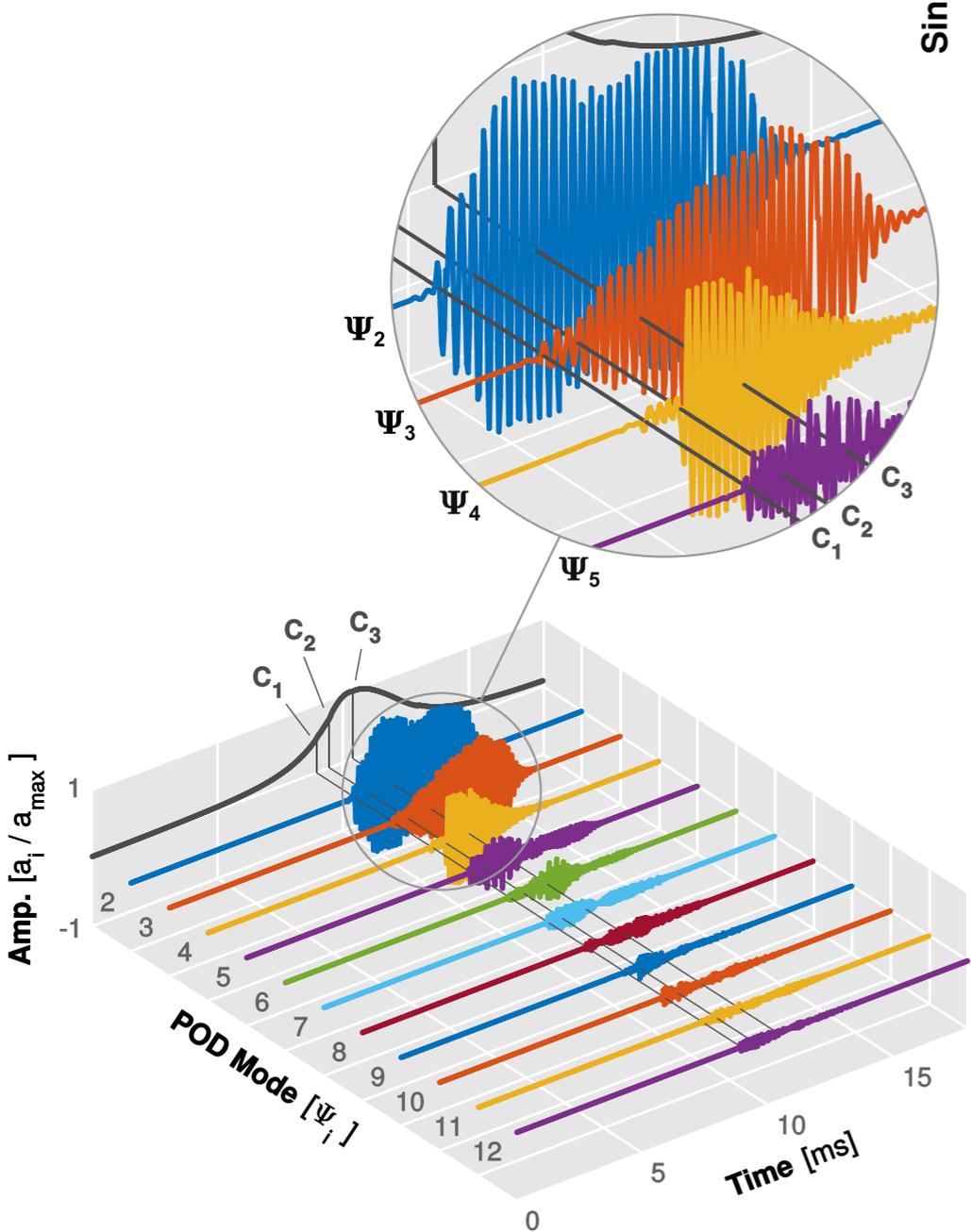
# Several Power Density modes have to be collected to describe the spatial pressure evolution



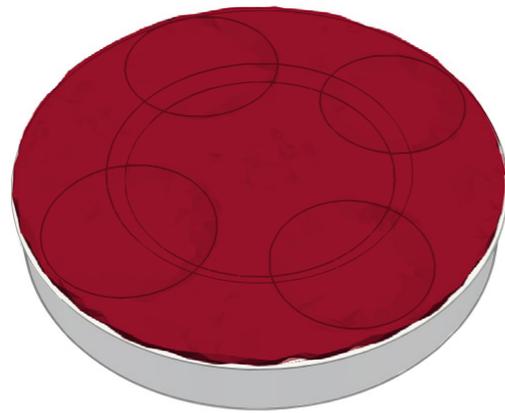
The signal is not periodic.

# A reduced number of POD modes concentrate most of the energy.

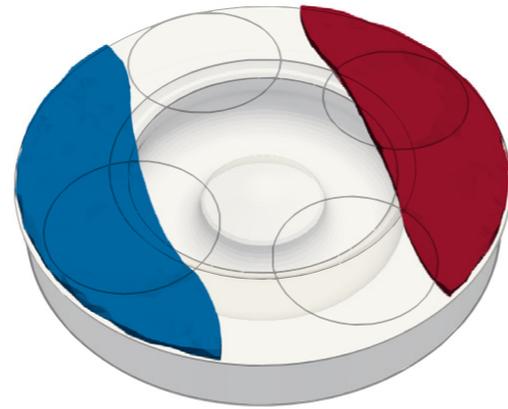
$$\mathbf{V}(\mathbf{x}, t) = \sum_{i=1}^N \boldsymbol{\Psi}_i(\mathbf{x}) \mathbf{a}_i(t)$$



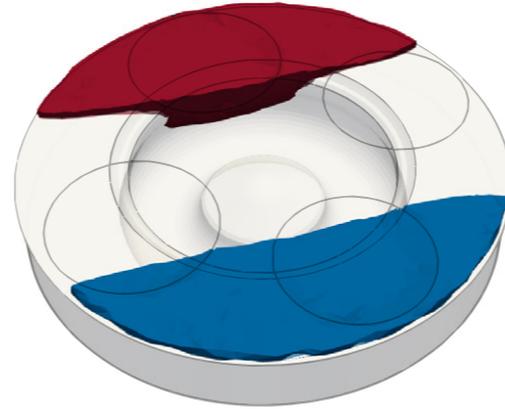
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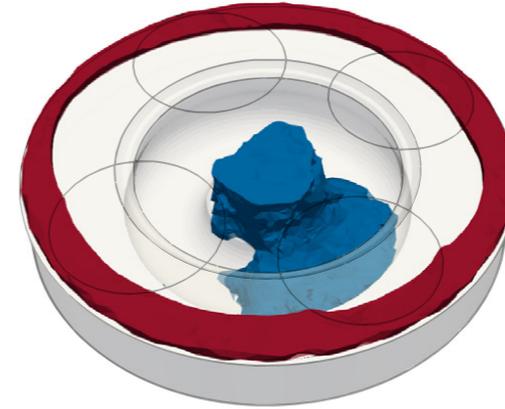
$\Psi_1$



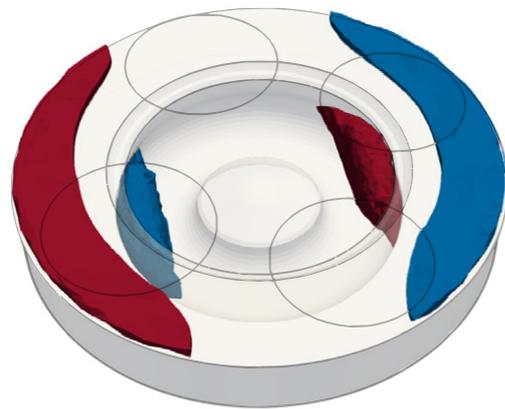
$\Psi_2$



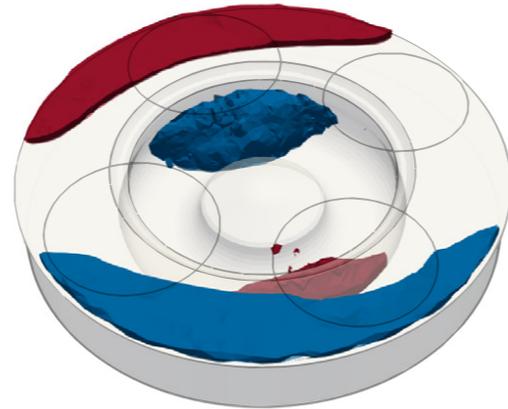
$\Psi_3$



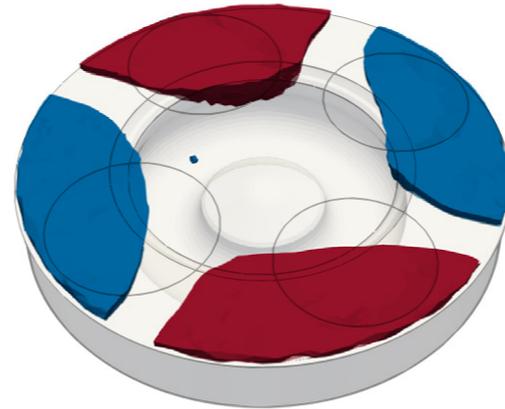
$\Psi_4$



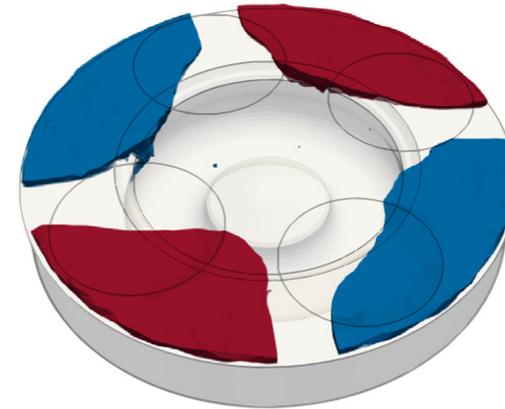
$\Psi_5$



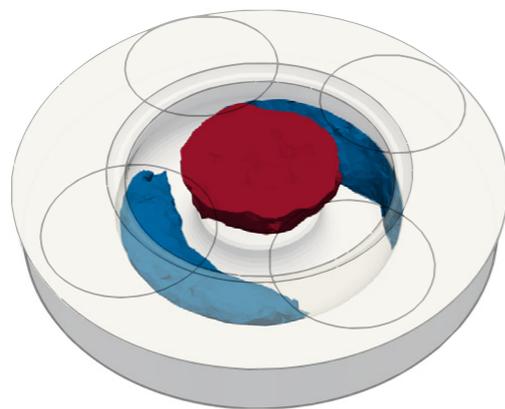
$\Psi_6$



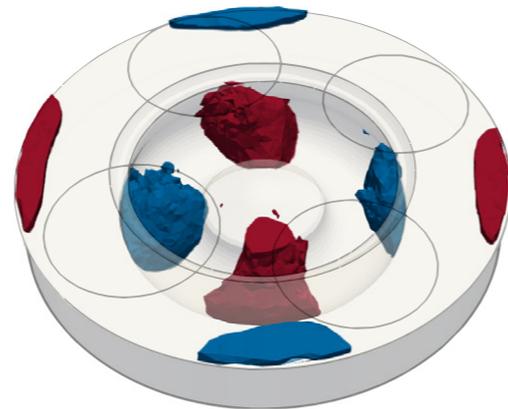
$\Psi_7$



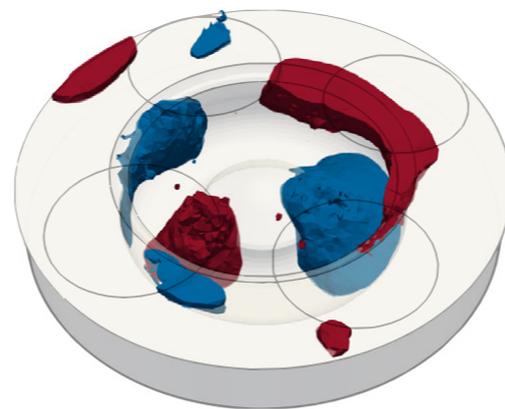
$\Psi_8$



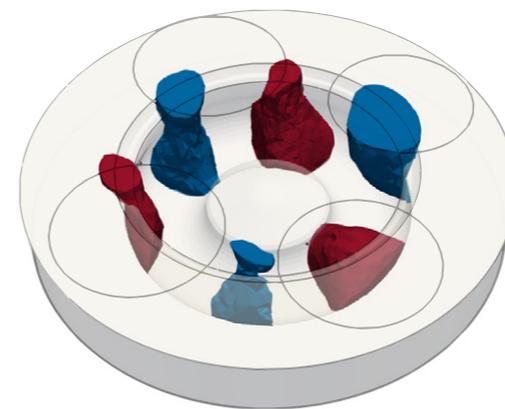
$\Psi_9$



$\Psi_{10}$

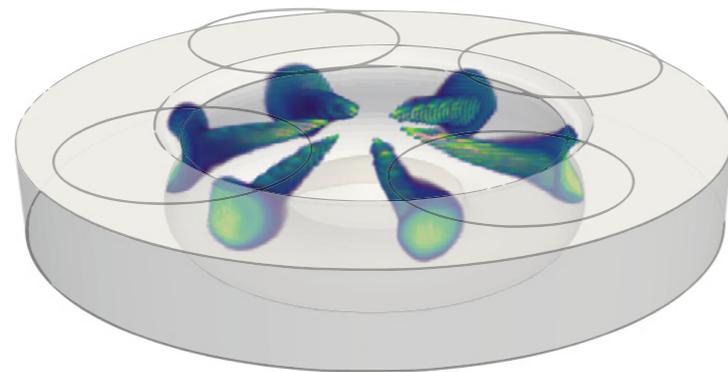


$\Psi_{11}$

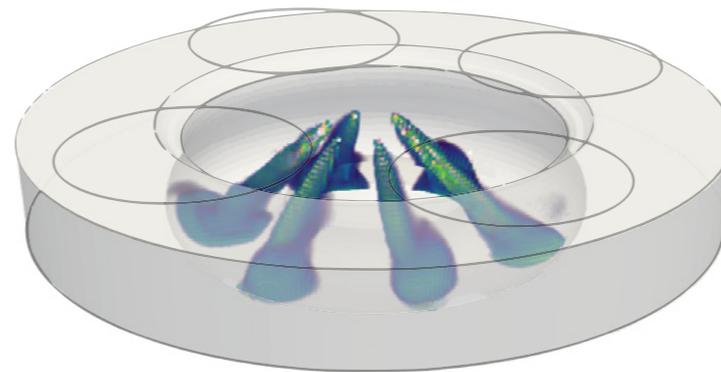


$\Psi_{12}$

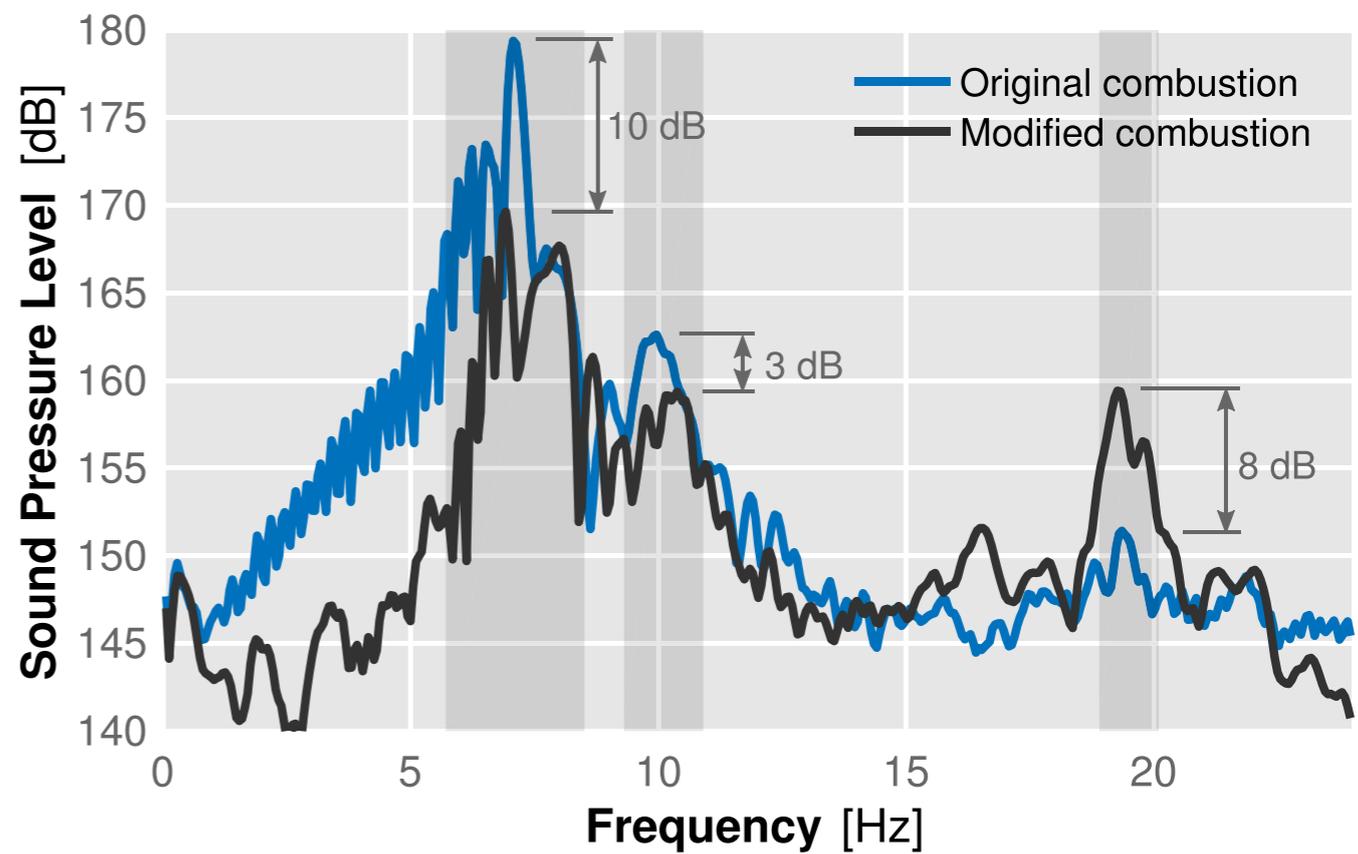
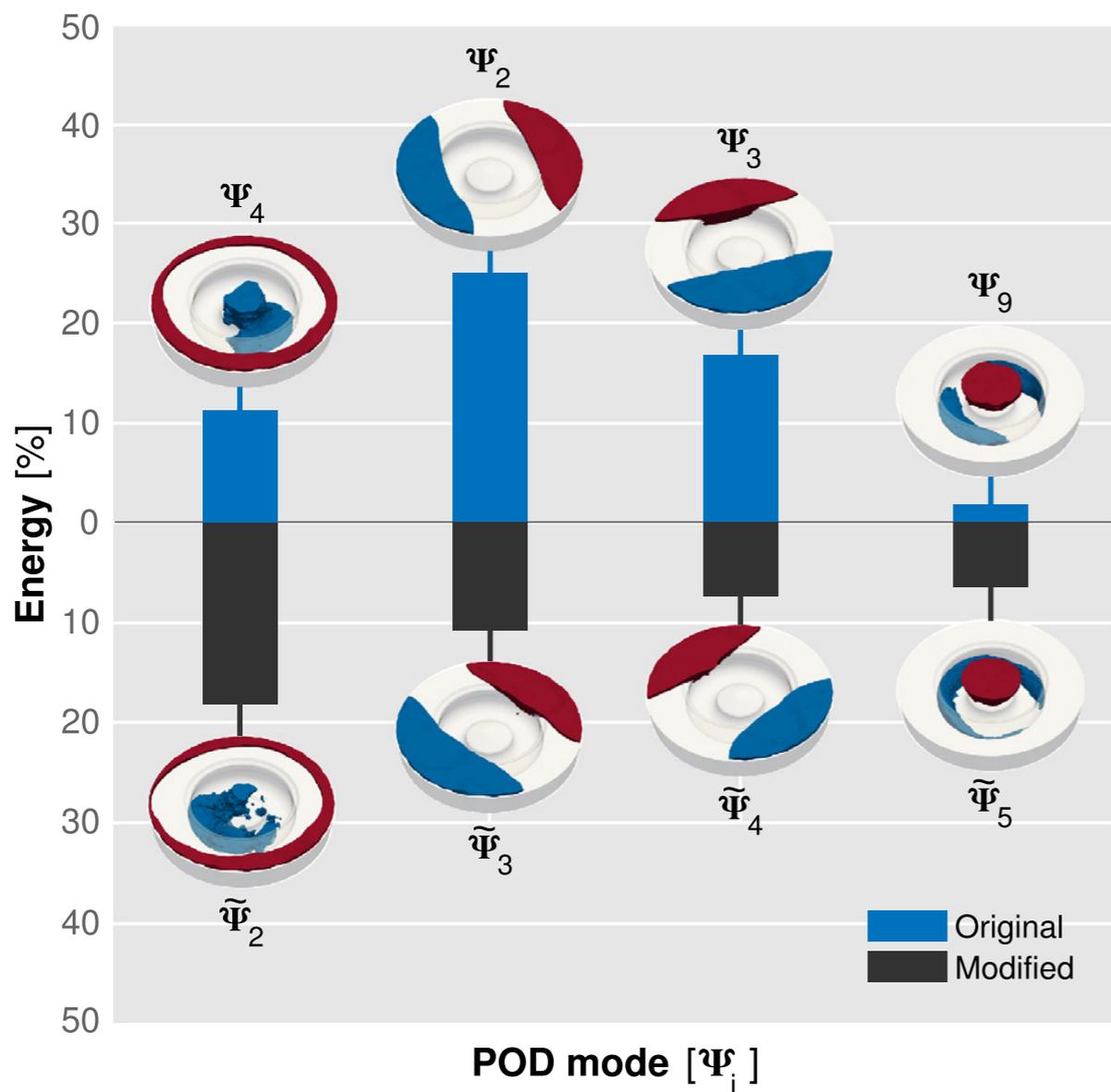
# A geometrical modification of the injection reduces the pressure amplitude fluctuations



Original [150°]



Modified [90°]



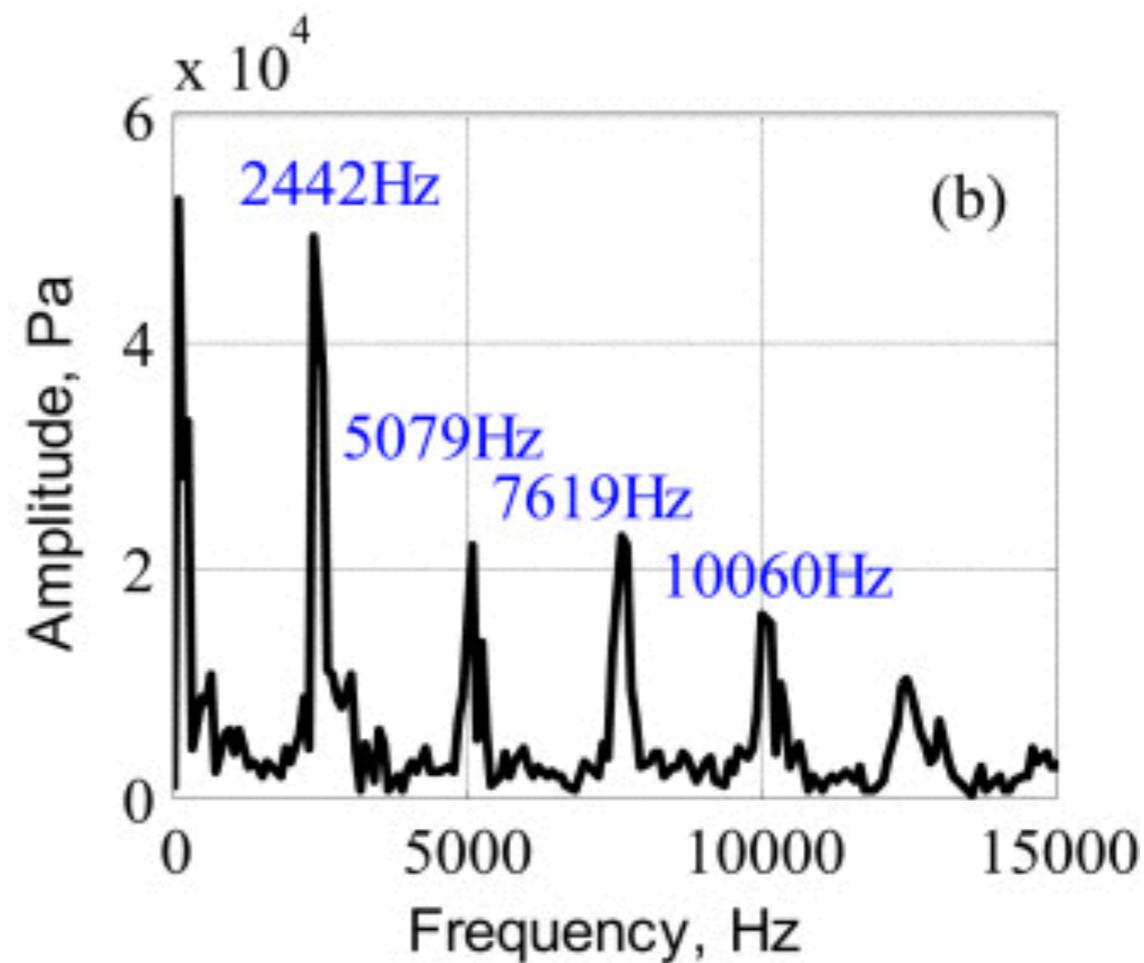
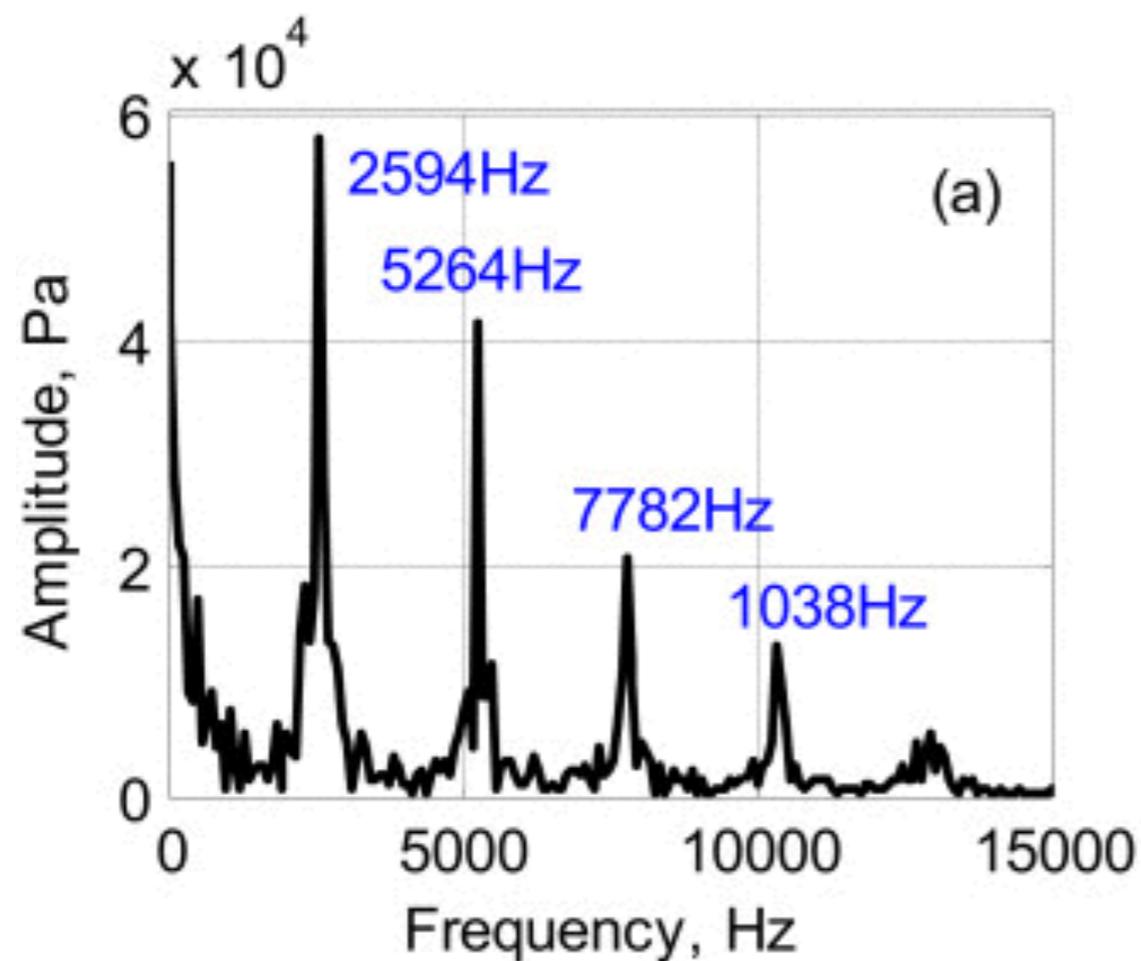
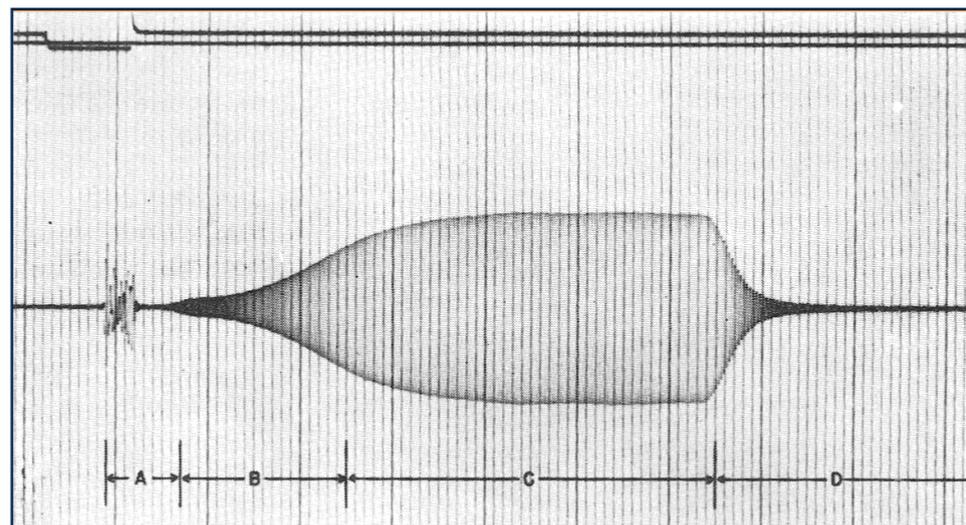
# POD used on numerical simulations

POD is efficient when

- a few modes concentrate a large amount of energy
- the energy is distributed over a large range of frequencies
- the signal is not periodic
- the time resolution is low

DMD cannot be used to decompose transition signals and require a high time resolution.

# Limit cycles observed in combustion chamber feature



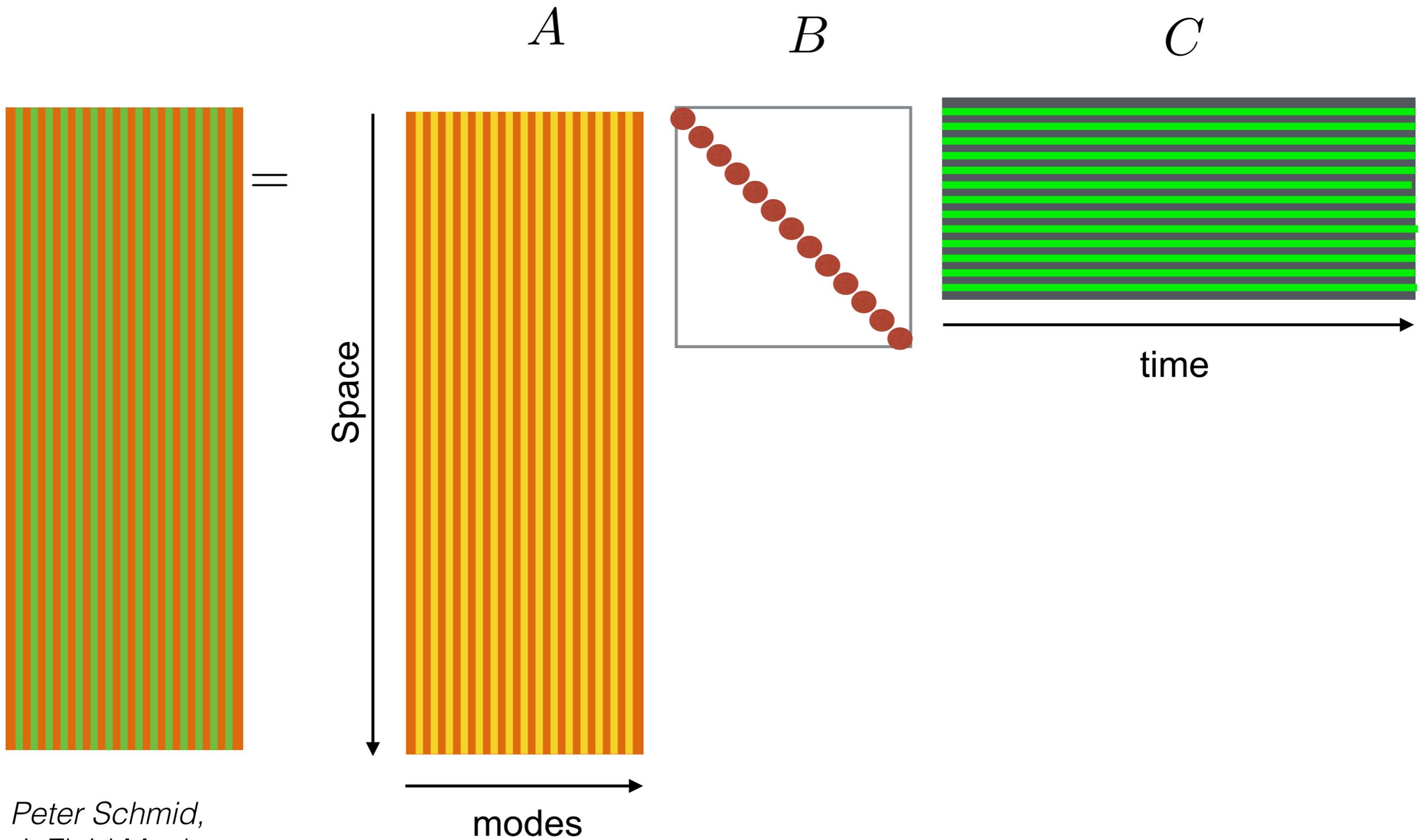
Computational investigation on combustion instabilities in a rocket combustor

Acta Astronautica

Volume 127, October–November 2016, Pages 634-643

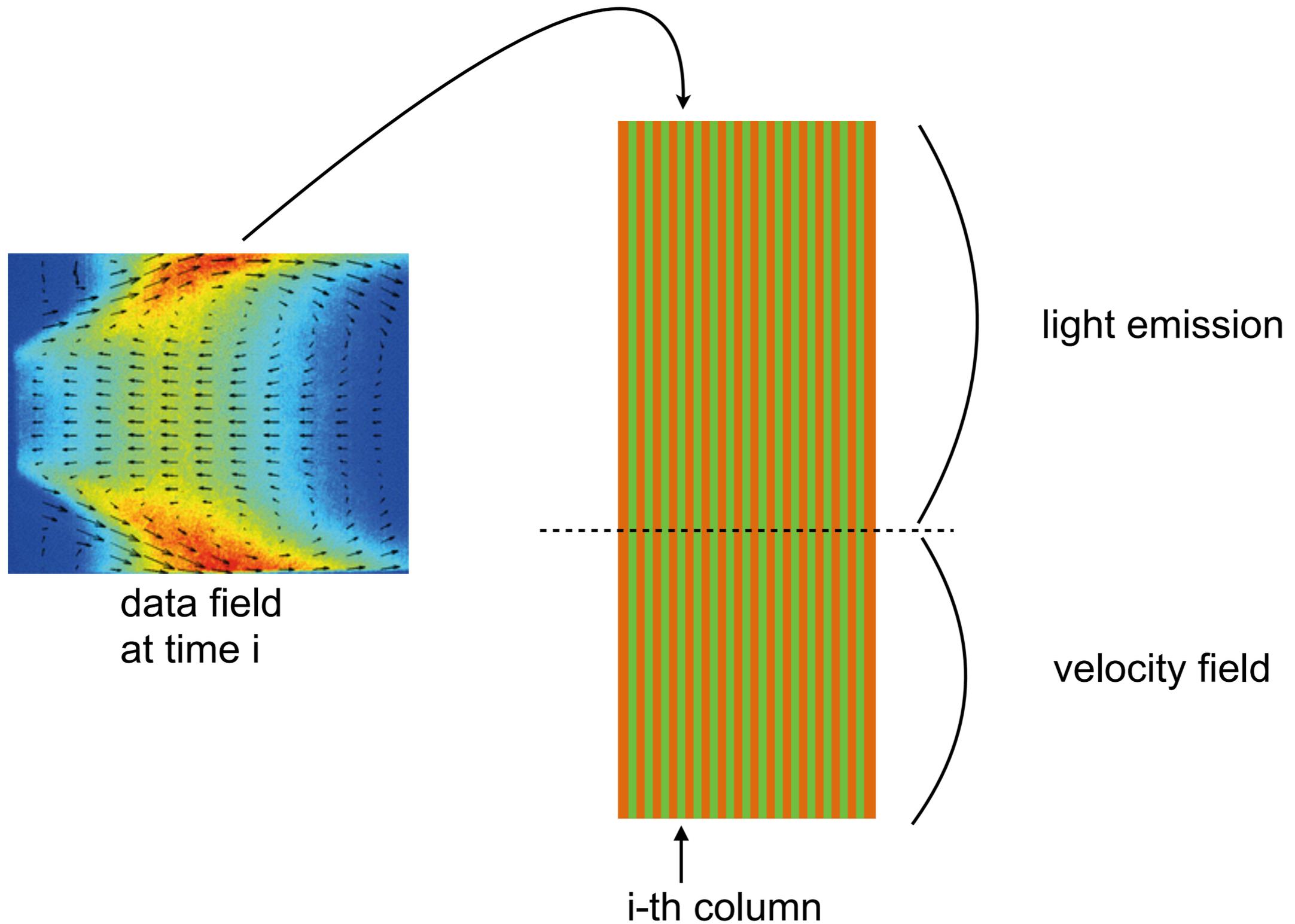
LeiYuan, Chibing Shen

DMD generates a decomposition of the data based on the frequencies.

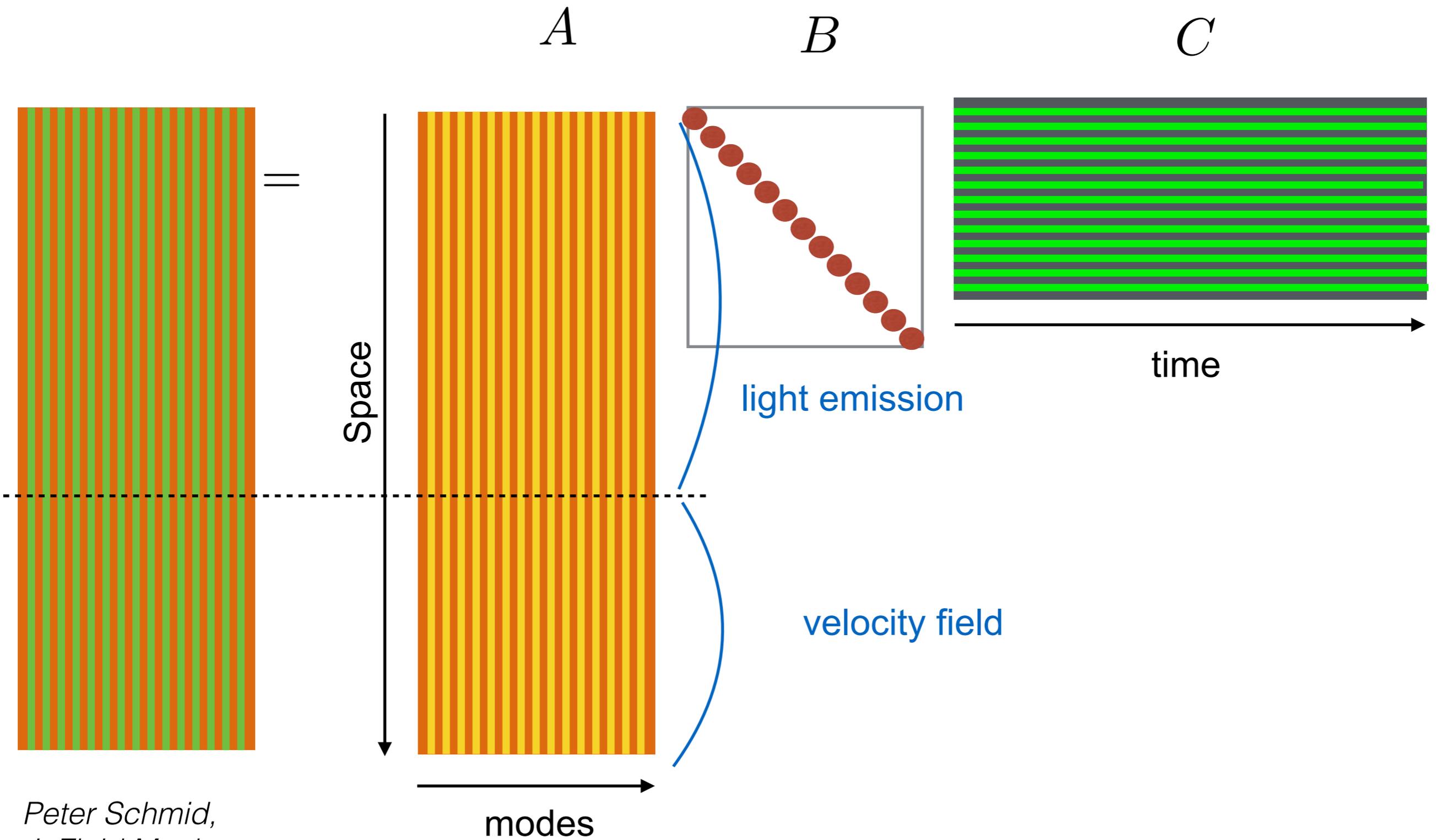


*Peter Schmid,  
J. Fluid Mech.  
(2010)*

# Multi-variables DMD

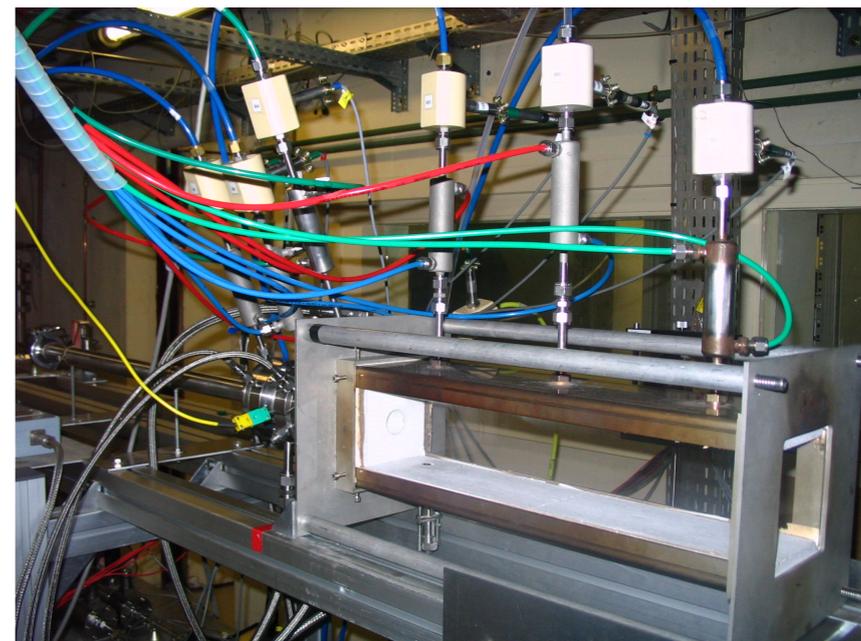
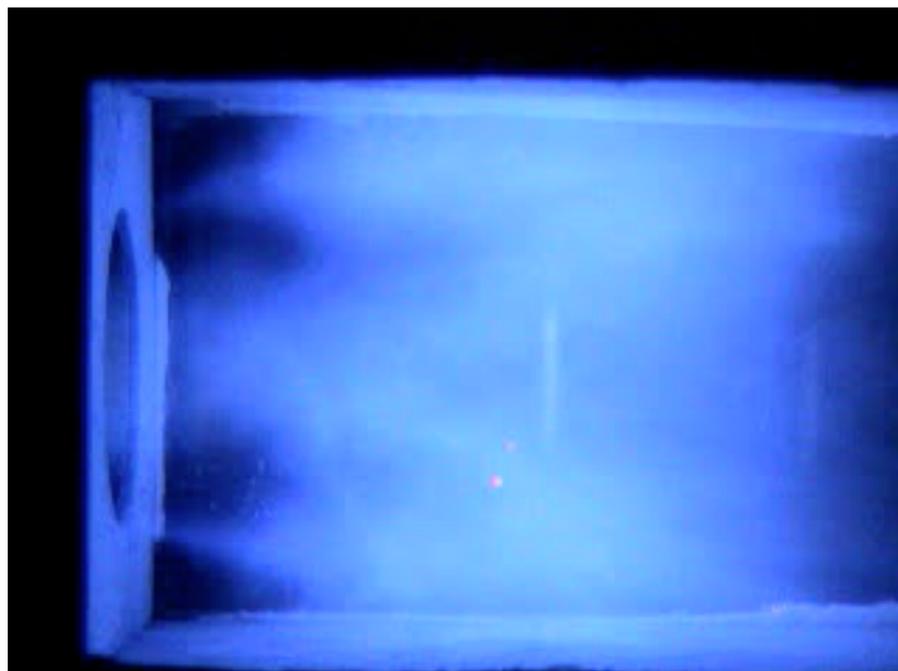
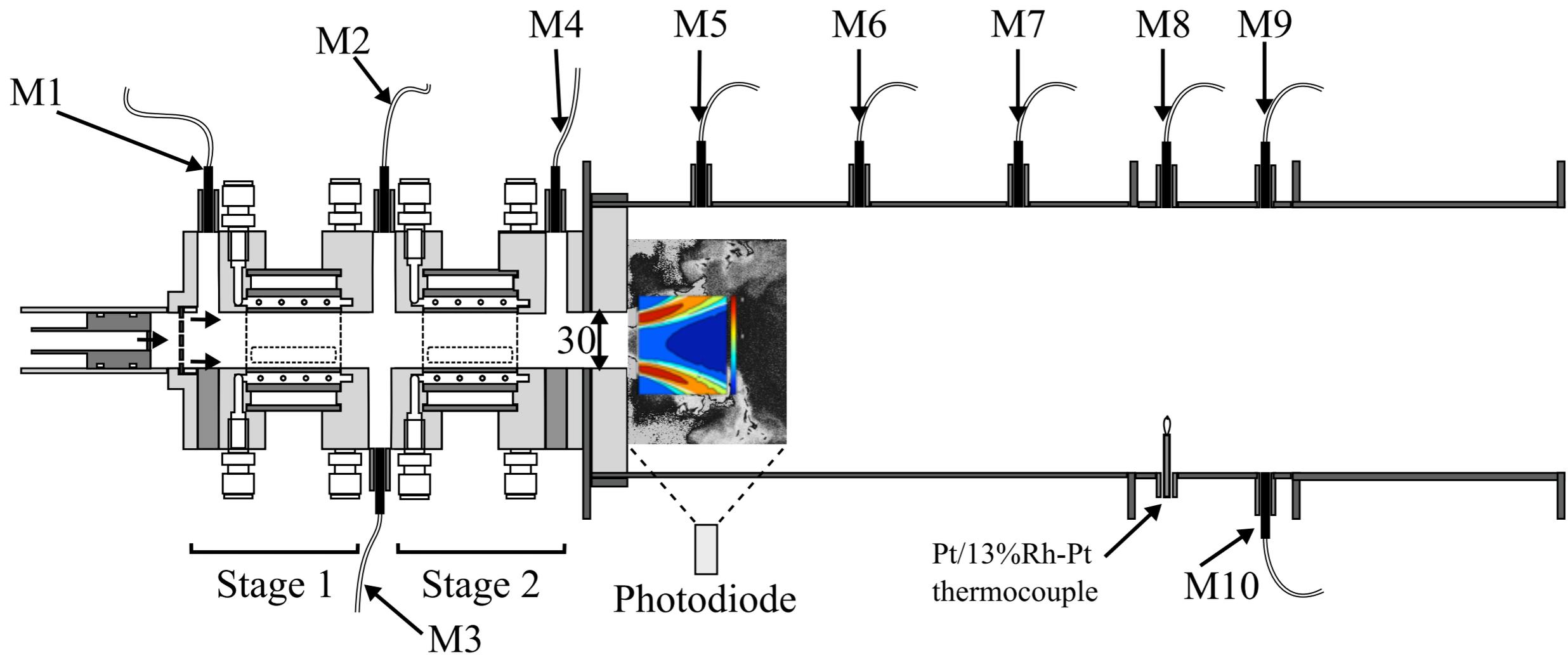


# DMD generates a decomposition of the data based on the frequencies.

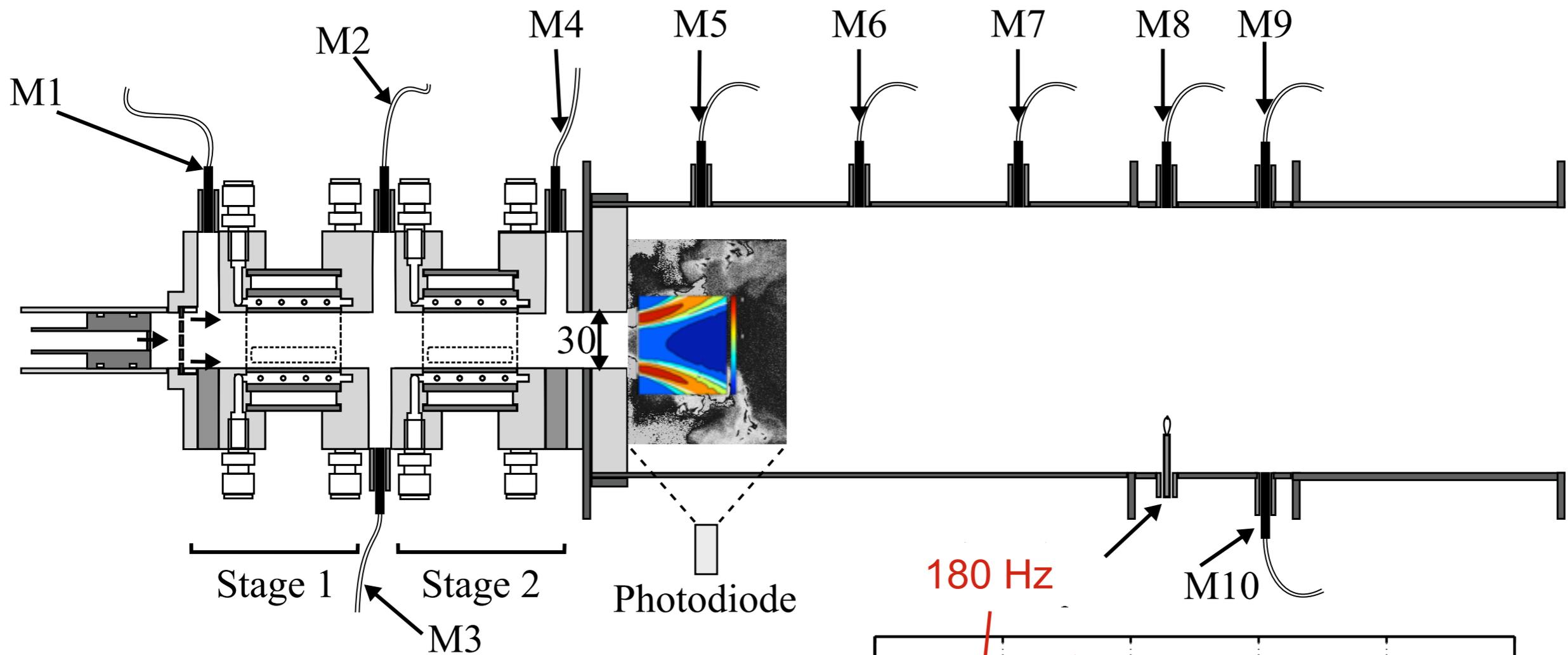


*Peter Schmid,  
J. Fluid Mech.  
(2010)*

# CESAM Experimental test bench reproduce most of the coupling phenomena encountered in aeronautical combustion chamber

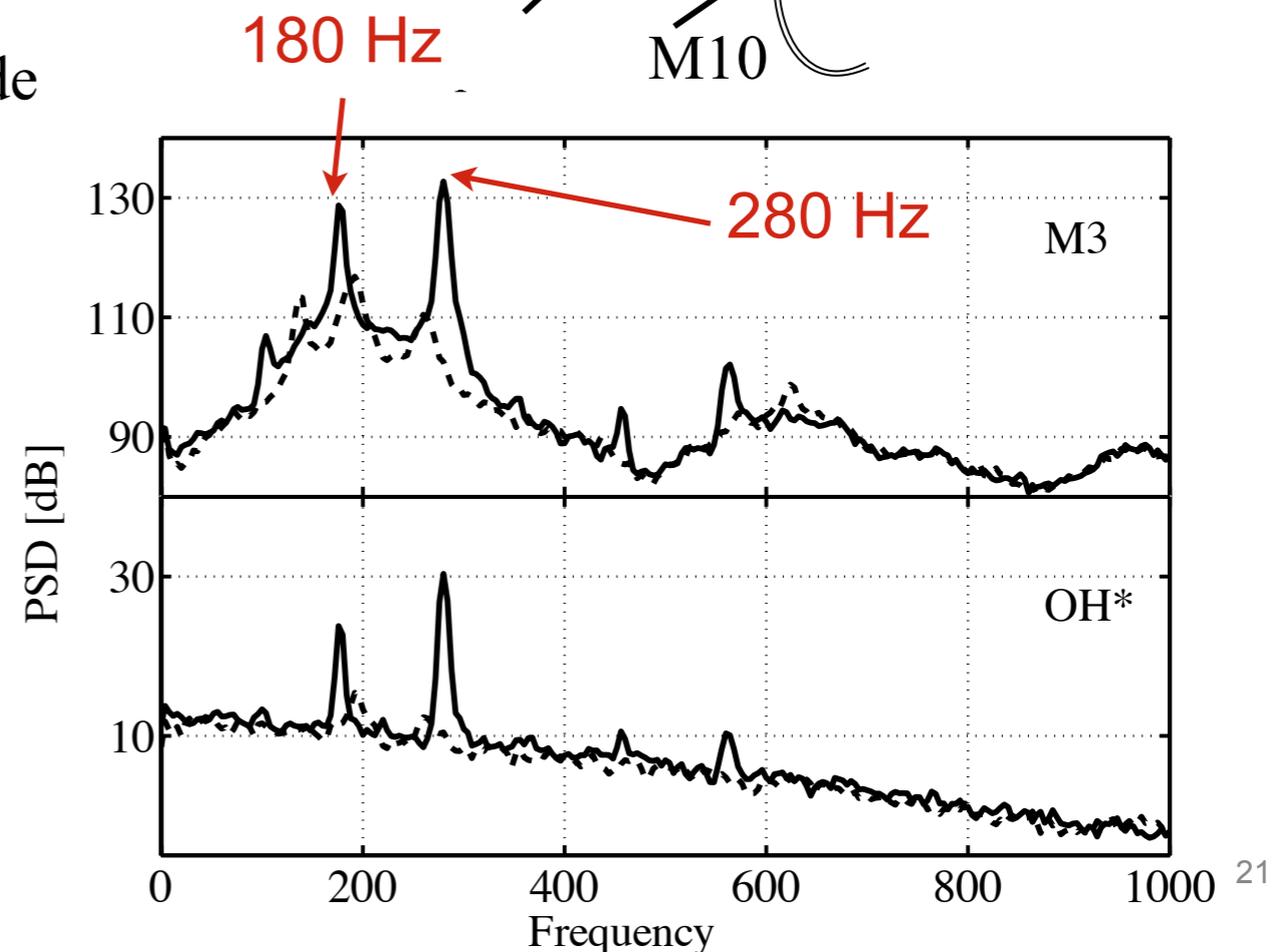


# CESAM Experimental test bench reproduce most of the coupling phenomena encountered in aeronautical combustion chamber



## Simultaneous acquisition

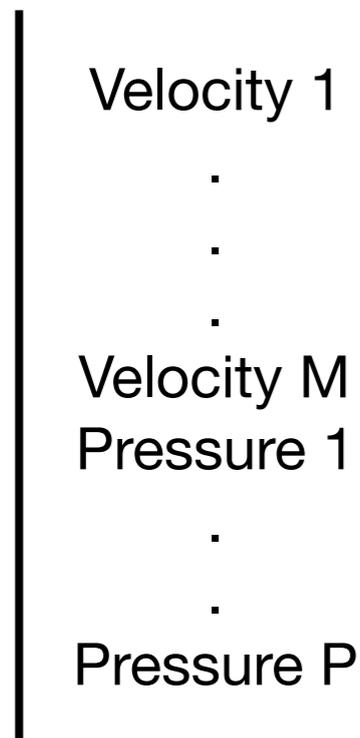
- PIV @15kHz
- 10 microphones @16kHz
- OH\* et CH\* emission @16kHz



# Multi-variable DMD highlights correlation between physical quantities

Two strategies are still available.

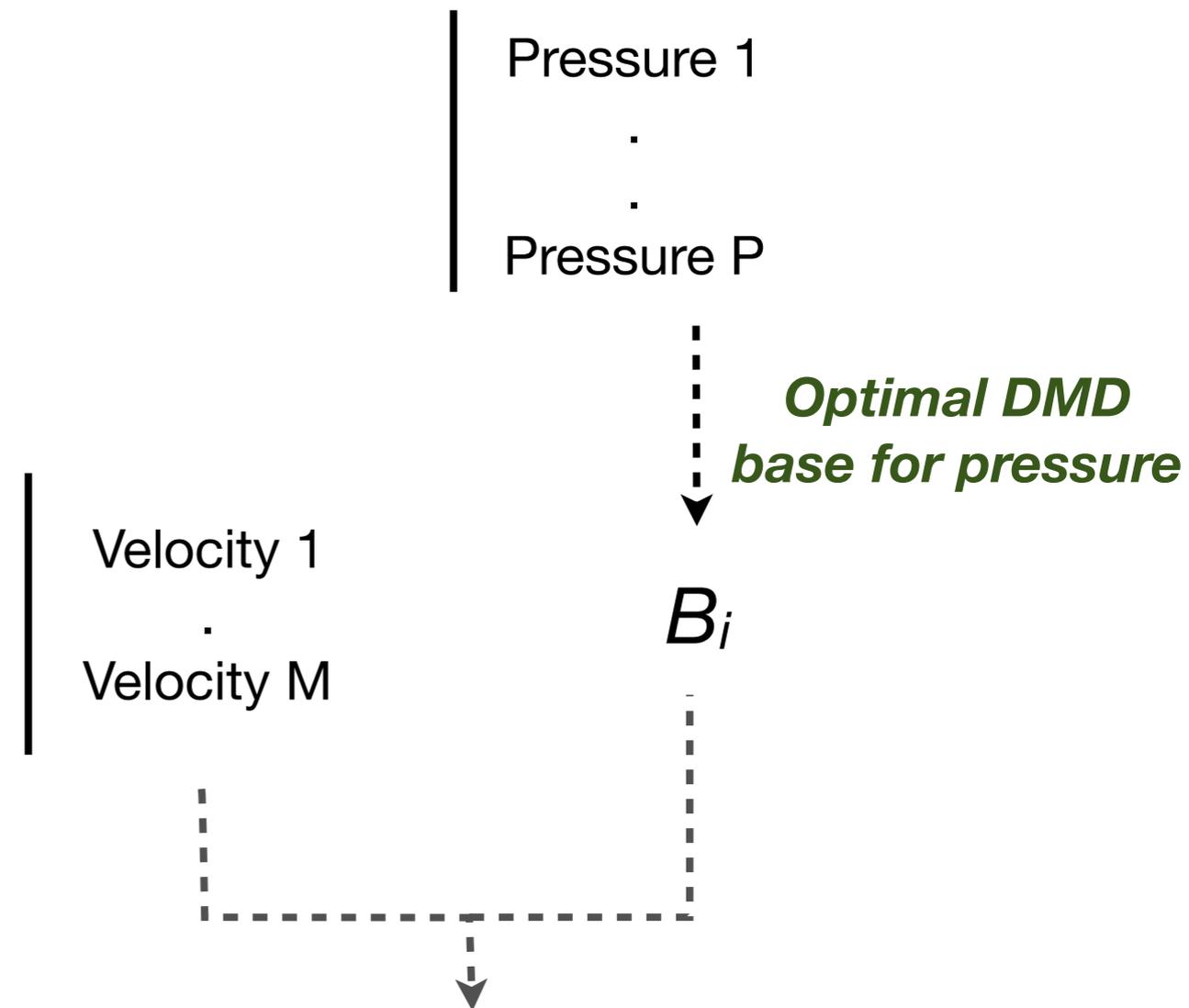
## 1 - Multi-variable DMD



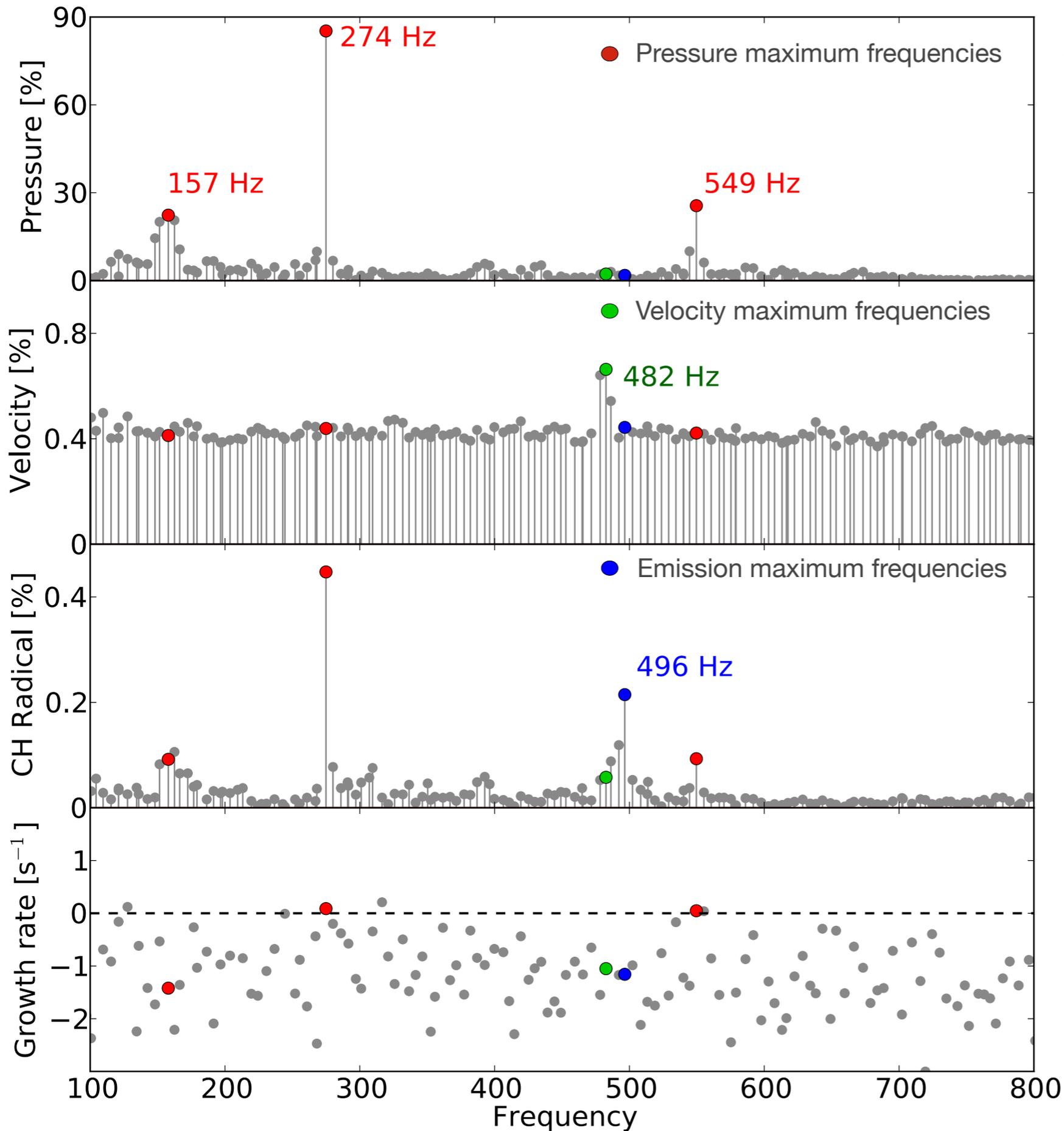
This gives a unique optimal base for all the diagnostics

Values have to be normalized to take into account their volume and their absolute value.

## 2 - Extended DMD

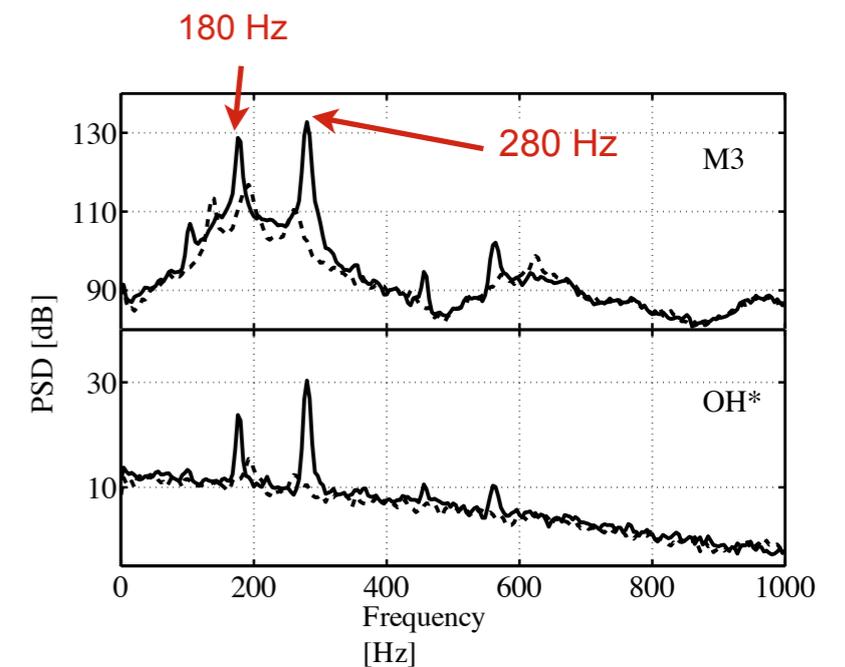


*The amplitude of velocity modes projected on the base  $B_i$  highlight the correlation between the two quantities.*



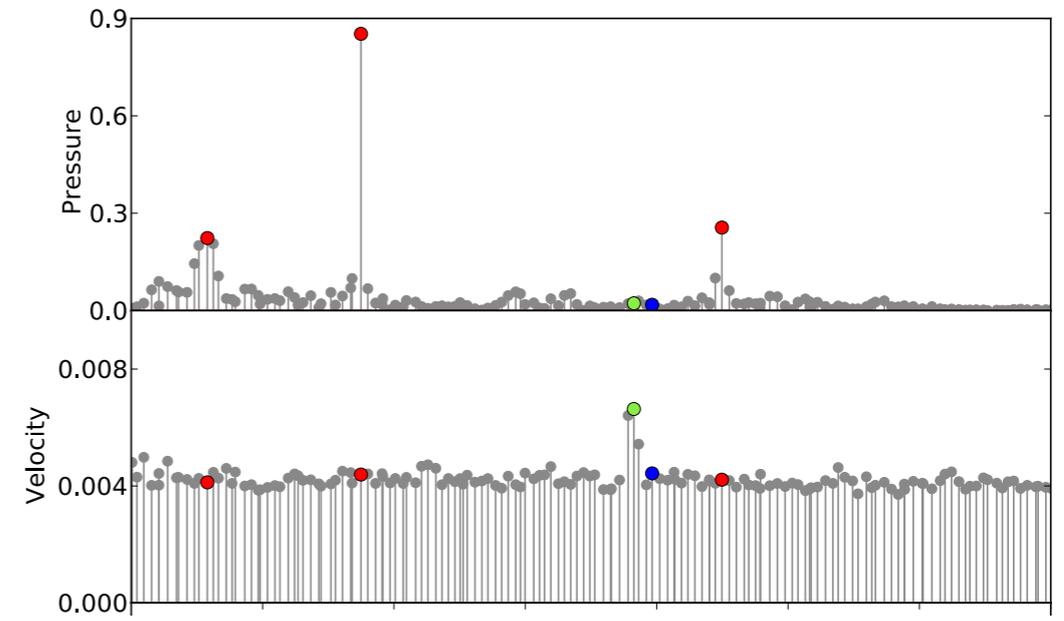
## Quantitative informations

- Velocity has a different dynamic than pressure
- Heat release is sensitive to both the pressure and velocity fluctuations
- Damping strongly depends on the modes



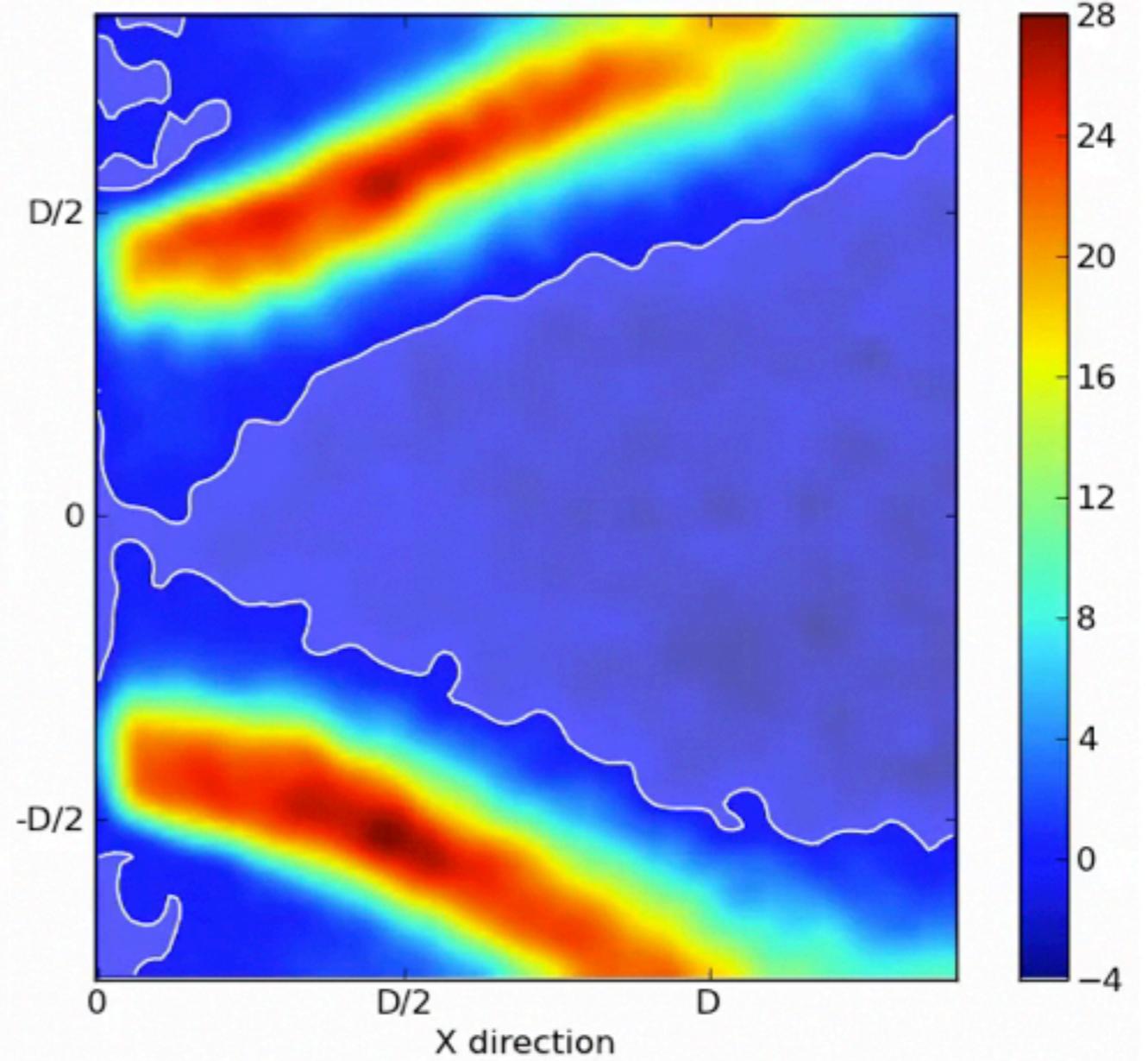
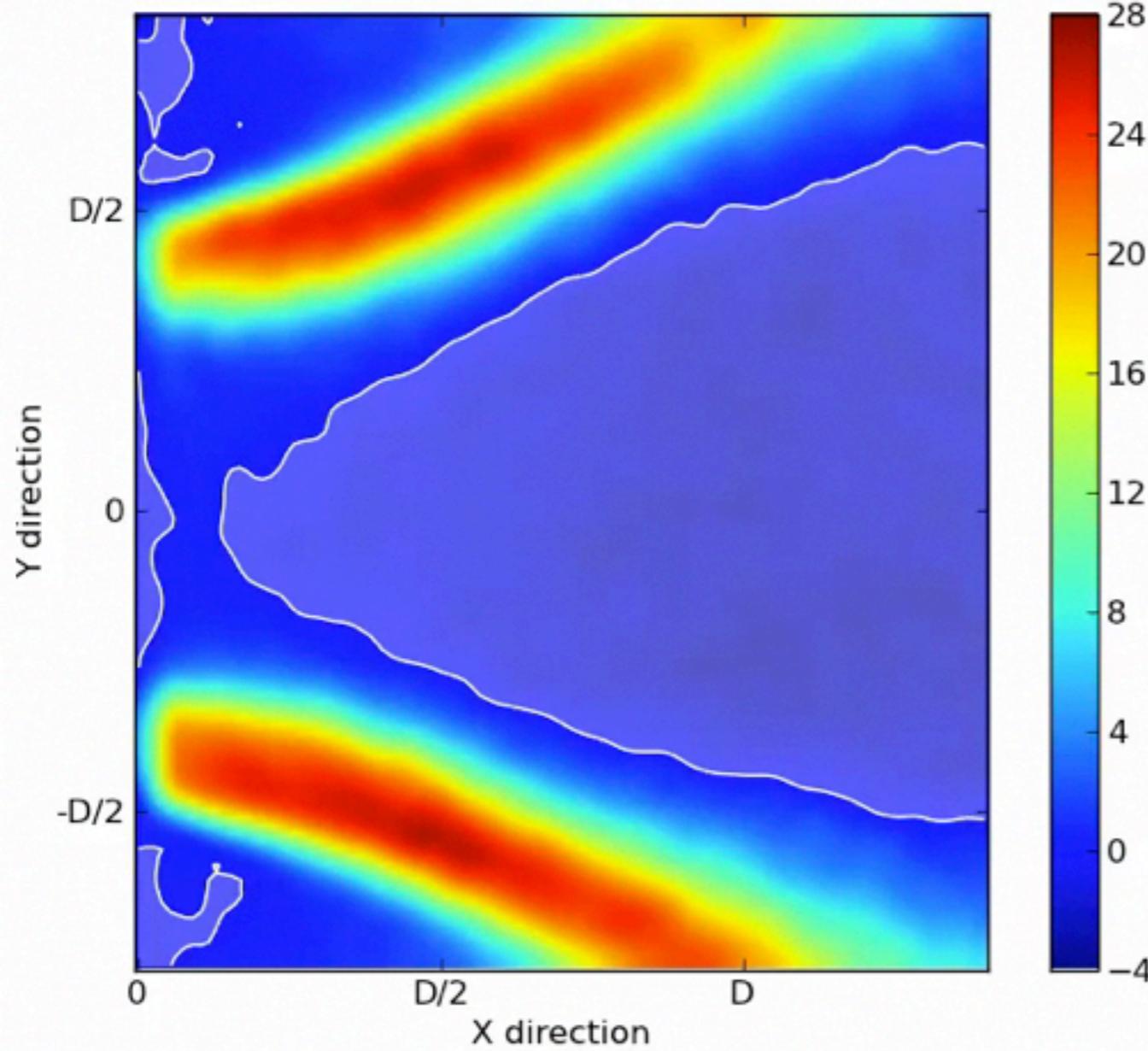
# Flow dynamics

DMD highlight coupling between acoustics and flow.



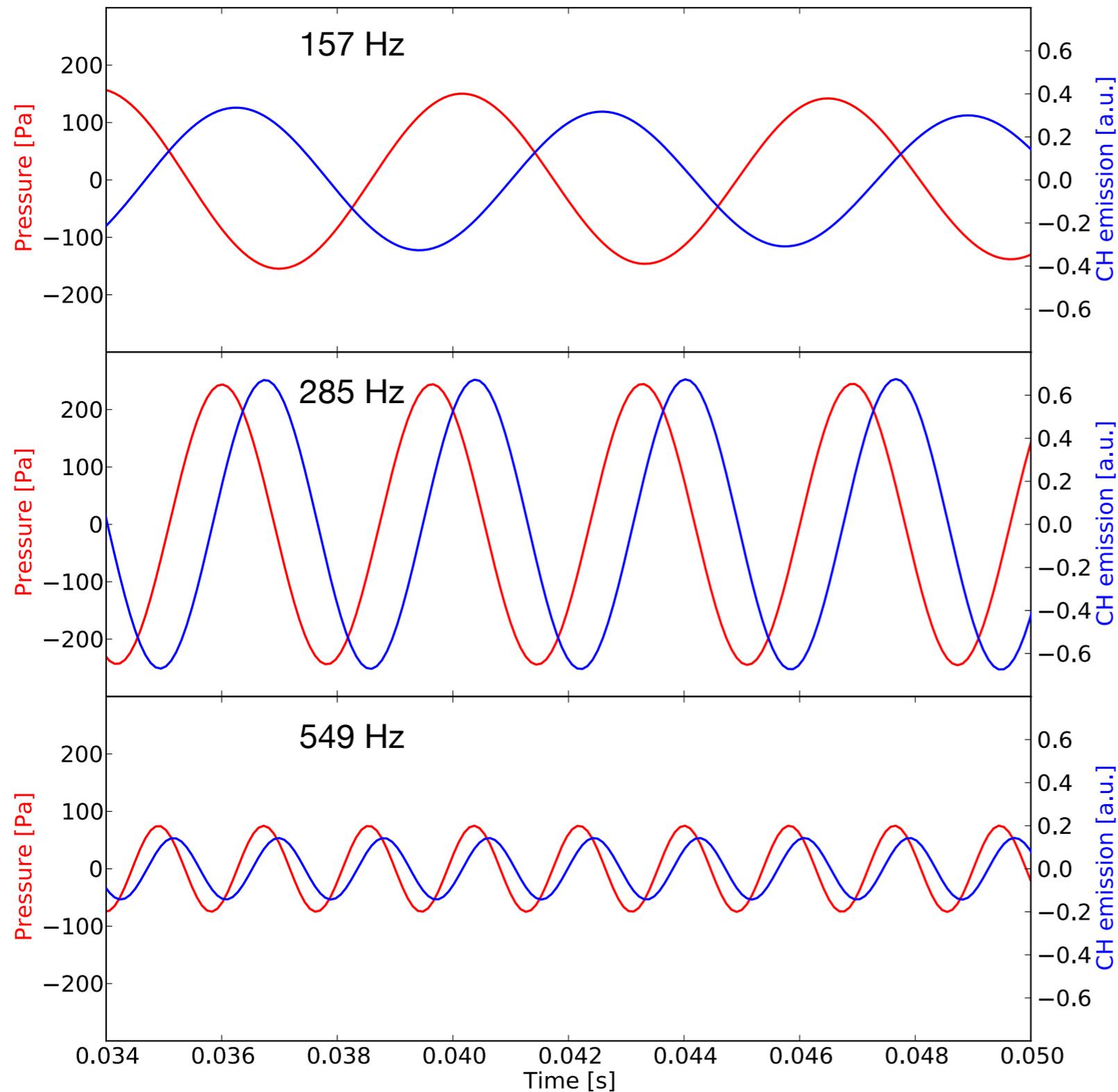
275 Hz

482 Hz



# Reconstruction of the acoustic source

The phase between the diagnostics is recovered within the modes.

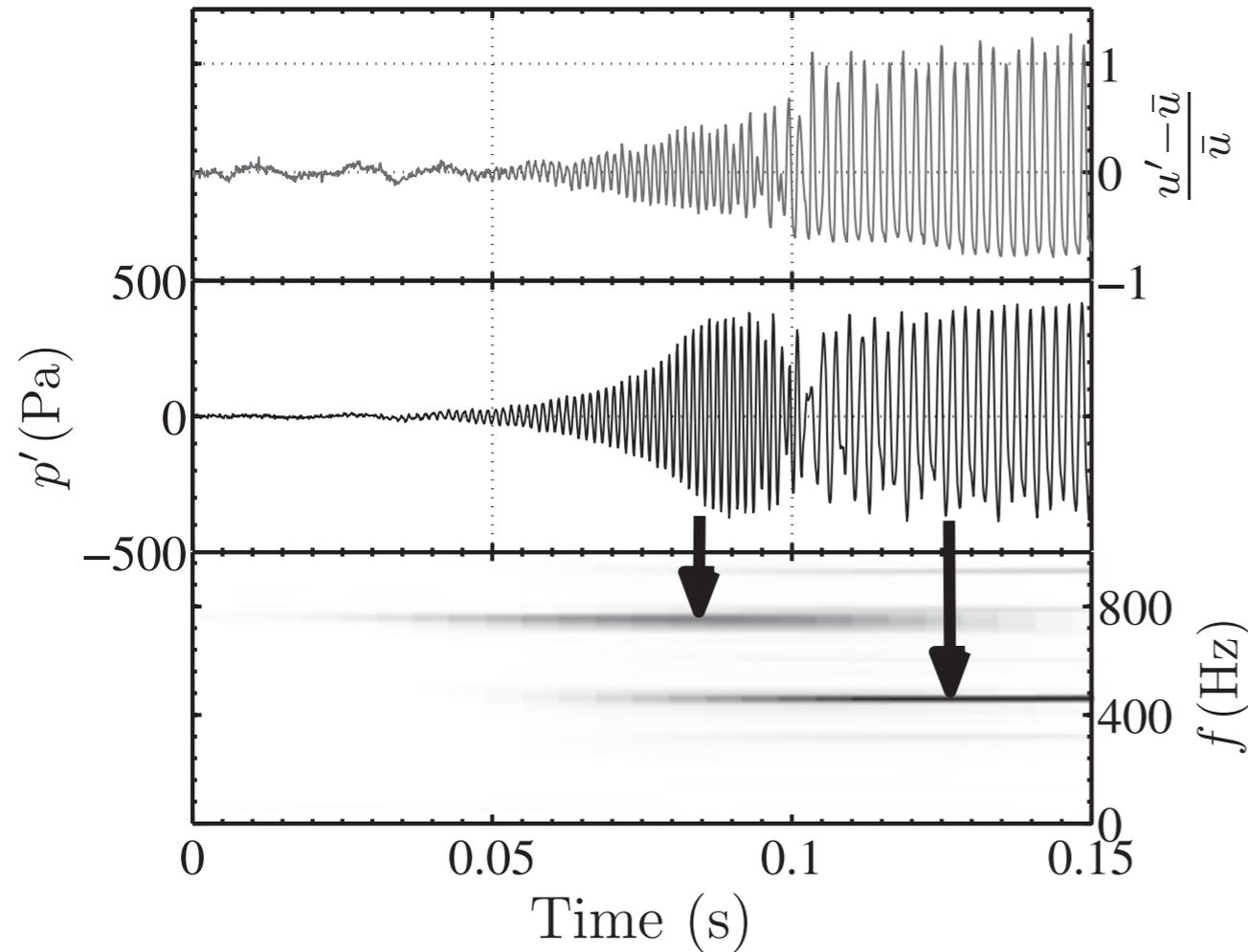


Acoustic energy

$$\frac{\partial e}{\partial t} + \nabla \cdot (p_1 u_1) = \frac{\gamma - 1}{\gamma p_0} p_1 \dot{q}_1$$

Acoustic source term

# Data decomposition for model reduction



F. Boudy et al. / Proceedings of the Combustion  
Institute 33 (2011) 1121–1128

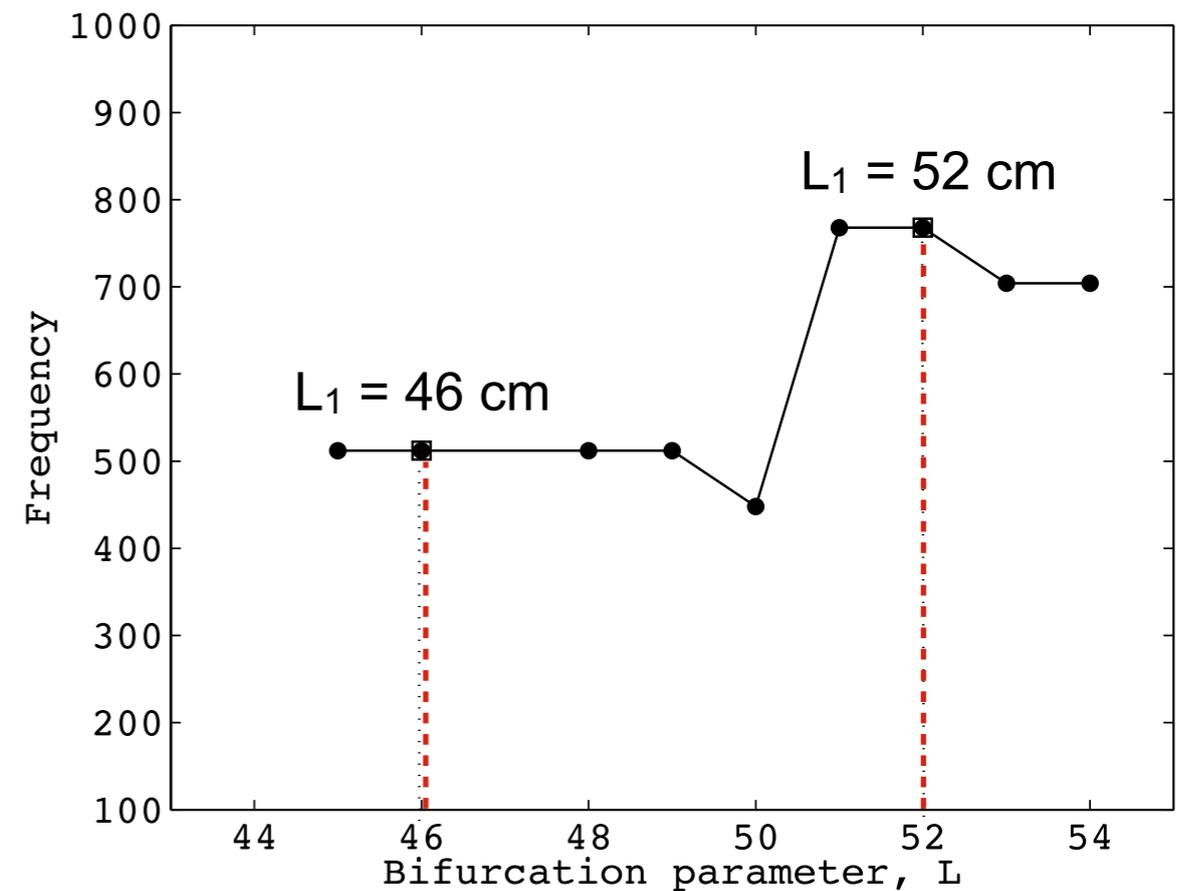
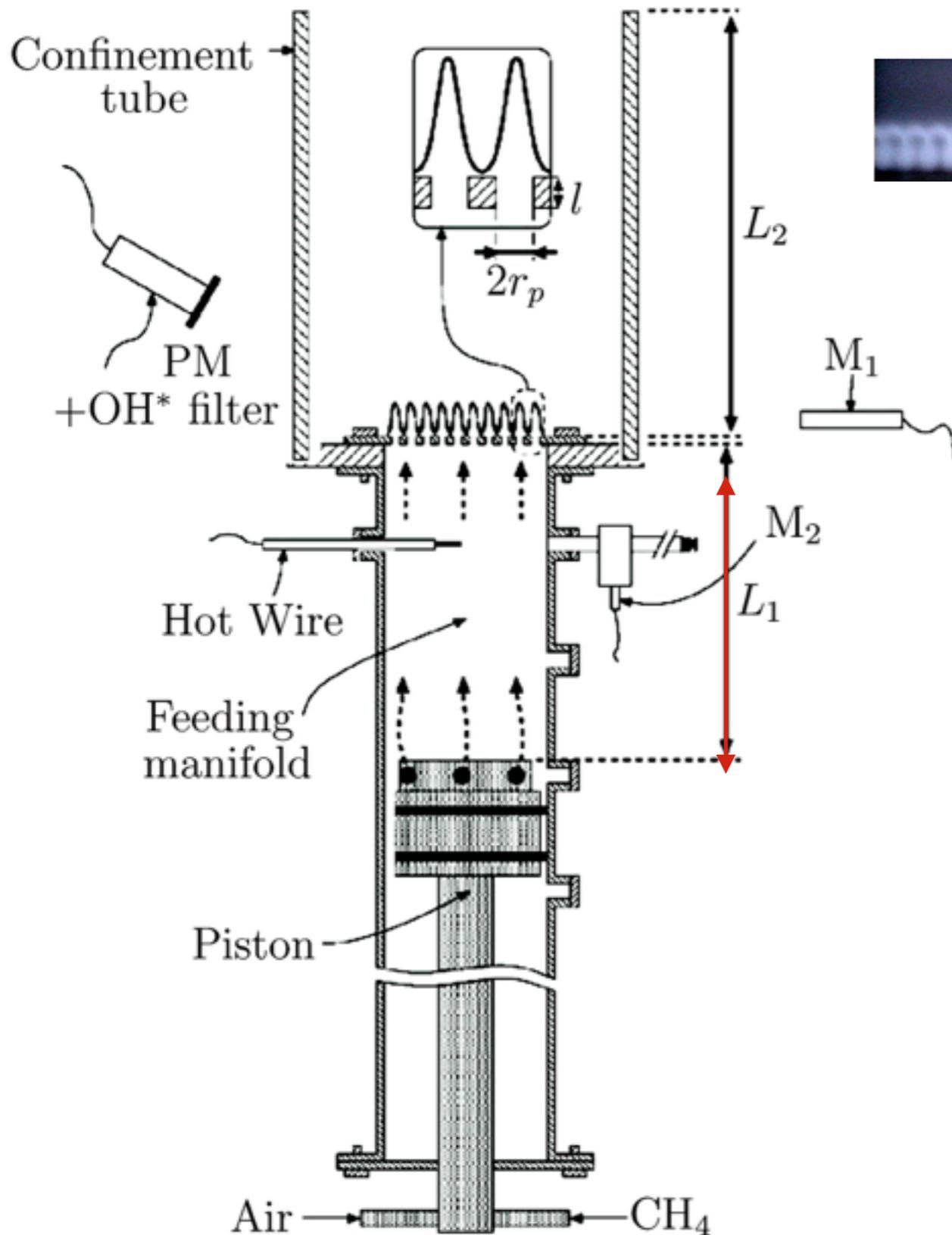
To create models, the strategy is to separate the signal in few modes and amplitudes :

$$p'(t, x) = a_1(t)\Psi_1(x) + \dots + a_4(t)\Psi_4(x)$$

$$\frac{da_i(t)}{dt} = \mathcal{F}(a_i)$$

The problem is to identify the modes that allow to describe the system whatever the operating point.

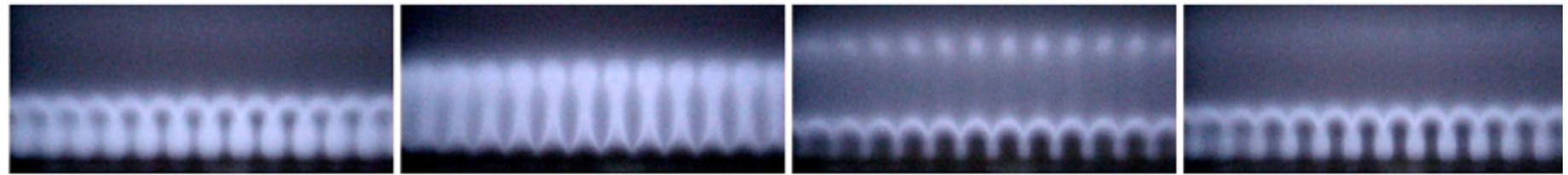
# Combustion systems experience bifurcation that make the modeling complex.



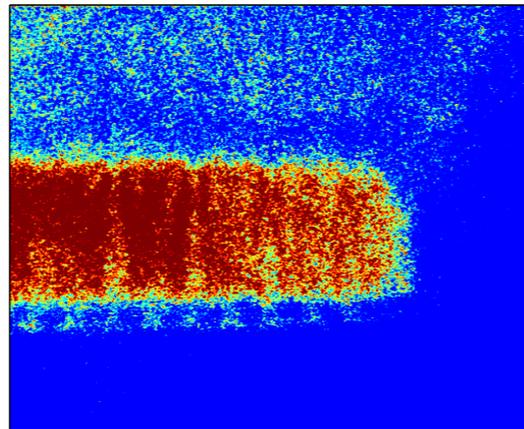
The system experience two kind of dynamics depending on the bifurcation parameter.

# Chemiluminescence from flames is recorded for different values of the bifurcation parameter.

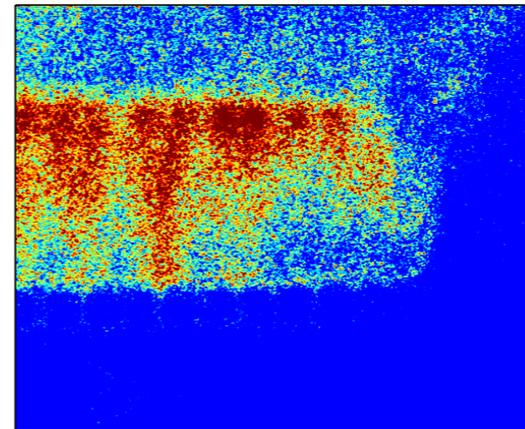
80 images, 6 kHz



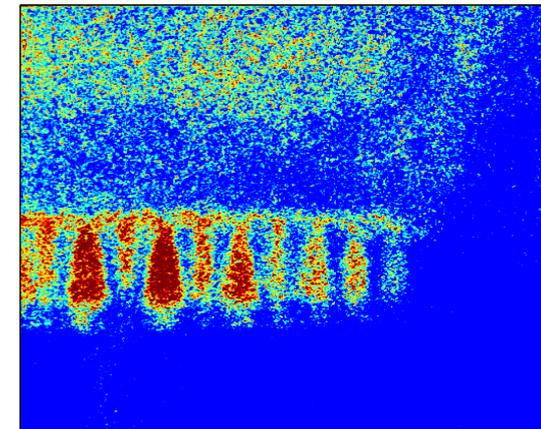
$$L_1 = 46 \text{ cm}$$



(a) Case A,  $t_1$

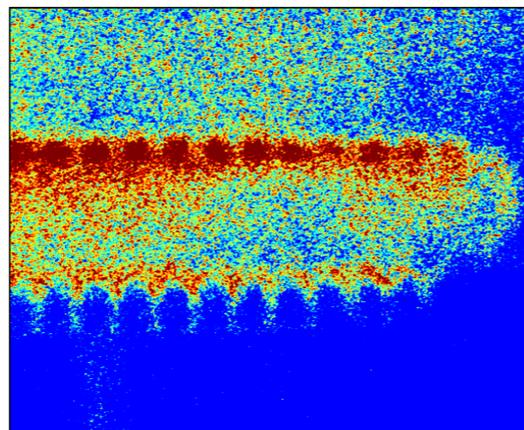


(b) Case A,  $t_2$

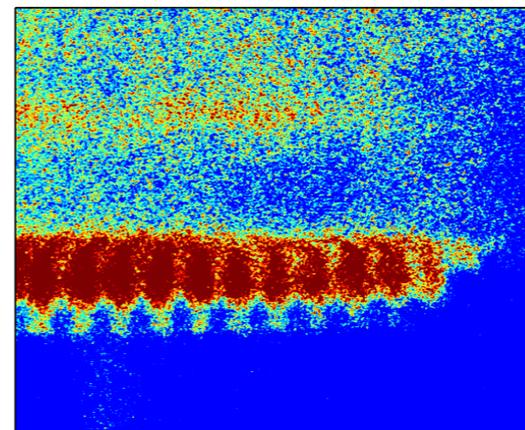


(c) Case A,  $t_3$

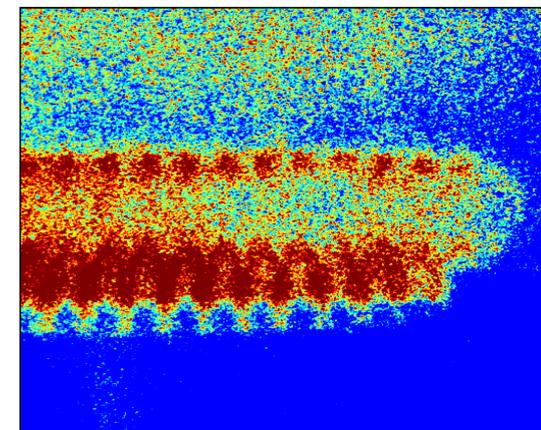
$$L_1 = 52 \text{ cm}$$



(d) Case B,  $t_1$



(e) Case B,  $t_2$



(f) Case B,  $t_3$

The poor quality of the images is due to the high sampling frequency and the low emission level of the flames.

# Data processing of the experiments

We saw the multivariable DMD :

$$\left( \begin{array}{ccc} u_1^1 & u_1^2 & u_1^t \\ | & | & | \\ u_n^1 & u_n^2 & u_n^t \\ p_1^1 & p_1^2 & p_1^t \end{array} \right)$$

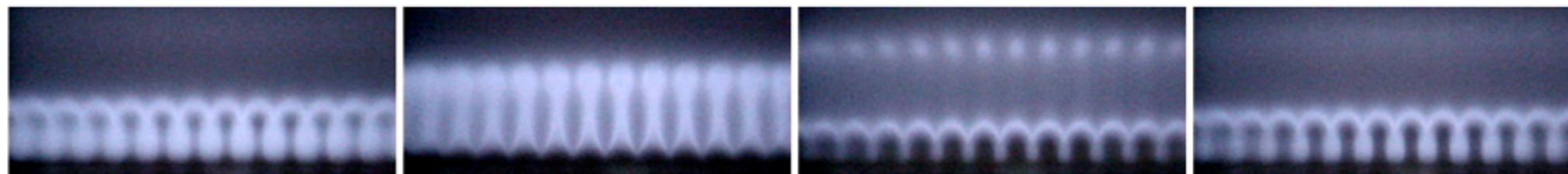
Parametric DMD uses data recorded at different time but with the same sampling rate.

$$\left( \begin{array}{ccc} p_{L_1}^1 & \text{-----} & p_{L_1}^t \\ p_{L_2}^1 & \text{-----} & p_{L_2}^t \\ | & & | \\ p_{L_n}^1 & \text{-----} & p_{L_n}^t \end{array} \right)$$

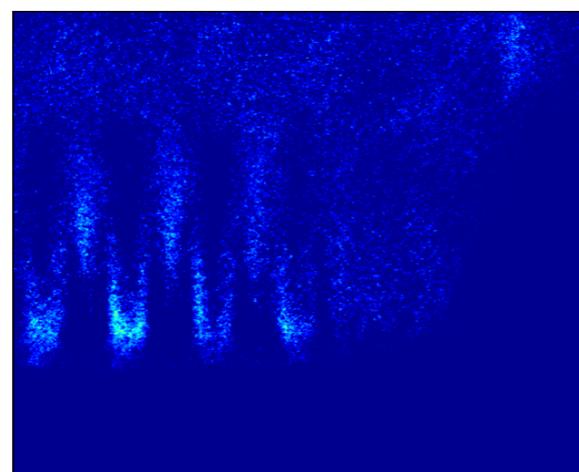
	$L_1 = 46$ cm (case A)	$L_1 = 52$ cm (case B)
Experiment	512 Hz	768 Hz
Parametrized DMD	493 Hz	740 Hz
Dominant mode	mode 3	mode 4

Four dominant modes are identified (1 to 4).

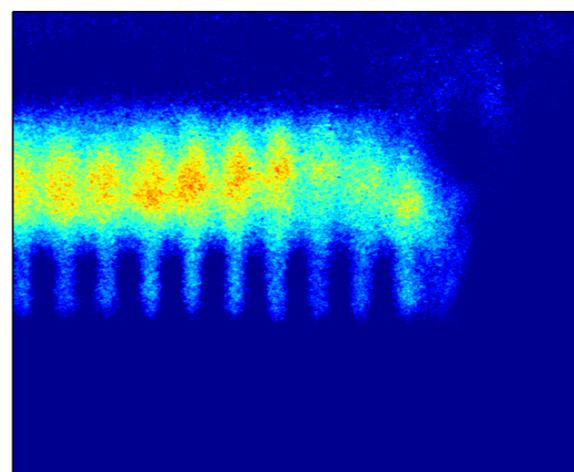
Depending on the bifurcation parameter, the weight of each mode changes.



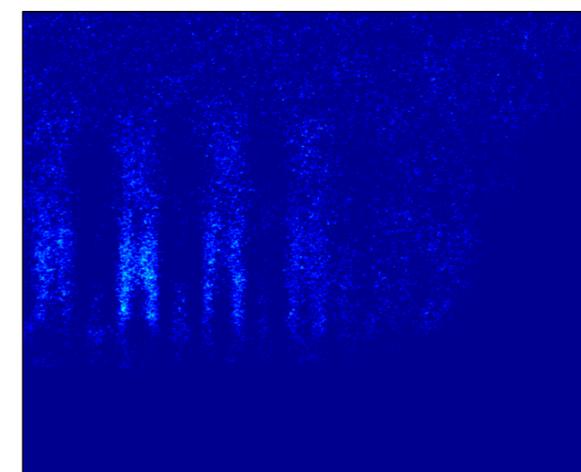
$L_1 = 46$  cm



(a) Case A, mode 2

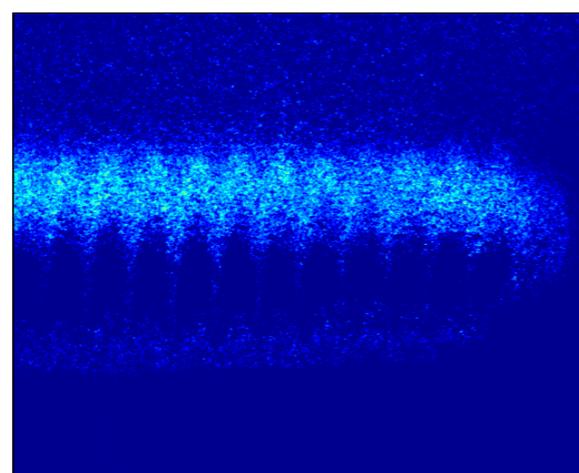


(b) Case A, mode 3

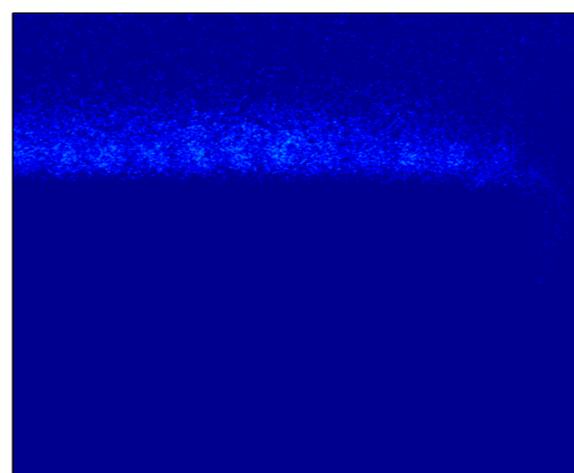


(c) Case A, mode 4

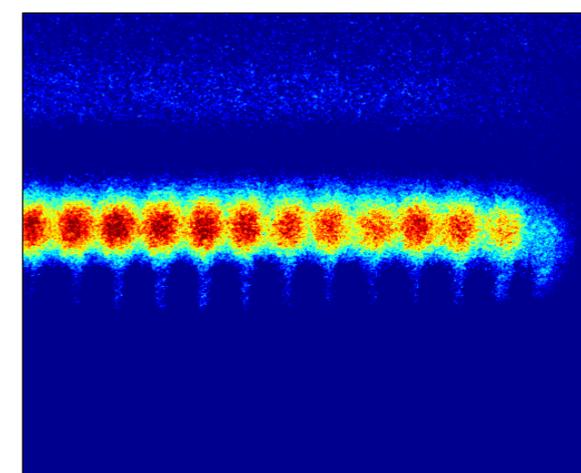
$L_1 = 52$  cm



(d) Case B, mode 2

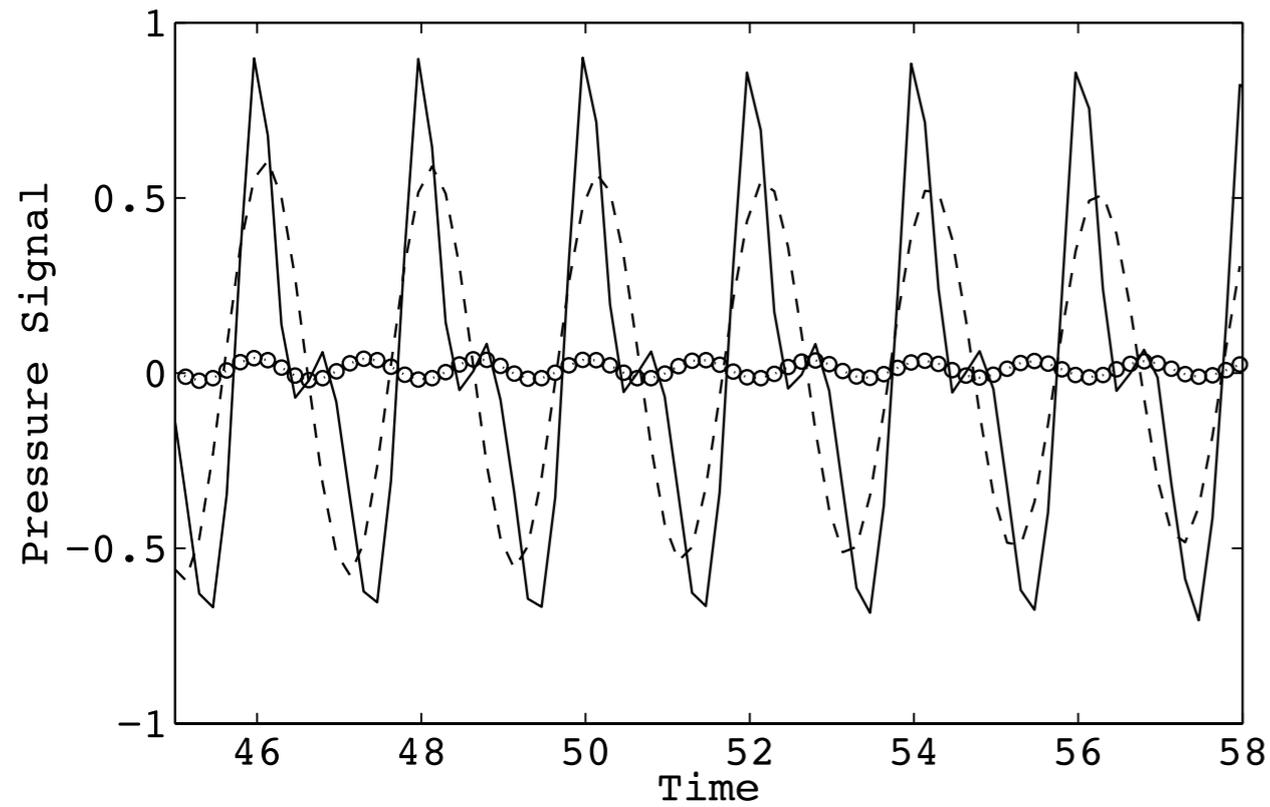


(e) Case B, mode 3



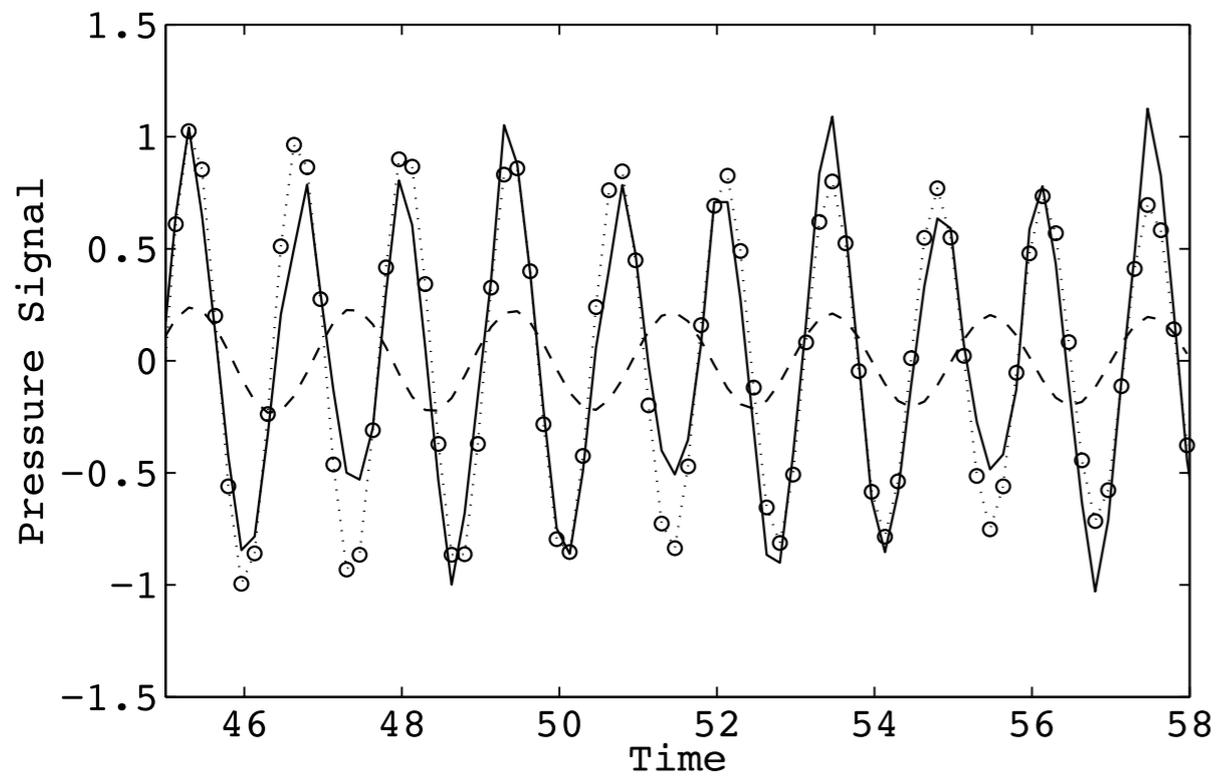
(f) Case B, mode 4

Depending on the bifurcation parameter, the weight of each mode changes.



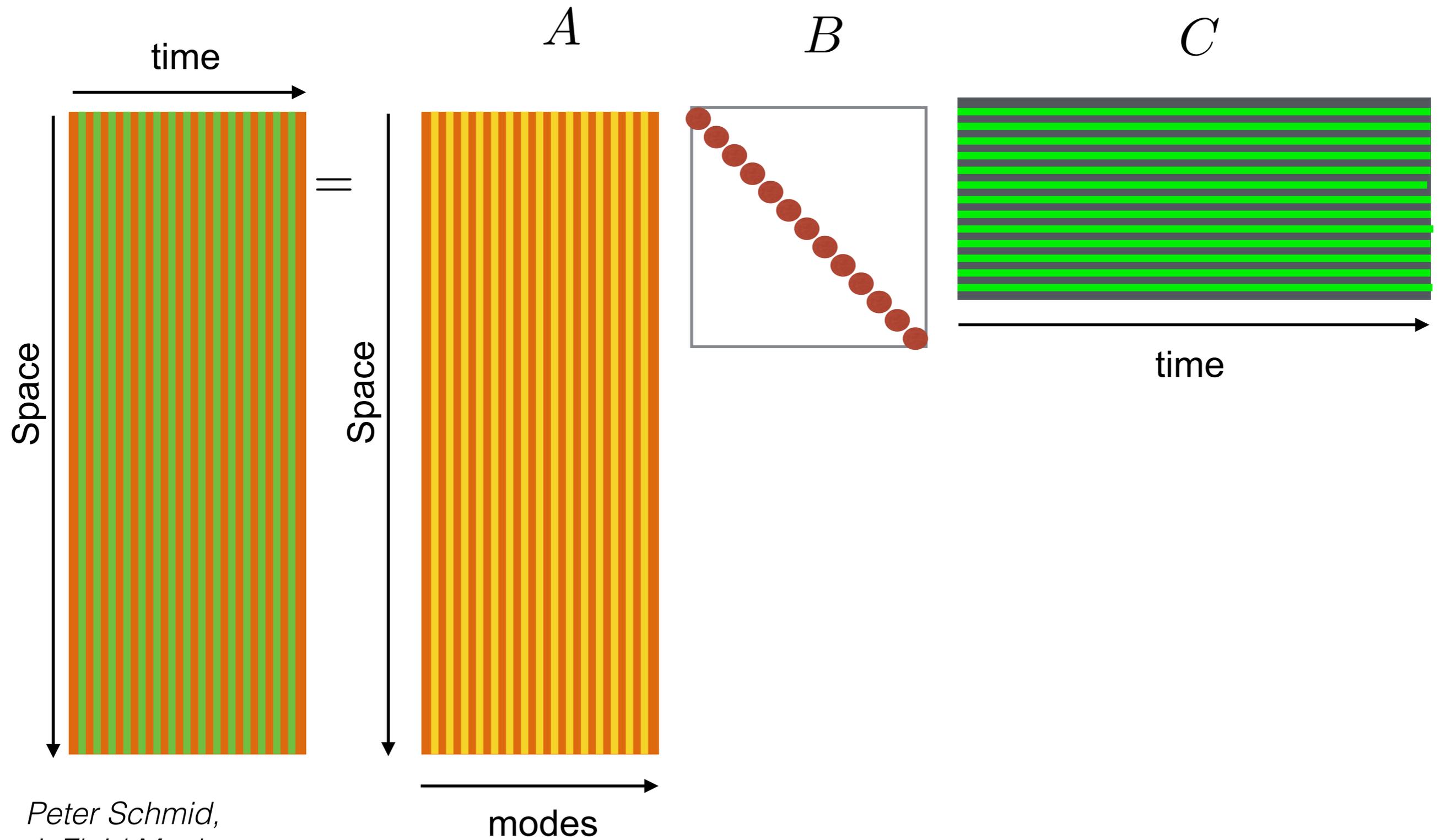
Before bifurcation

- experiment
- mode 3
- o mode 4



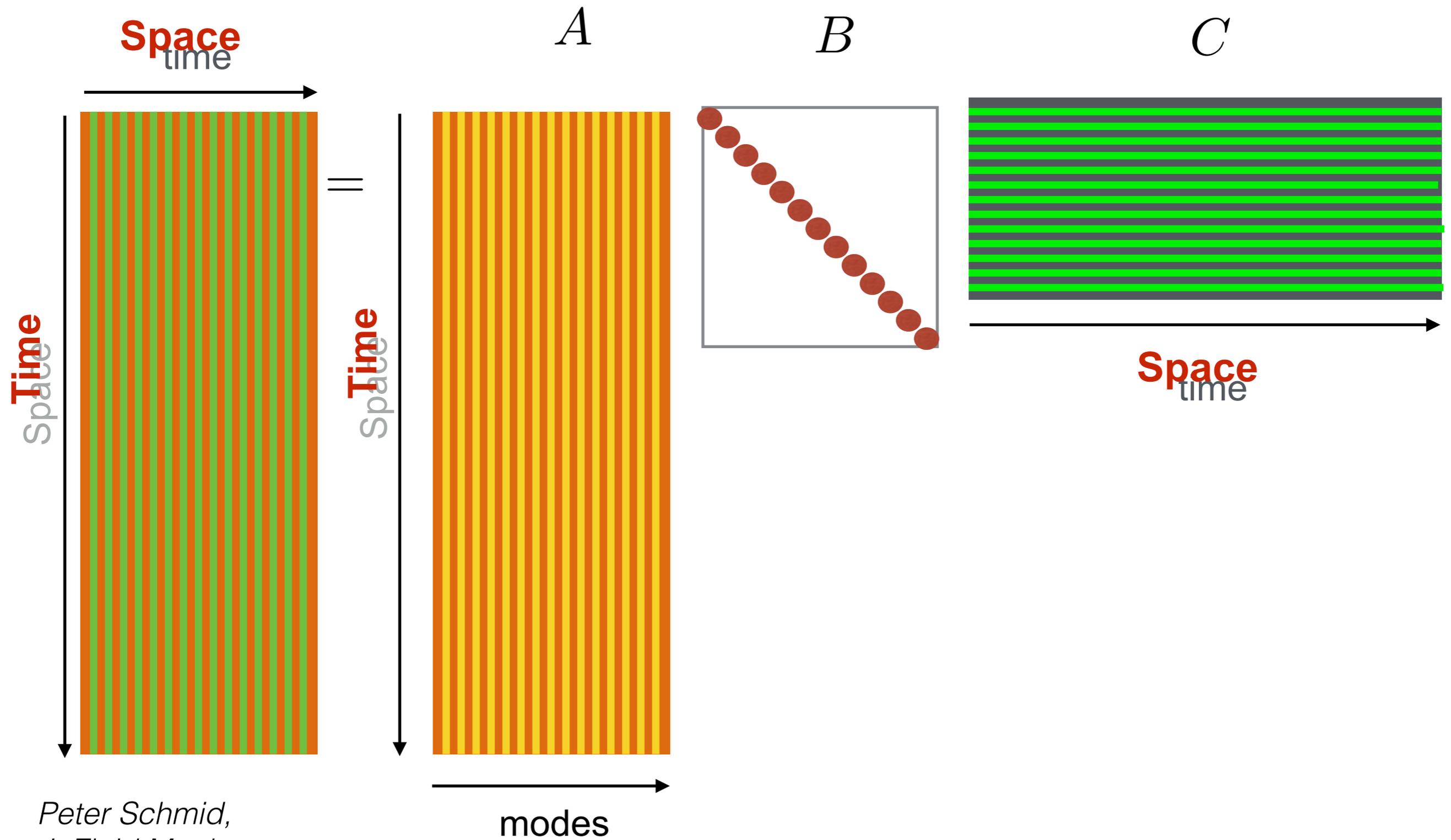
After bifurcation

**DMD generates a decomposition of the data based on the frequencies.**



*Peter Schmid,  
J. Fluid Mech.  
(2010)*

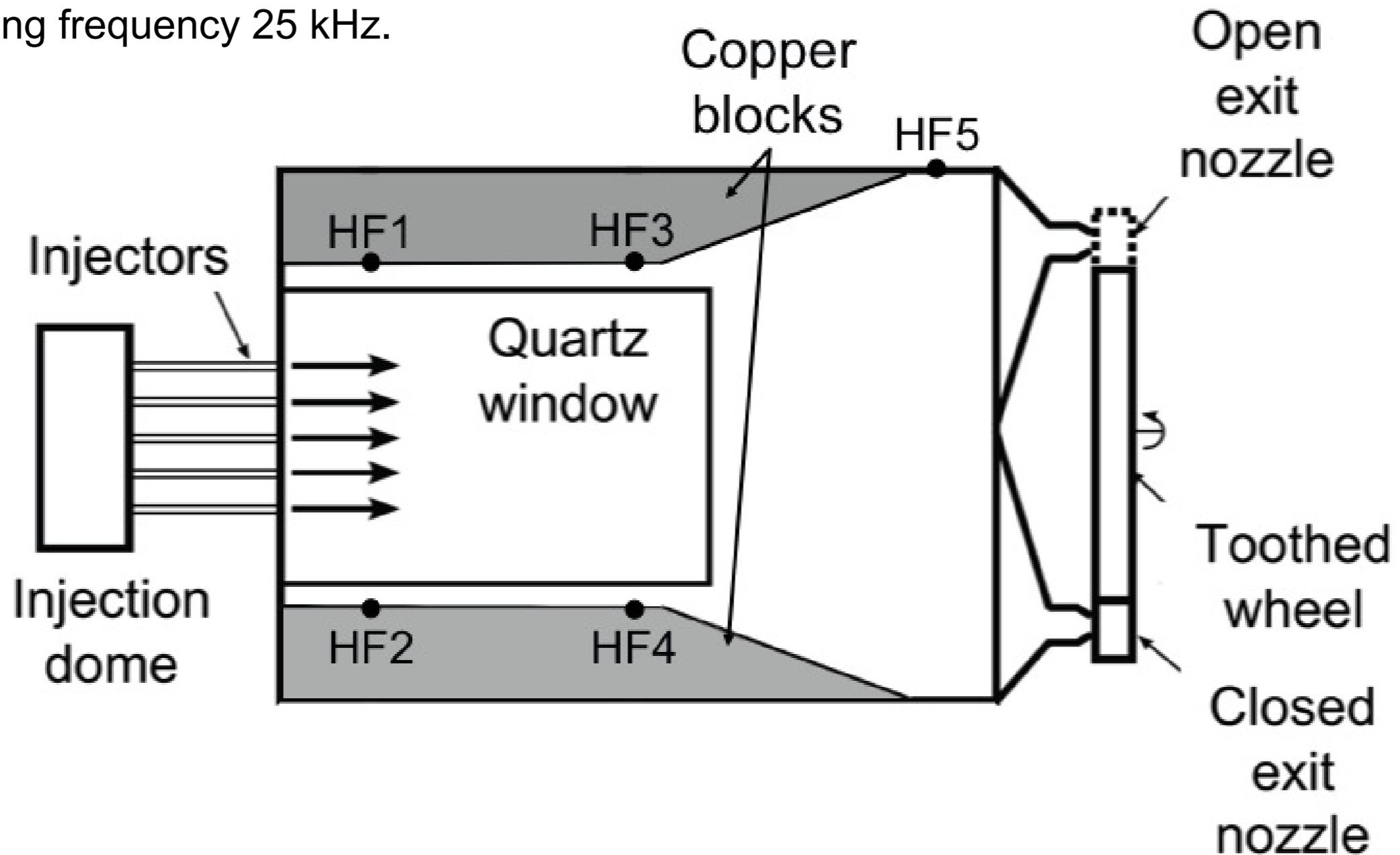
DMD generates a decomposition of the data based on the frequencies.



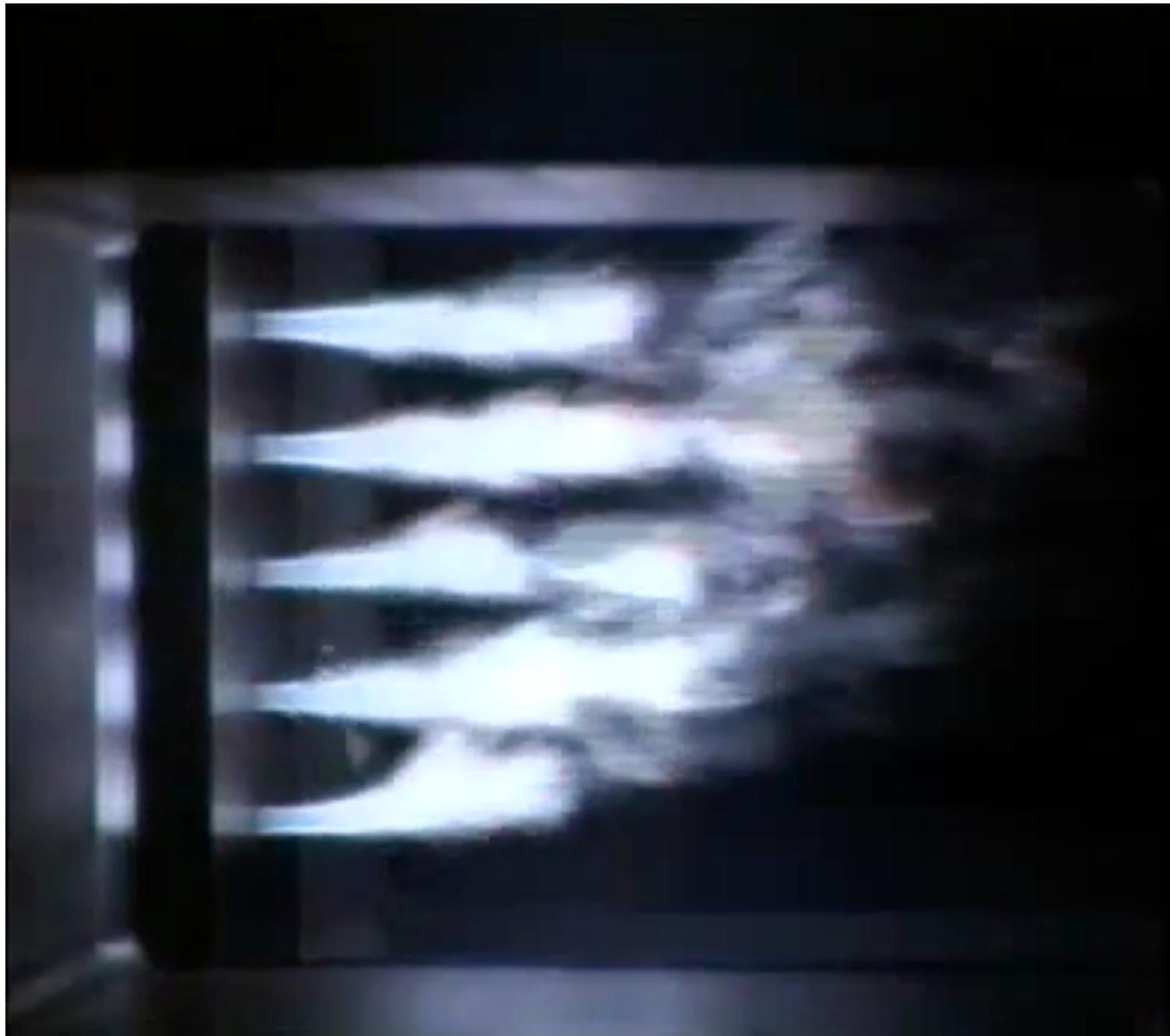
Peter Schmid,  
*J. Fluid Mech.*  
(2010)

# Spatial DMD

- Combustion LOx/LH2 ;
- Chamber pressure 30 bar ;
- External acoustic modulation ;
- Sampling frequency 25 kHz.



# Spatial DMD



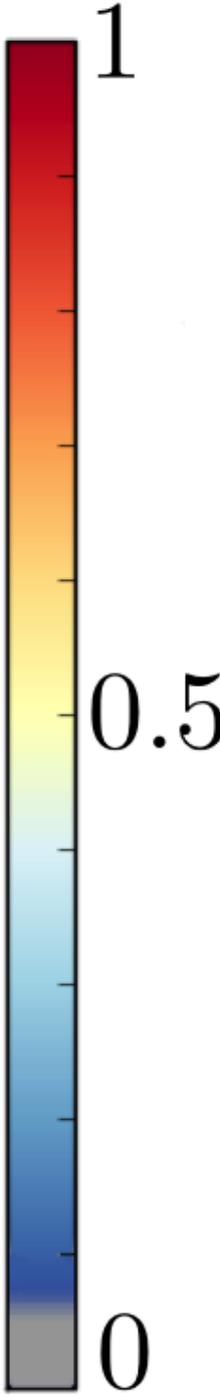
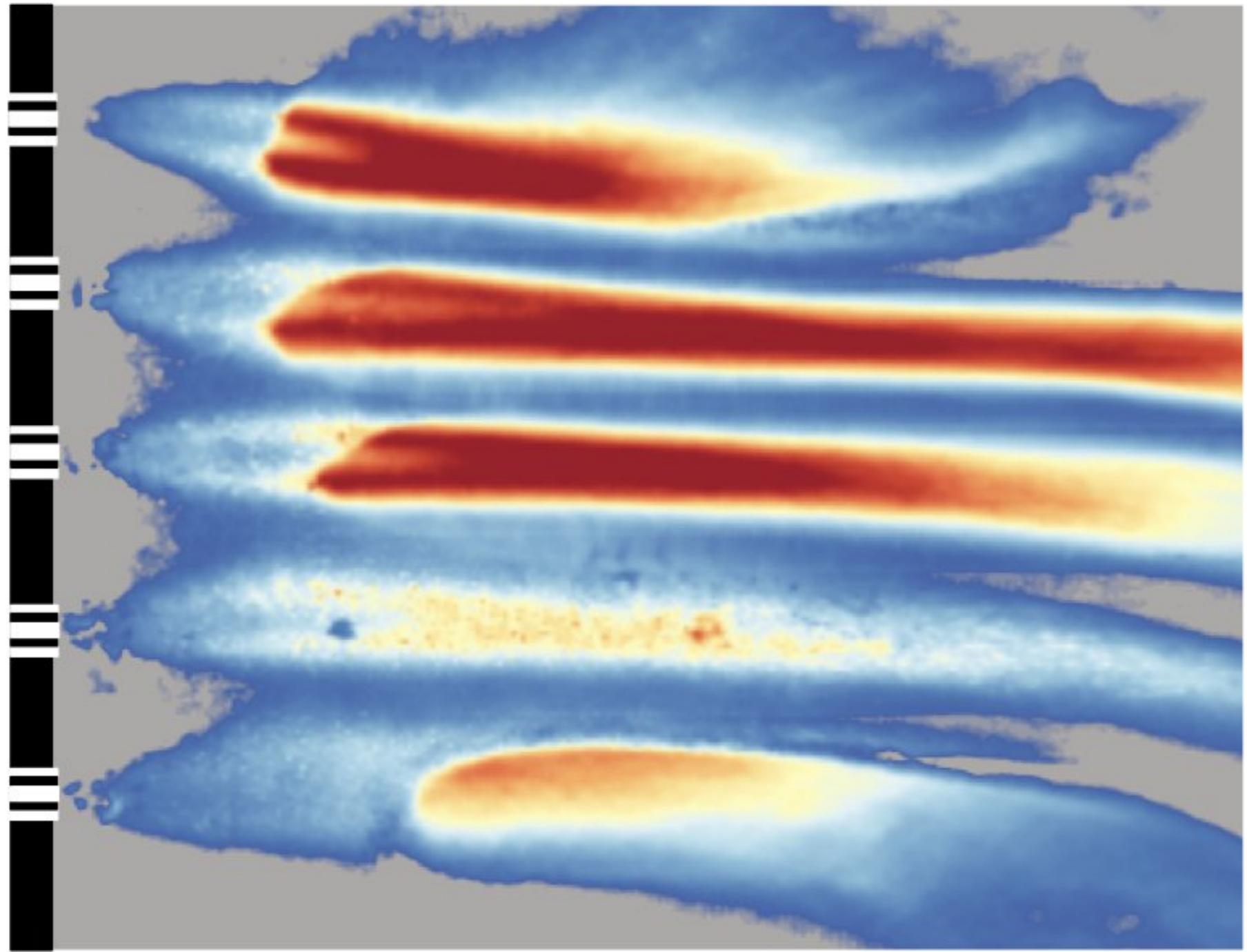
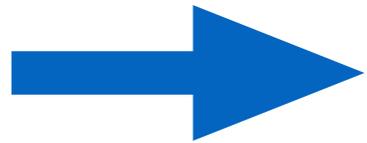
MASCOTTE (ONERA, Palaiseau)

# Spatial DMD

Average light emission from the flames

Coaxial injectors

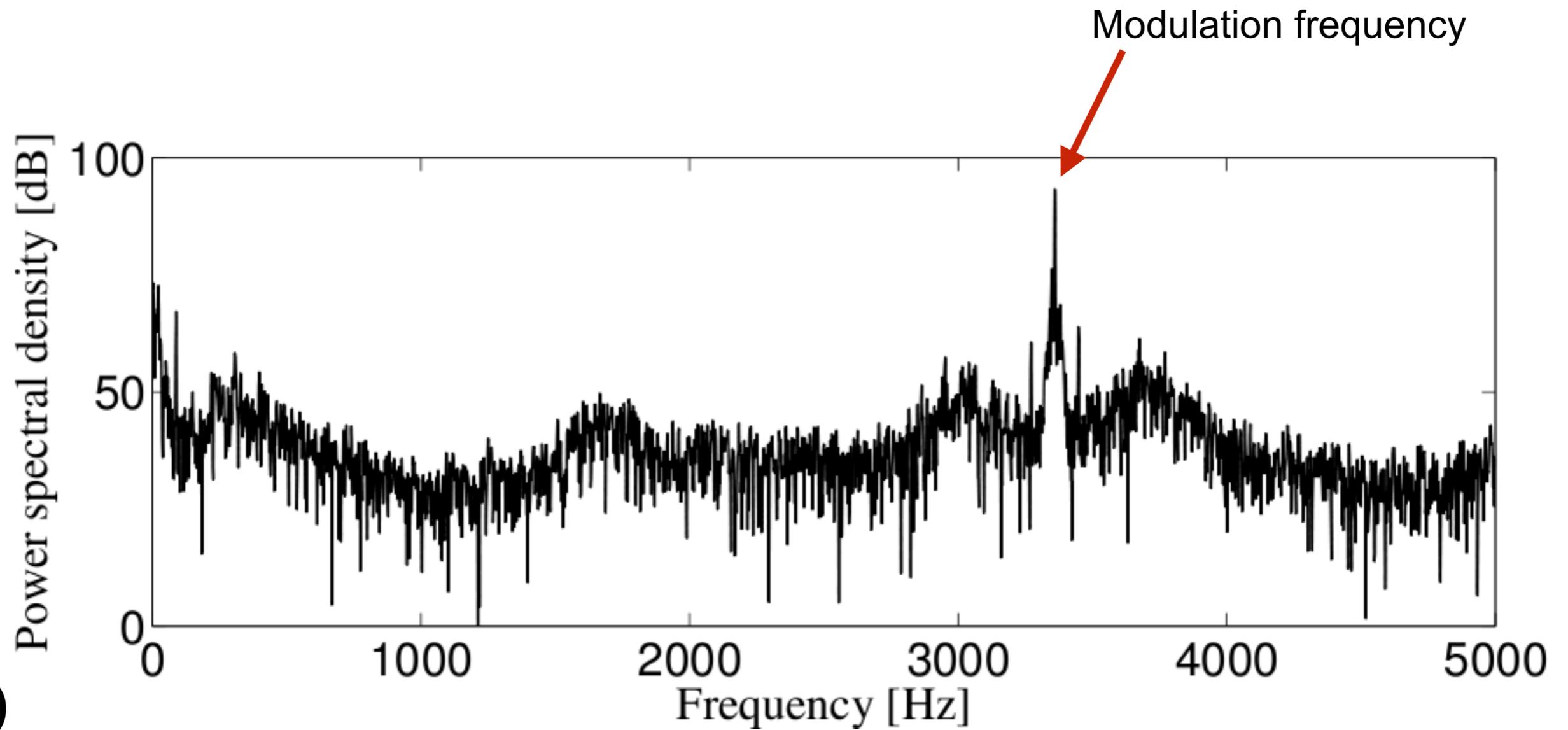
Flow direction



15 cm

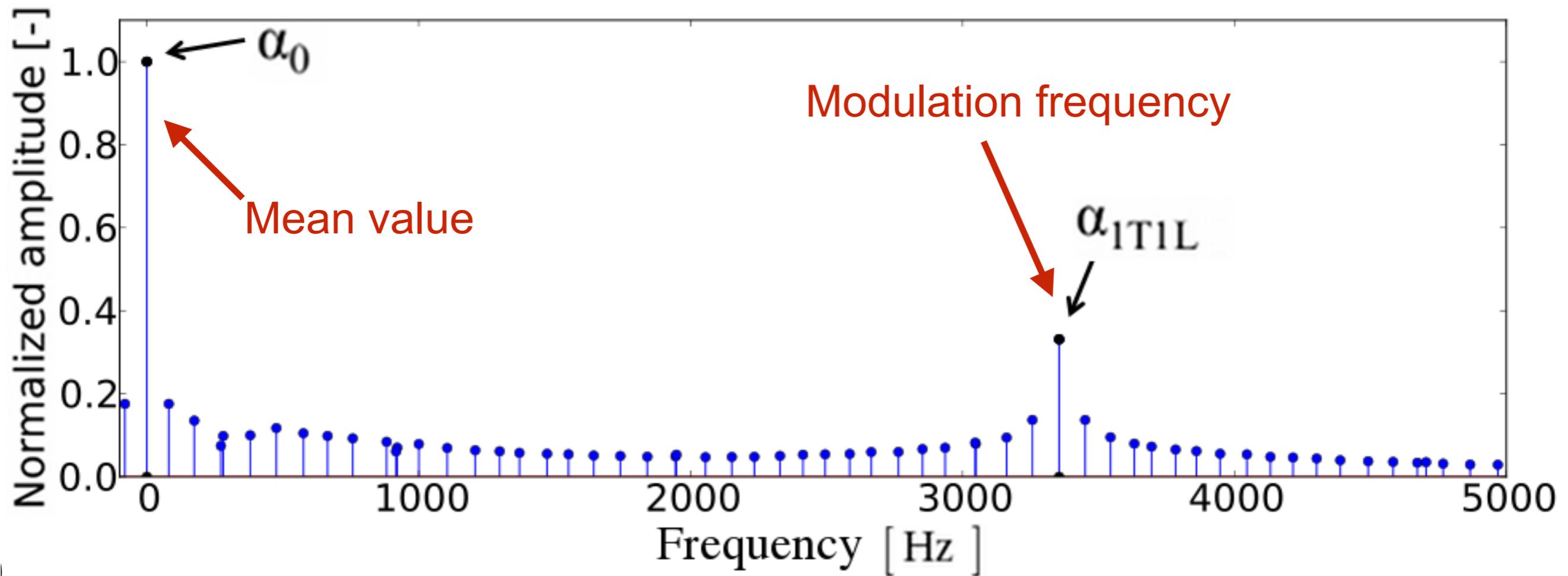
# Spatial DMD

Since the flow is acoustically modulated from an external source, pressure sensors detect mainly this frequency



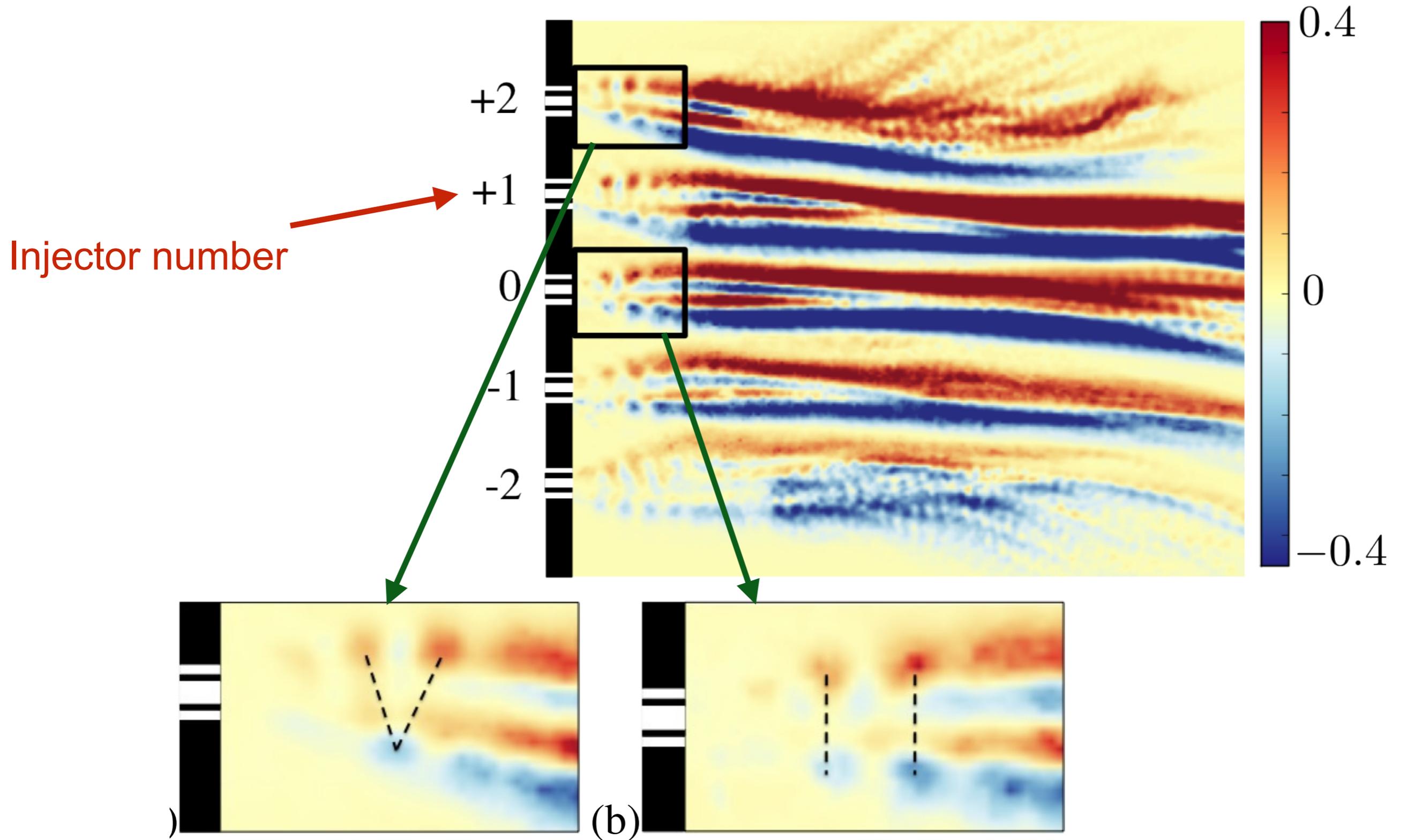
# Spatial DMD

A first multi-variable DMD is performed with the pressures and light emission.



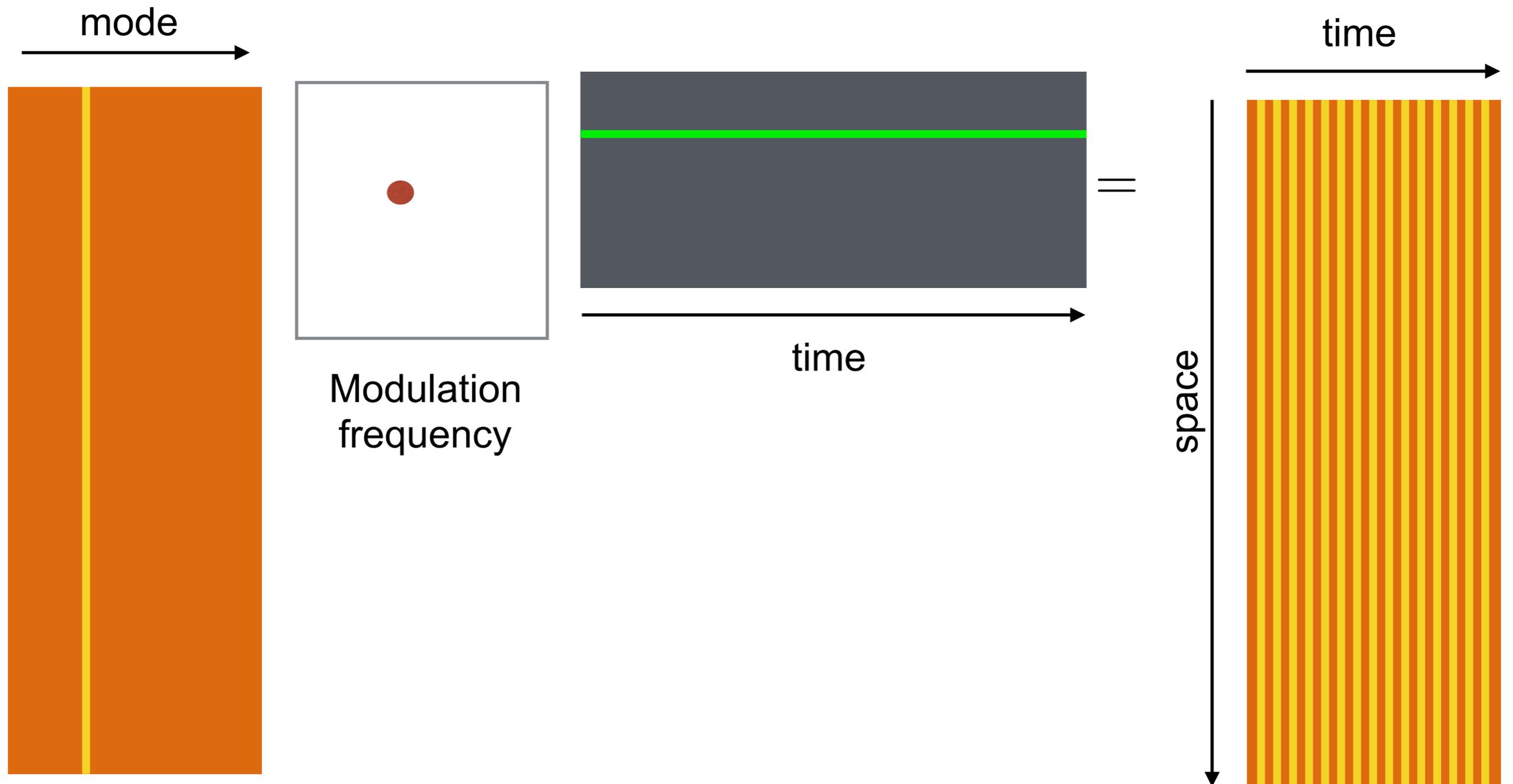
# Spatial DMD

All the dynamics is concentrated on one mode.



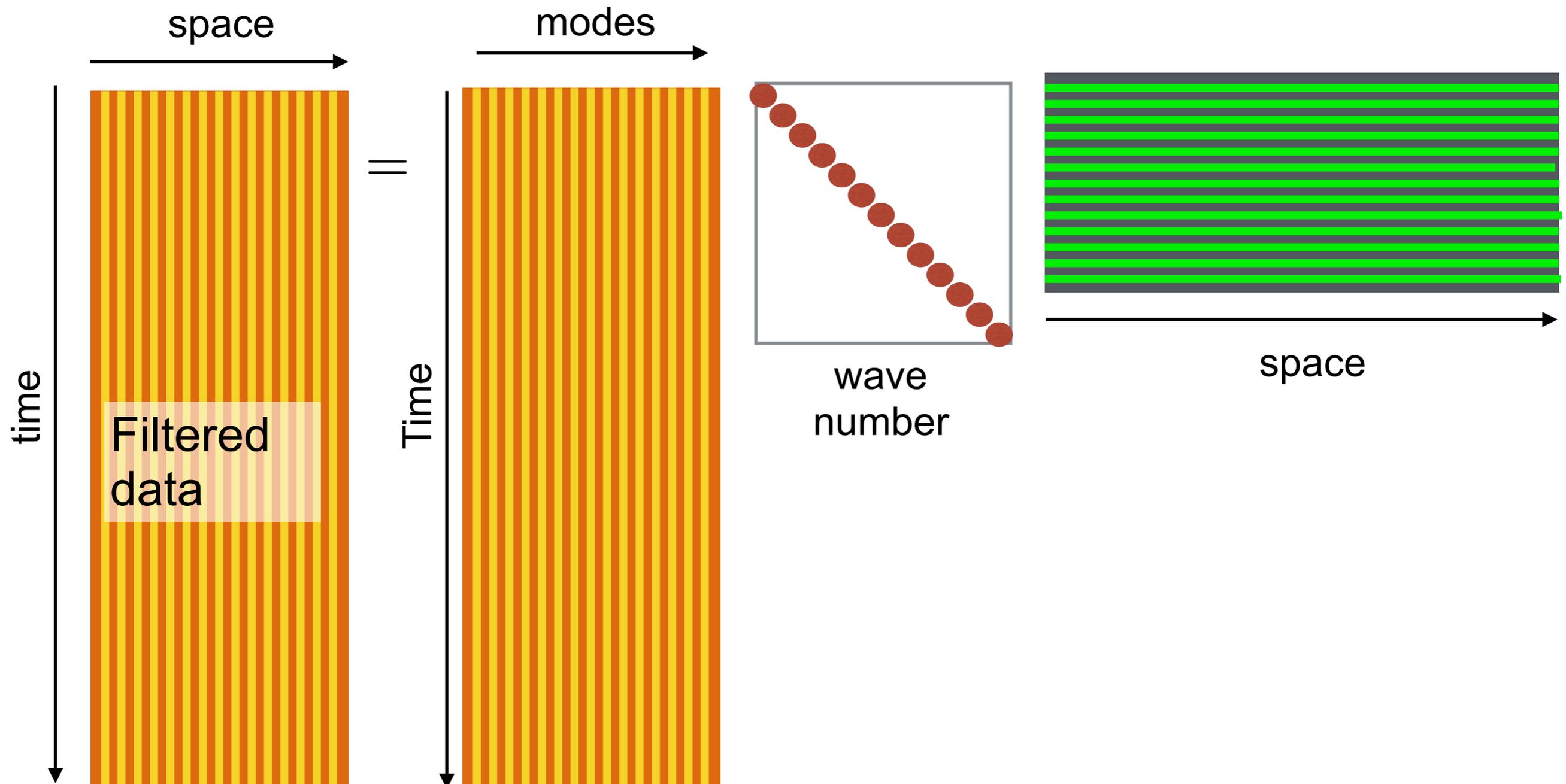
# Spatial DMD

DMD is first used to filter the emission field at the modulation frequency.



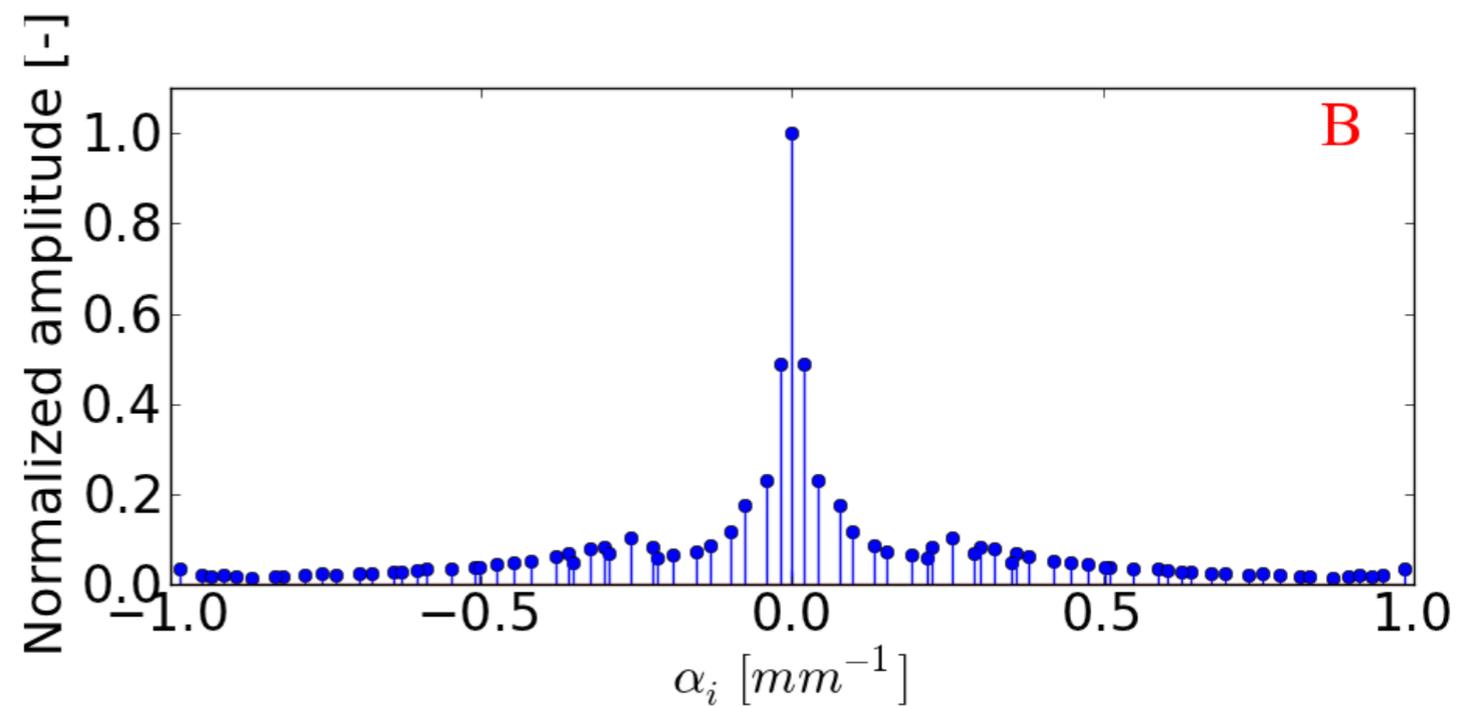
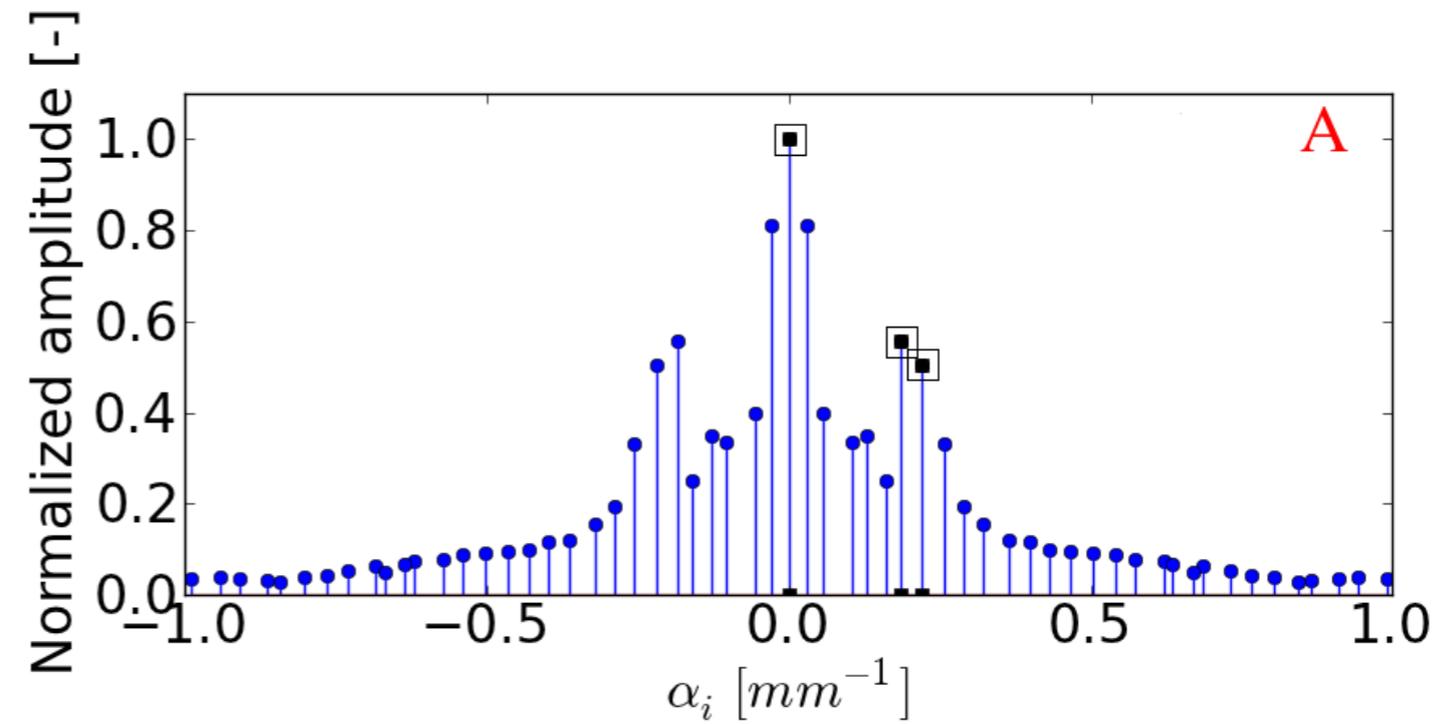
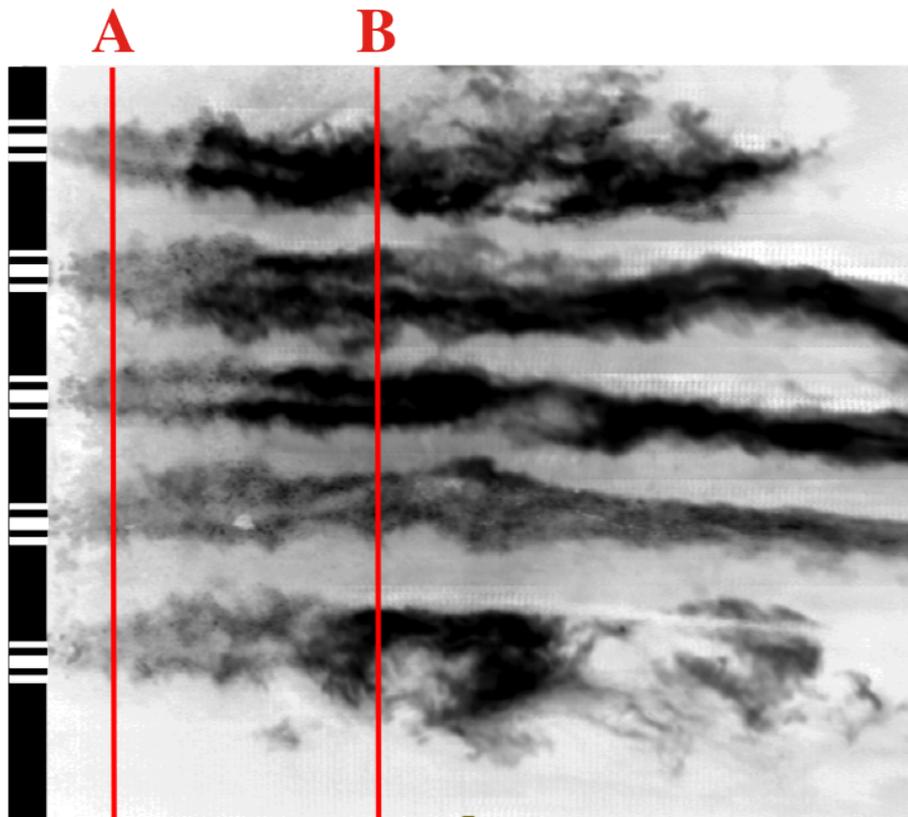
# Spatial DMD

The time evolution of one mode is rebuilt together with the average convective field.



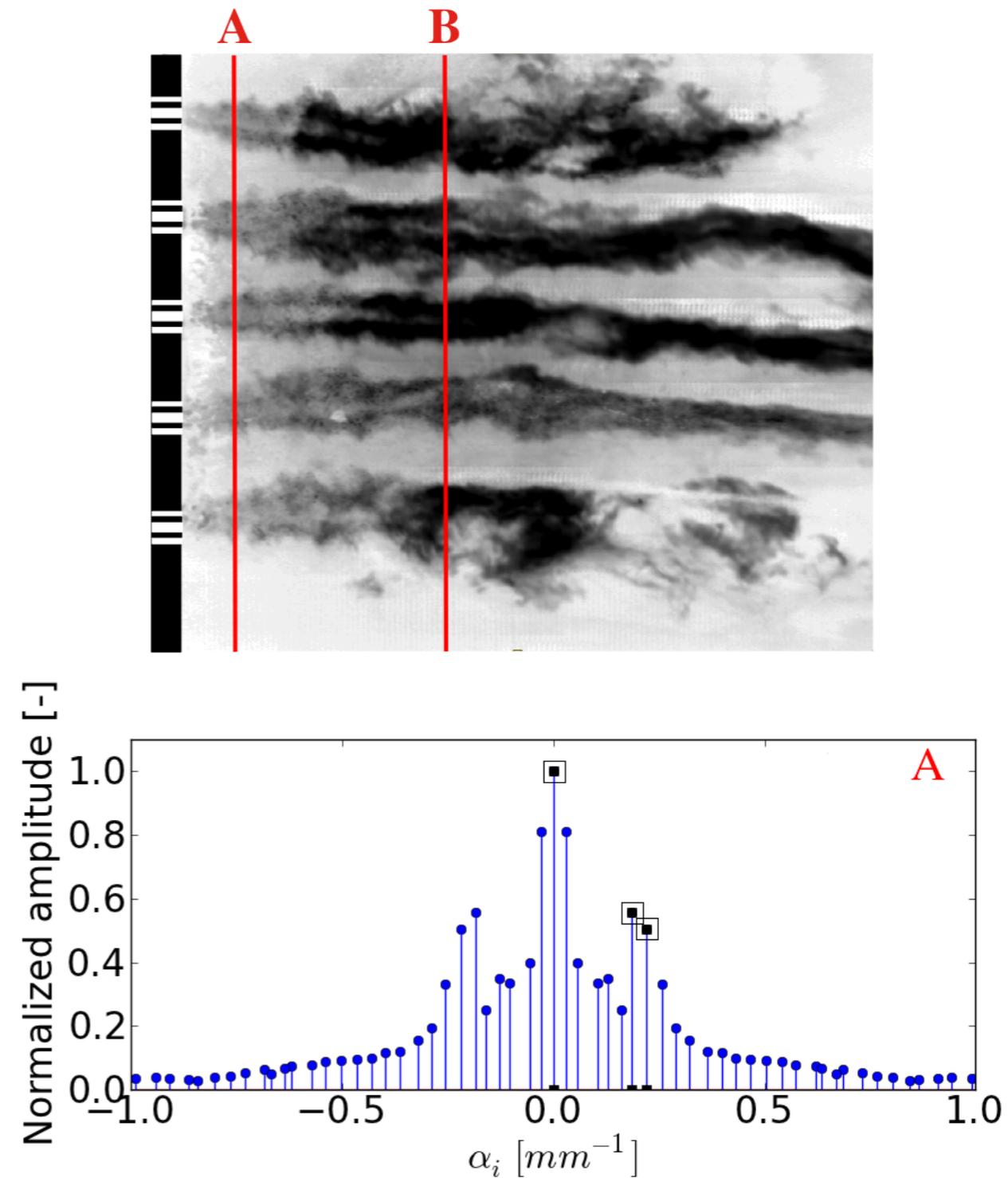
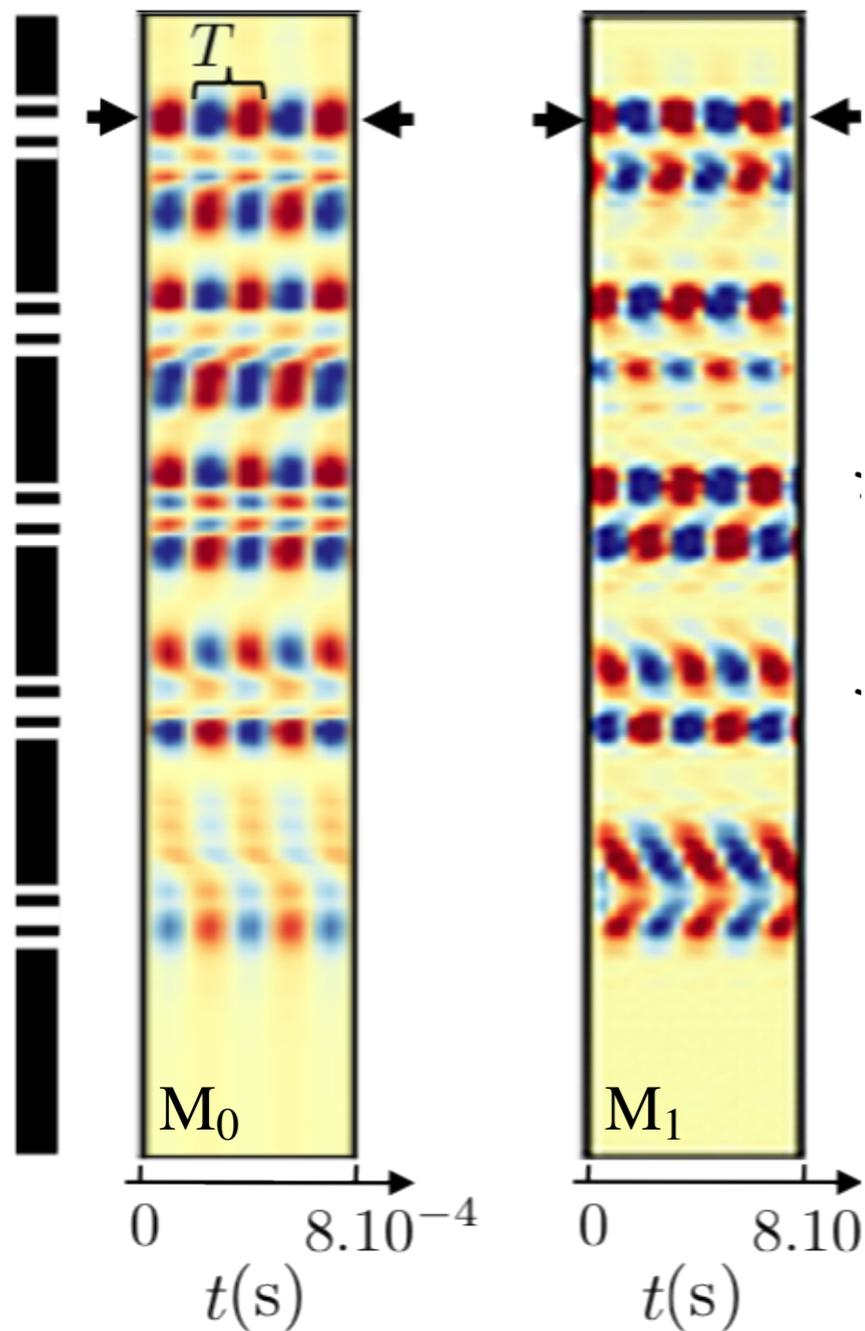
# Spatial DMD

Spatial DMD is performed on two subdomains of the chamber.



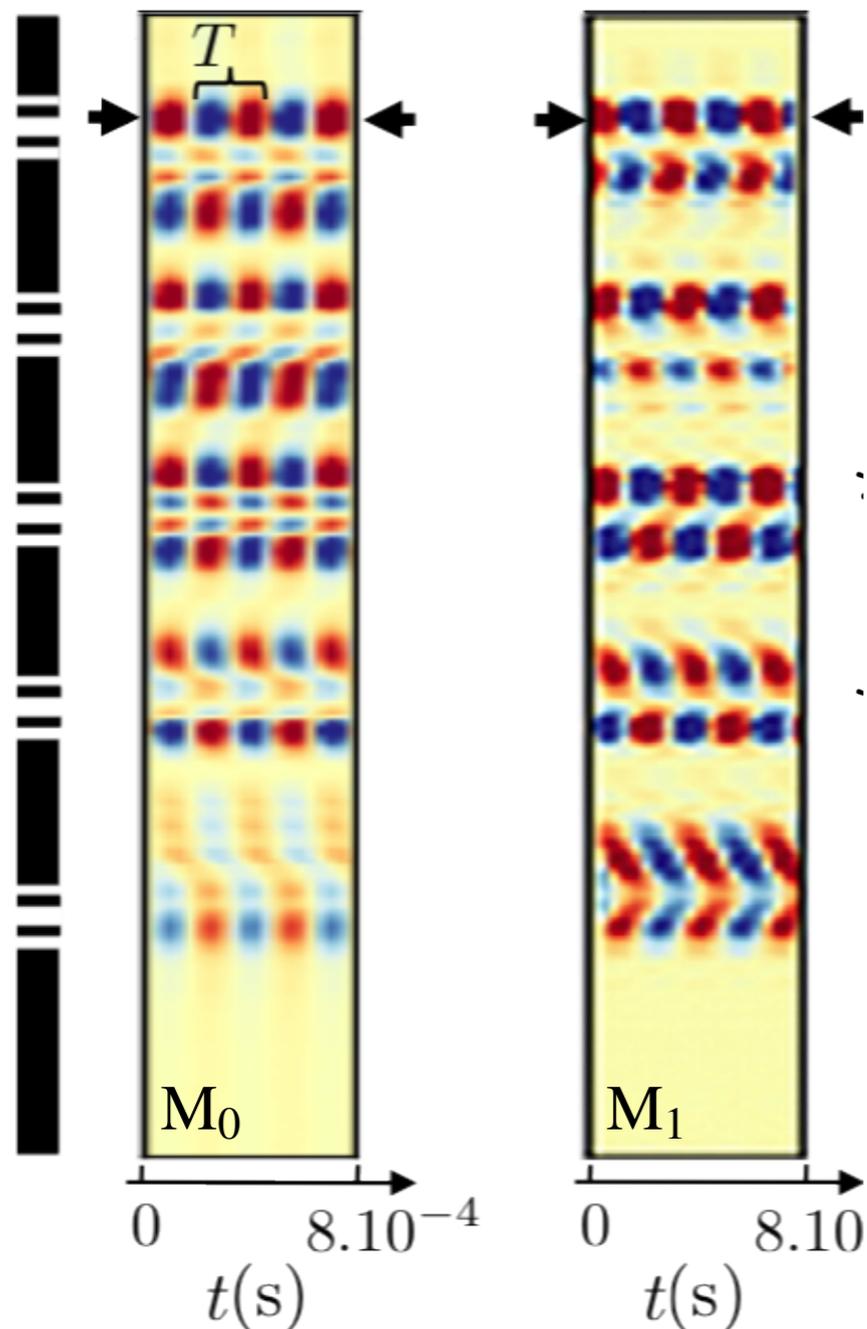
# Spatial DMD

Close to the injector (A),  
two modes (M0 and M1) evolve at the same frequency but different wavelength.



# Spatial DMD

One mode is associated to a transverse motion due to the modulation, the other to a longitudinal modulation generated by the injection lines.

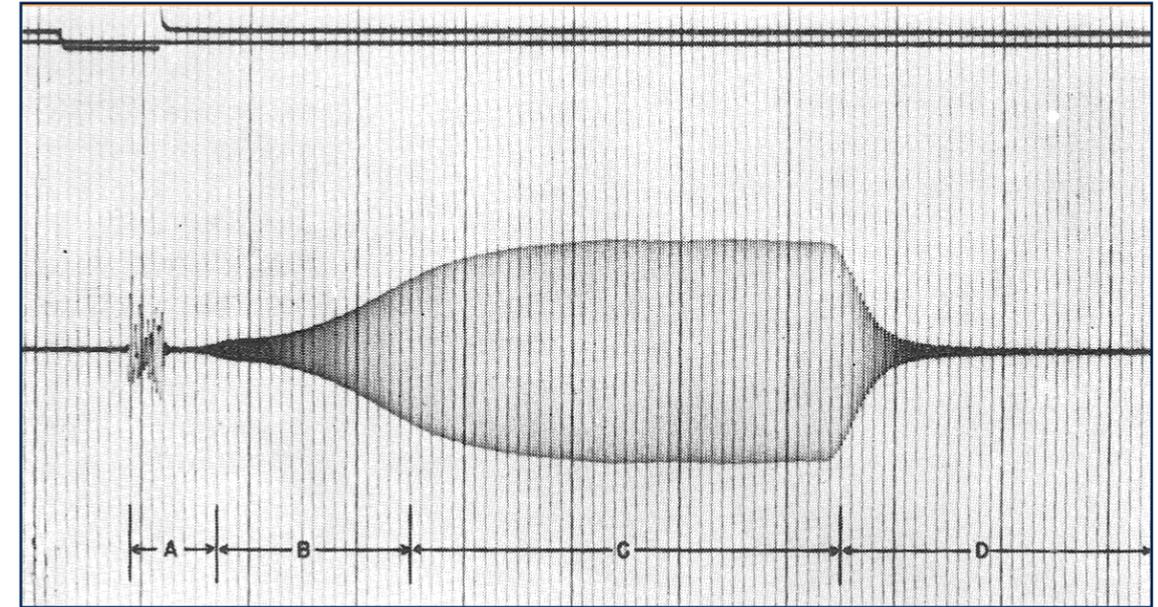


Spatial DMD separated to physical phenomena taking place at the same frequency but with different wavelength.

# Conclusion

One of the challenge in combustion is to predict the pressure evolution in the chamber.

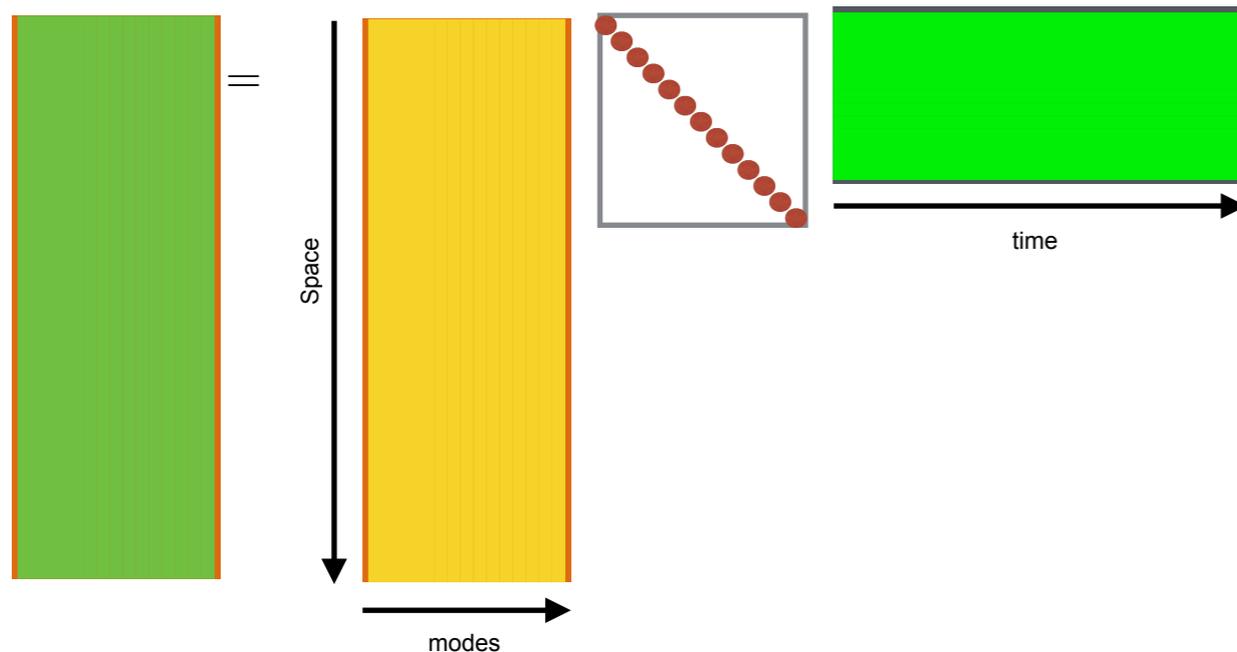
To reduce low order models, we need to understand the couplings taking place at different time and space scales.



Decomposition methods allow to separate time scales and space scales then simplify the understanding and the modeling.

$$p'(t, x) = a_1(t)\Psi_1(x) + \dots + a_4(t)\Psi_4(x)$$

$$\frac{da_i(t)}{dt} = \mathcal{F}(a_i)$$



By adapting the decomposition strategy to the case of study, relevant information can be extracted.